

no  
Action  
Ass  
Off

**GEORGIA INSTITUTE OF TECHNOLOGY**  
**OFFICE OF CONTRACT ADMINISTRATION**  
**SPONSORED PROJECT INITIATION**

Date: February 4, 1977

Project Title: Implementation of a N-Layer Computer Program for Pavement Design

Project No.: E-20-611

Project Director: Dr. James S. Lai

Sponsor: Federal Highway Administration

Agreement Period: From 1/21/77 Until 4/20/77

Type Agreement: Purchase Order No. 7-3-0031

Amount: \$5,900

Reports Required: Computer Program; Final Report

Sponsor Contact Person (s):

Technical Matters

Contractual Matters  
(thru OCA)

Mr. Robert L. Martin  
Federal Highway Administration  
Washington, D.C. 20590  
(202) 755-9370

Defense Priority Rating: None

Assigned to: Civil Engineering (School/Laboratory)

COPIES TO:

Project Director  
Division Chief (EES)  
School/Laboratory Director  
Dean/Director-EES  
Accounting Office  
Procurement Office  
Security Coordinator (OCA)  
Reports Coordinator (OCA)

Library, Technical Reports Section  
Office of Computing Services  
Director, Physical Plant  
EES Information Office  
Project File (OCA)  
Project Code (GTRI)  
Other \_\_\_\_\_

*b ossy 22*

GEORGIA INSTITUTE OF TECHNOLOGY  
OFFICE OF CONTRACT ADMINISTRATION  
SPONSORED PROJECT TERMINATION

*No action taken*  
*6/20/77*  
Date: June 20, 1977

Project Title: "Implementation of a N-Layer Computer Program for Pavement Design. (Vesys G)"

Project No: E-20-611

Project Director: Dr. J. S. Lai

Sponsor: DOT, Federal Highway Administration

Effective Termination Date: May 20, 1977

Clearance of Accounting Charges: May 20, 1977

Grant/Contract Closeout Actions Remaining:

- Final Invoice and ~~XXXXXXXXXXXXXX~~
- Final Fiscal Report
- Final Report of Inventions
- Govt. Property Inventory & Related Certificate
- Classified Material Certificate
- Other \_\_\_\_\_

Assigned to: Civil Engineering (School/Laboratory)

COPIES TO:

Project Director	Library, Technical Reports Section
Division Chief (EES)	Office of Computing Services
School/Laboratory Director	Director, Physical Plant
Dean/Director-EES	EES Information Office
Accounting Office	Project File (OCA)
Procurement Office	Project Code (GTRI)
Security Coordinator (OCA)	Other _____
Reports Coordinator (OCA)	

E-20-611

Final

SCEGIT-77-43

IMPLEMENTATION OF A N-LAYER COMPUTER  
PROGRAM FOR PAVEMENT DESIGN  
(VESYS G)

by

James S. Lai

Prepared for

Federal Highway Administration  
U. S. Department of Transportation  
Washington, D.C. 20590

April 30, 1977

School of Civil Engineering  
Georgia Institute of Technology  
Atlanta, Georgia 30332

SCEGIT-77-43

IMPLEMENTATION OF A N-LAYER COMPUTER  
PROGRAM FOR PAVEMENT DESIGN  
(VESYS G)

by

James S. Lai

Prepared for

Federal Highway Administration  
U. S. Department of Transportation  
Washington, D.C. 20590

April 30, 1977

School of Civil Engineering  
Georgia Institute of Technology  
Atlanta, Georgia 30332

**Technical Report Documentation Page**

1. Report No.	2. Government Accession No.	3. Recipient's Catalog No.	
4. Title and Subtitle  Implementation of a N-Layer Computer Program for Pavement Design (VESYS G)		5. Report Date  April 30, 1977	
7. Author(s)  James S. Lai		6. Performing Organization Code	
9. Performing Organization Name and Address  School of Civil Engineering Georgia Institute of Technology Atlanta, Georgia 30332		8. Performing Organization Report No.	
12. Sponsoring Agency Name and Address  Federal Highway Administration U.S. Department of Transportation Washington, D.C. 20590		10. Work Unit No. (TRAIS)	
		11. Contract or Grant No. Order No. 7-3-0031	
		13. Type of Report and Period Covered  Final	
15. Supplementary Notes			
16. Abstract  This report presents the information needed for use of the computer program VESYS G which integrates the n-layer viscoelastic closed form probabilistic primary response model with the current FHWA version of VESYS IIM flexible pavement structural design subsystem.  A limited sensitivity analysis was conducted on the VESYS G and the primary response model (PRIME). Results of these analyses were presented.  The theoretical formulation of the viscoelastic closed-form probabilistic solution for the primary response predictions in VESYS G was also included in this report.			
17. Key Words  Flexible Pavement Analysis, Probabilistic solution, viscoelastic solution, sensitivity analysis, permanent deformation, cracking, rutting, slope variance, Present Serviceability Index.	18. Distribution Statement		
19. Security Classif. (of this report)  Unclassified	20. Security Classif. (of this page)  Unclassified	21. No. of Pages	22. Price

## PREFACE

This report presents the information needed for the use of the computer program VESYS G which integrates the n-layer viscoelastic closed form probabilistic primary response model developed by the University of Utah with the current FHWA version of VESYS IIM structural subsystem.

Support for the work was provided by the Federal Highway Administration, Office of Research and Development, Order No. 7-3-0031. The author wishes to thank Mr. William J. Kenis and James Sherwood for the valuable technical assistance.

The program listing for VESYS G and the printouts for sensitivity analysis of PRIME are included in a separate volume.

## TABLE OF CONTENTS

	Page
PREFACE . . . . .	i
TABLE OF CONTENTS . . . . .	ii
CHAPTER 1. DESCRIPTION OF VESYS G . . . . .	1
CHAPTER 2. OPERATING INSTRUCTIONS . . . . .	5
2.1 VESYS Source Code . . . . .	6
2.2 Input Data . . . . .	7
2.3 Types of Run Available . . . . .	11
2.4 Directory of New Commands . . . . .	14
CHAPTER 3. PROGRAM DOCUMENTATION . . . . .	16
3.1 Macro Program Structure . . . . .	16
3.2 PRIME Program . . . . .	16
CHAPTER 4. SENSITIVITY ANALYSIS . . . . .	22
4.1 Sensitivity Analysis of VESYS G . . . . .	22
4.2 Sensitivity Analysis of PRIME . . . . .	23
CHAPTER 5. RECOMMENDATIONS . . . . .	28
APPENDIX 1: Sample Input/Output . . . . .	29
2: Output for Sensitivity Analysis of PRIME . . . . .	83
3: Viscoelastic Closed Form Probabilistic Solution for PRIME . . . . .	84
4: Program Listing . . . . .	100

CHAPTER 1  
DESCRIPTION OF VESYS G

Work in the FHWA Project 5C entitled "New Methodology for Flexible Pavements" has developed a new flexible pavement design procedure called the VESYS design system. One part of this system is the pavement analysis computer program VESYS IIM which will predict flexible pavement response, distress and performance from a set of mechanistic model formulations. The overall flow chart for the computer program is shown schematically in Figure 1. The primary response model in the computer program uses a three-layer viscoelastic and probabilistic model. This three-layer model is not highly practical considering today's pavement types which are made up of from three to six layers of different materials. In addition, subdivision of any of these layers, where moisture and temperature gradients exist, is desirable and these given pavement systems could be considered to consist of even more numbers of layers.

The N-layer Viscoelastic closed form probabilistic primary response computer program, developed under FHWA contract with the University of Utah has the capabilities to analyze any number of layers and is ideally suited for use in the VESYS design system.

Thus, the main objective of this research effort is to integrate the Utah N-layer primary response model with the current VESYS IIM computer program. The approach is illustrated in Fig. 2a and Fig. 2b. Figure 2a shows the current VESYS IIM Macro program structure. In addition to MAIN program, the VESYS IIM consists of three tasks, CURVIT STATIC, and RANDOM where:

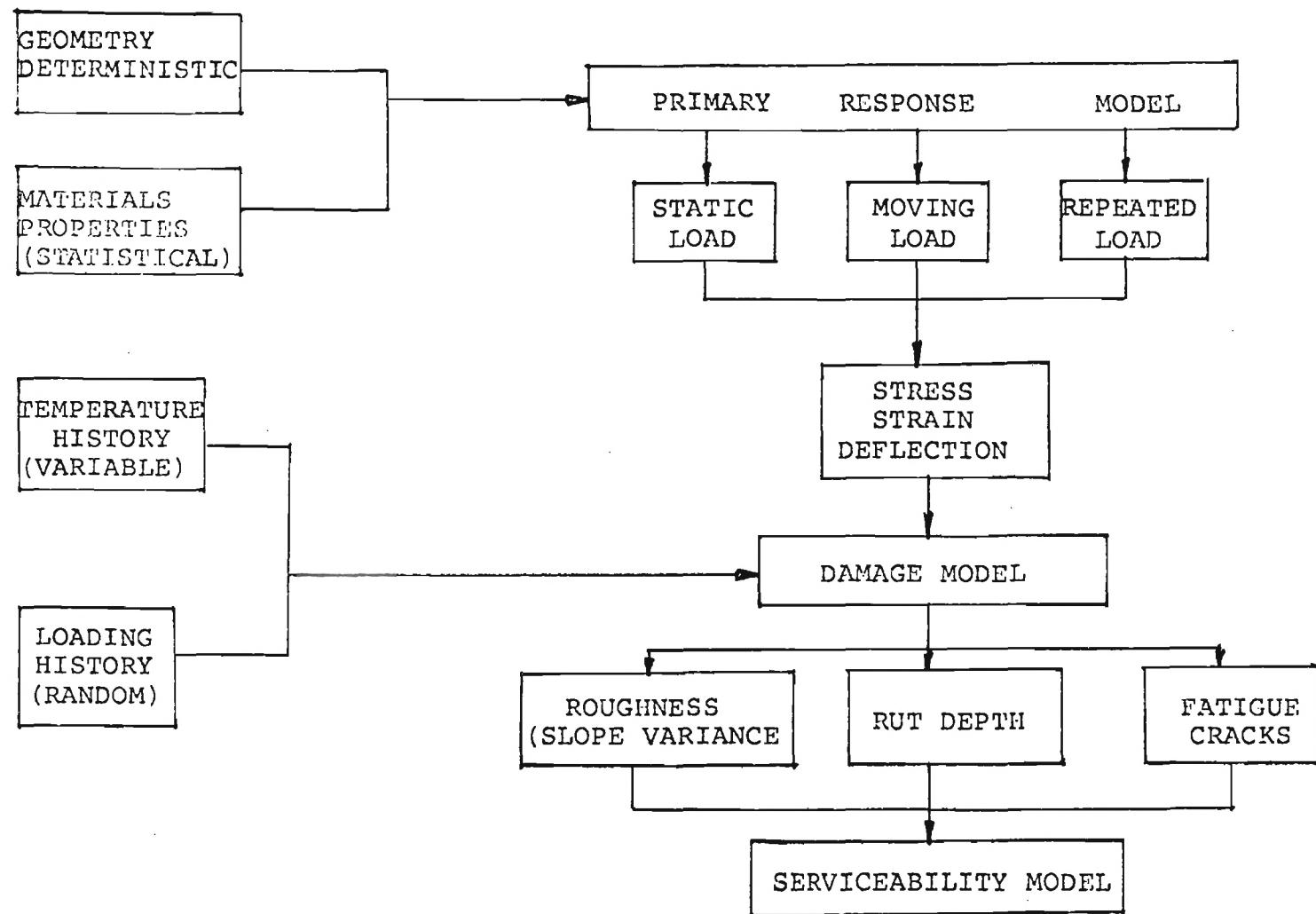


Figure 1. Flow Chart of VESYS IIM Computer Program.

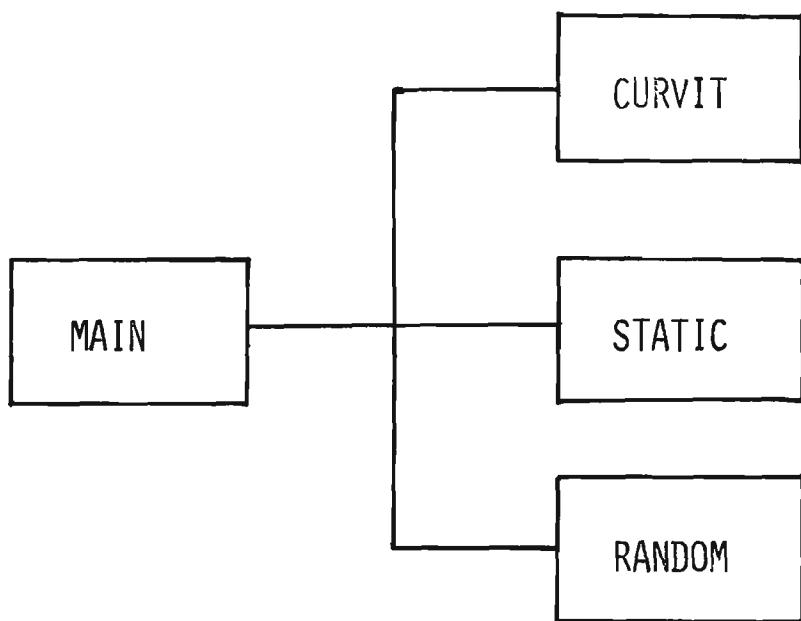


Figure 2A. VESYS IIM Marco Program Structure

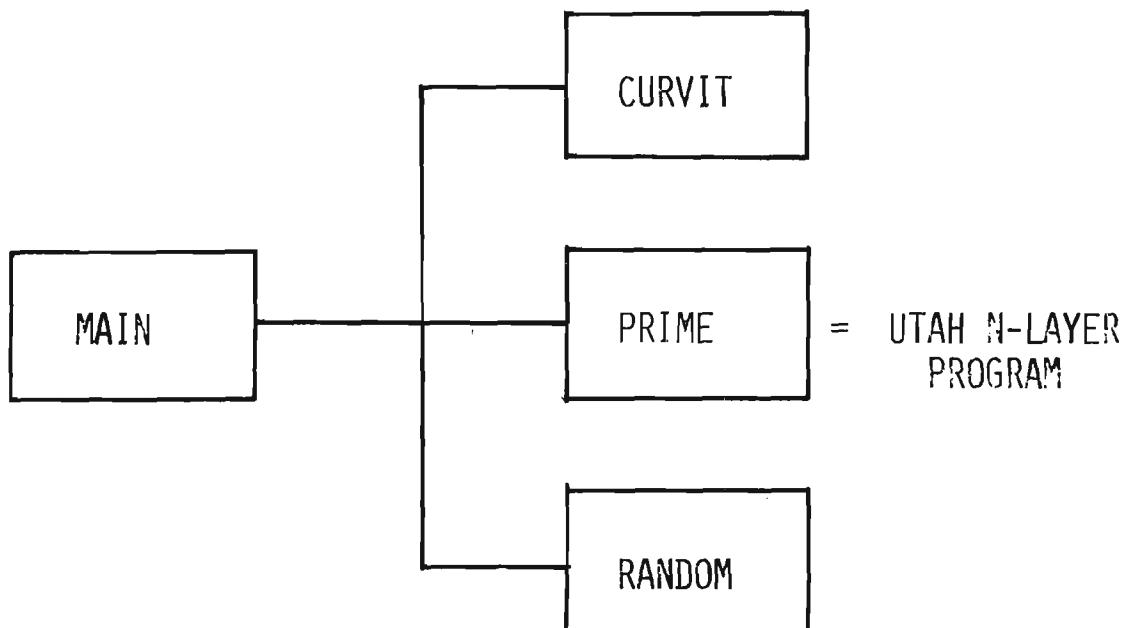


Figure 2B. VESYS G Marco Program Structure

- MAIN - handles all data input and stores it in a form suitable for use by the other subcomponents.
- CURVIT - performs a least square curve-fitting technique using Dirichlet Series.
- STATIC - computes the closed form probabilistic solution, mean and deviation of the stress, strain and deflection at the prescribed positions of a three-layer viscoelastic pavement system.
- RANDOM - computes the pavement response to a repeated load, the pavement distresses (rutting, cracking and slope variance), and the pavement serviceability.

In Fig. 2b, the integrated system (VESYS G), the Utah n-layer program (PRIME) was substituted for the STATIC. Thus, PRIME will perform the similar function as the STATIC does in computing the closed form probabilistic solution of the stress, strain and deflection at the prescribed positions in the pavement system. The main difference is that PRIME can handle pavement systems with the number of layers in the system greater than 3,\* while STATIC can only handle three layer pavement systems.

---

\* The program version listed in the Appendix has the capability to handle systems up to 7 layers. However, the program is designed to solve systems of any number of layers. For solving problems of more than 7 layers, layer-number related variable dimensions should be changed for all subroutines.

CHAPTER 2  
OPERATING INSTRUCTIONS

The operating instructions for VESYS G are similar to the current VESYS IIM version. It is recommended that anyone planning to use this program should first familiarize himself with the VESYS USER'S MANUAL for VESYS IIM prepared by FHWA, Office of Research.\*

---

\* VESYS USER'S MANUAL, Prepared by the Federal Highway Administration, Office of Research, Developed under FCP Prospect 5C.

## 2.1 VESYS SOURCE CODE

Developed on: IBM 360/65 under OS

Compiler: IBM FORTRAN G Level 21

Typical Statistics:

<u>Step</u>	<u>CPU Seconds</u>	<u>Core (Bytes)</u>
COMPILE	60.55	144 K
GO (Execute)	168.99 (5 layers)	582 K
	54.96 (3 layers)	

Machine-Dependent Considerations:

Word Size - 32 bits

Input/Output Files:

- Input is via logical unit 5 (card reader)
- Output is via logical unit 6 (line printer)

## 2.2 INPUT DATA

### 2.2.1 Commands

Input data values for VESYS are read from cards. Each input variable is associated with a "command" word, which is punched before the value of the variable on the data card. The program reads the command first, and then uses it as a keyword for matching the subsequent data value to the proper variable. This allows some flexibility in the order to input data.

After the value of a variable is read, it is checked against a predefined "reasonable" range of values. If the value of any variable lies outside this range, an error message is printed and the program is terminated.

Because many of the input variables have predefined "default" values stored within the program, it is not necessary to input a value for every variable. VESYS automatically assumes the default value for any of these variables which are not input.

VESYS is designed to cycle back to its own starting point so that multiple sets of data may be run with one execution of the program. The data for each separate problem is terminated in the input deck by an ENDOFRUN command. This command signals the program to begin executing with the data that has been read thus far. When the problem is complete, VESYS begins reading data for the next "run" with the first card after the previously read ENDOFRUN command. The last ENDOFRUN command in a data deck is followed immediately by an ENDOFRUN command, which causes the program to cease processing.

When a "job" is submitted with multiple "runs", the first run uses the default values, supplanted where indicated by the input data. Each

subsequent run begins with the data values left over from the previous run and supplants where necessary with its own input data. It may be useful to think of each run of a job as having the values used in the previous run by default. Any variables to be changed can be explicitly input. Others will remain the same. This feature allows the user to see the effect of changing a few variables without having to reread all of the data deck.

The input commands recognized by VESYS are of 3 basic types:

1. LOGICAL - No data value is needed. The presence or absence of the command indicates which way a decision is to be made in the program.
2. SCALAR - One data value is read in the field immediately following the command. Absence of the command causes the default value, if any, to be assumed. Absence of a data value following the command will cause "zero" to be assumed, since blanks are read as zeroes.
3. ARRAY (or "Vector") - An array of several data values are read on the card(s) following this command. (No defaults exist for array values). No other commands or values can be on the same cards as an ARRAY COMMAND. The number of data values stored in the array is either determined by an associated SCALAR COMMAND.

#### 2.2.2 Formats

There are only two input formats; one for reading commands and the values associated with scalar commands, and another for reading the array values following an array command.

All commands, and the values with scalar commands, are read with FORMAT (4(A8,E12.4),

1	9	21	29	41	49	61	69	80
COMMAND	VALUE	COMMAND	VALUE	COMMAND	VALUE	COMMAND	VALUE	

After an array command, the subsequent card(s) contain the array values in this format: FORMAT (6E12.4), that is

1	13	25	39	49	61	72
VALUE	VALUE	VALUE	VALUE	VALUE	VALUE	

### 2.2.3 General Instructions and Suggestions for Command Use

1. All commands must begin in the first column of the command field. All commands consist of eight (8) characters, including trailing blanks. All 8 characters, including blanks, must be correct for a command to be recognized.
2. All data values must have a decimal point punched in the field. This includes integers.
3. Blank command fields are ignored. Blanks in data fields are read as zeros.
4. When an array (or vector) command is read, the data value field is ignored. Data values for the array are read from subsequent cards.
5. Each array command must appear on a card by itself. No other commands may appear on this card.
6. There are no default values for arrays.

#### 2.2.4 Deck Structure

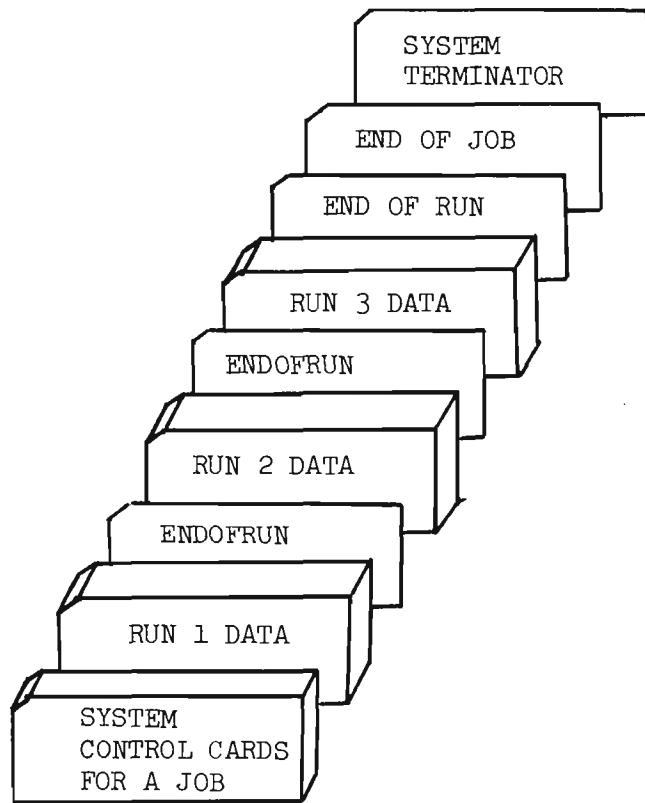


Figure 3. Example of a VESYS "job" with three "runs".

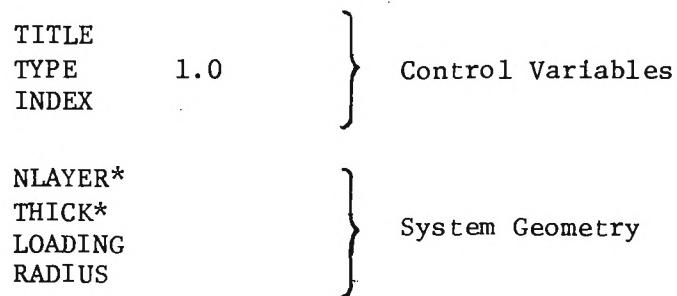
### 2.3 TYPES OF RUNS AVAILABLE

There are four types of runs available to the user of VESYS. The particular analysis desired is selected by specifying the appropriate value for the TYPE command:

<u>Value of TYPE</u>	<u>Analysis Performed</u>
1.0	Primary Response and Damage/Performance
2.0	Primary Response Only
3.0	Damage/Performance Only
4.0	Curve-Fitting Only

#### 2.3.1 Type 1 Analysis:

This calls for a full run which produces predictions of pavement response and expected lifetime on the basis of system geometry, materials characteristics, and environment. The primary response model passes primary response information to the damage model. Serviceability predictions are then made, based on the primary response information and the various environmental parameters. The following variables will ordinarily be supplied by the user for a TYPE 1 run:




---

\* Variables added in VESYS G. The meaning of these variables will be discussed in Section 2.4.

NZPOINTS*	{ }	Output Positions
NRPOINTS*		
ZPOINTS*	{ }	Creep Compliance
RPOINTS*		
ZCRACK		
NTSTATIC	{ }	
TSTATIC		
LAYER1*	{ }	
LAYER2*		
:		
LAYER7*	{ }	
POISSON*		
STRNCOEF	{ }	
STRNEXP		
COEFK1	{ }	
COEFK2		
K1K2CORL	{ }	
GNU		
ALPHA	{ }	
CORLOEF		
CORLEXP		
TOLERNCE	{ }	
QUALITYO		
STDEVO	{ }	
PSIFAIL		
NTRANDOM	{ }	
TUNITS		
TRANDOM	{ }	
LAMBDA		
AMPLITUD	{ }	
VCAMP		
DURATION	{ }	
VCDUR		
NTEMPS	{ }	
TEMPS		
REFTEMP	{ }	
BETA		
ENDOFRUN		Signals end of inputs, begins execution of run.

Materials Properties

Fatigue and Damage Variables

Serviceability Bounds

Traffic Variables

Temperature Variables

System Environment

The primary response information is computed in the primary response model and passed to the damage model.

Subsequent TYPE 3 runs (damage/performance only) may use the values which are generated for these variables. The advantage of this is that it

permits analysis of the same pavement under several different traffic, temperature and serviceability conditions without recomputing the primary response information. A considerable amount of computer time can be saved in this way, because the static load analysis requires a significant computational effort.

A sample input/output for Type 1 analysis is shown in Appendix 1.

#### 2.3.2 TYPE 2 Analysis:

This calls for primary response calculations only. It produces values for stresses, strains and deflections in a N-layered linear viscoelastic system.

#### 2.3.3 TYPE 3 Analysis:

It is recommended that TYPE 3 runs be made only after the primary response variables have been calculated in a TYPE 1 run. For a user to input all the required information requires a good understanding of the program. Primary response values can be passed directly from a TYPE 1 run to subsequent TYPE 3 runs of the same job. This is explained above for the TYPE 1 analysis.

#### 2.3.4 TYPE 4 Analysis:

This runs the curve-fitting routines on a set of data. The coefficients and DELTAS for a Dirichlet series which approximates the input curve are printed out. The approximation used is based on a least-squares fit. This type of analysis may be used in order to find the user input DELTA values which result in accurate curve fits for the creep compliance data from layer materials. Although VESYS will compute values

for DELTAS when they are not specifically input, a better set of curve fits may be obtained by a carefully chosen set of DELTAS.

#### 2.4 DIRECTORY OF NEW COMMANDS

The meaning of the new commands used in VESYS G that are not found in VESYS IIM will be explained in this section. The meaning of the other commands that are common for VESYS G and VESYS IIM were explained in the VESYS USER'S MANUAL cited before.

NLAYER	(Default 1.0) The number of layers in a pavement system including subgrade
THICK	(Array-no default) The array of layer thicknesses, excluding subgrade
NZPOINTS	(Default 1.0) The number of vertical positions for output
NRPOINTS	(Default 1.0) The number of radial positions for output
ZPOINTS	(Array - no default) The array of vertical positions from the surface at which the primary response information is desired
RPOINTS	(Array - no default) The array of radial positions from the center of the load at which the primary response information is desired.
LAYER1	(Array - no default)
LAYER2	.
⋮	.
LAYERX	The array of the master creep compliance curves, mean and coeff. of variation, for layer 1 to layer X where X equals to NLAYER. Mean value and the coeff. of variation of the creep compliance are read in as a pair. Thus, the first two numbers in the array of LAYER1 represent the mean of coeff. of variation of the creep compliance for layer 1 material at time corresponding to the first point in TSTATIC. Similarly the third and fourth numbers represent the mean and coeff. of variation at time corresponding to the second point in TSTATIC.

The array for each layer should consist of exactly two times NTSTATIC elements (maximum 2 X 25), and NTSTATIC must be input before any of these arrays. The format for the array is 6E12.4. See Appendix 1. Sample Input/Output for the Input of LAYER1 to LAYERX.

ZCRACK      (Default to depth of layer 1)  
Depth at which strain is obtained to determine cracking index.

POISSON      (Array - No Default)  
Poisson's Ratio for each layer, including subgrade.

Commands in VESYS IIM that are not used in VESYS G are:

ITYPES  
THICK1  
THICK2  
LAYER1  
LAYER2  
LAYER3  
VARCOEF1  
VARCOEF2  
VARCOEF3

## CHAPTER 3

## PROGRAM DOCUMENTATION

**3.1 Macro Program Structure**

The VESYS G package contains the following four major programs which are further divided into subroutines.

MAIN - handles all data input and stores it in a form suitable for use by the other subcomponents.

CURVIT - performs a least square curve-fitting technique using Dirichlet series.

PRIME - computes closed-form probabilistic solution, mean and deviation of the stress, strain and deflection at the prescribed positions of a n-layer viscoelastic pavement system.

RANDOM - computes the pavement response to a repeated load, the pavement distress (rutting, cracking and slope variance), and the pavement serviceability.

The macro flow chart of the overall program is shown in Fig. 4.

Since the major difference between VESYS IIM and VESYS G is in PRIME, the remaining part of this chapter will be concentrated on discussion of the PRIME program.

**3.2 PRIME Program**

The Primary Response Program consists of the following subroutines:

PRIME - reads in all input data.

NLAYER - computes closed-form solutions (mean and deviation) of stresses, strains and deflections.

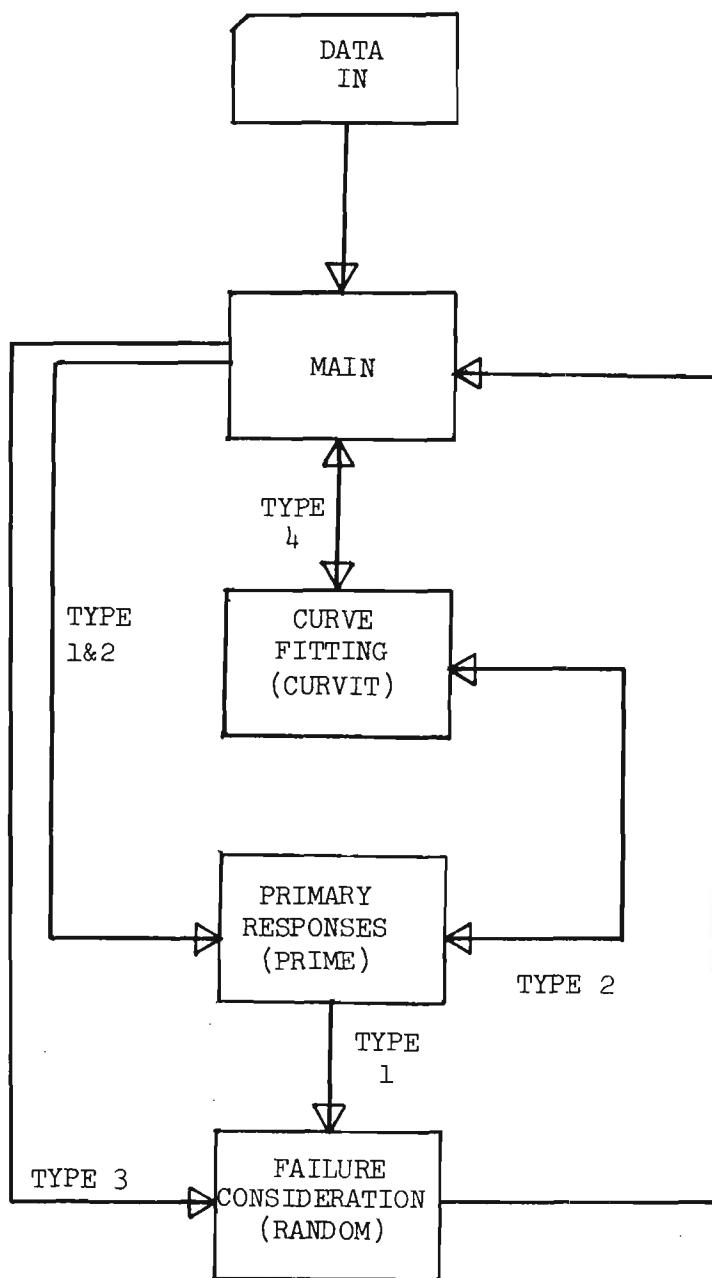


FIGURE 4. MACRO FLOW CHART OF VESYS G

PART - computes integration m; see eq. (9) of Appendix 3.

COEF - computes  $A_i$ ,  $B_i$ ,  $C_i$ ,  $D_i$ ,  $dA_i$ ,  $dB_i$ ,  $dC_i$ ,  $dD_i$ , see Eq. (5) and eq. (19).

TERMS - computes matrix  $X_i$  and  $\partial X_i / \partial E_j$ , see eq. (8) and (18)

DMULT - matrix multiplication

BESSEL - evaluates Bessel functions  $J_0$  and  $J_1$

FINTG1 - evaluates integration and performs deterministic calculation of stresses, strains, and deflection, see eq. (9) and (23).

FINTG2 - evaluates integration and performs probabilistic solution of deviations of stresses, strains and deflection, see eq. (24).

PROBL - evaluates variances, see eq. (12).

The interrelationship of these subroutine is illustrated in Fig. 5.

The main PRIME program acts essentially as a supervisor program handling all input and output operations. It also computes the permanent deformation and systems GNU and ALF which are to be used in RANDOM for pavement rut depth computation. A flow chart for this subroutine is shown in Fig. 6. The NLAYER subroutine is the main subroutine which computes the stresses, strains and deflections. A flow chart for this subroutine is shown in Fig. 7.

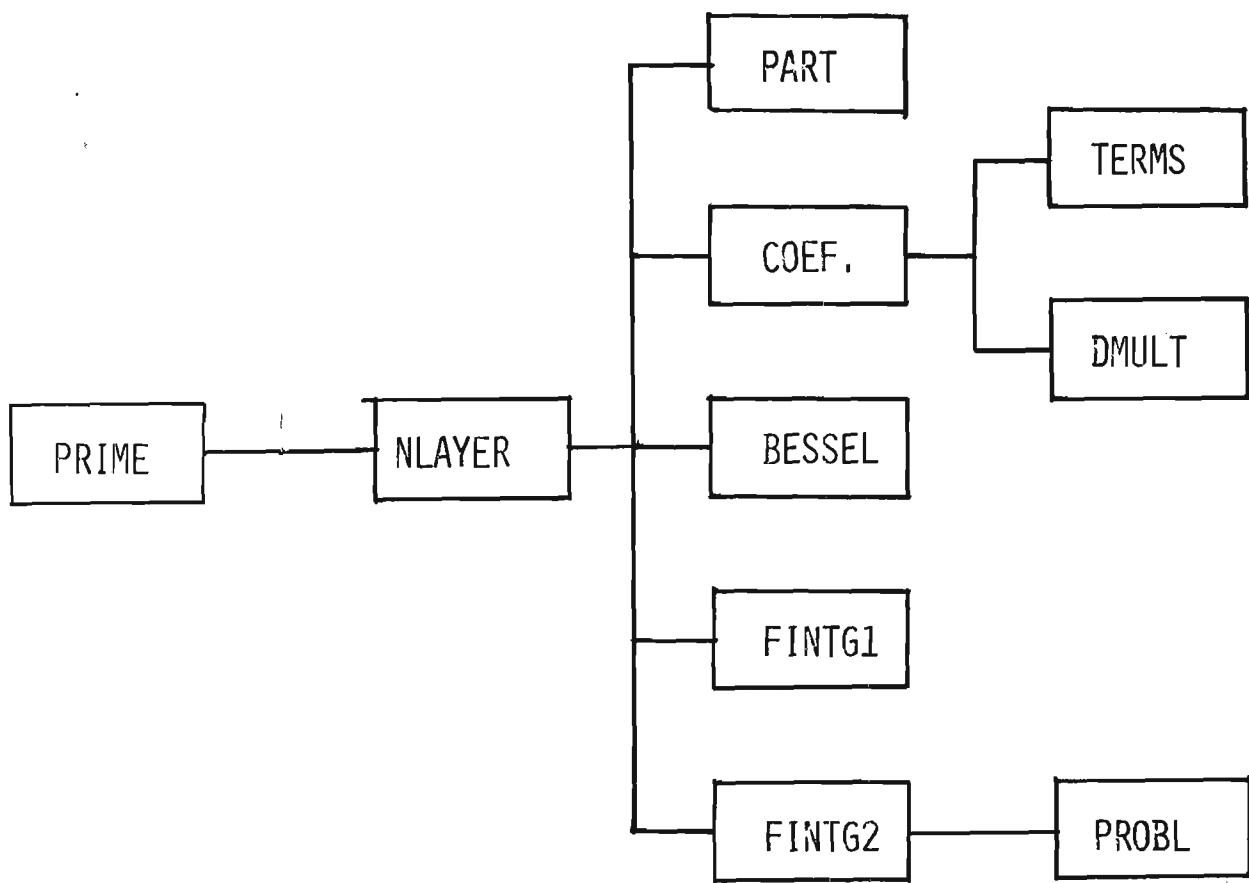


FIGURE 5. PROGRAM STRUCTURE OF PRIME

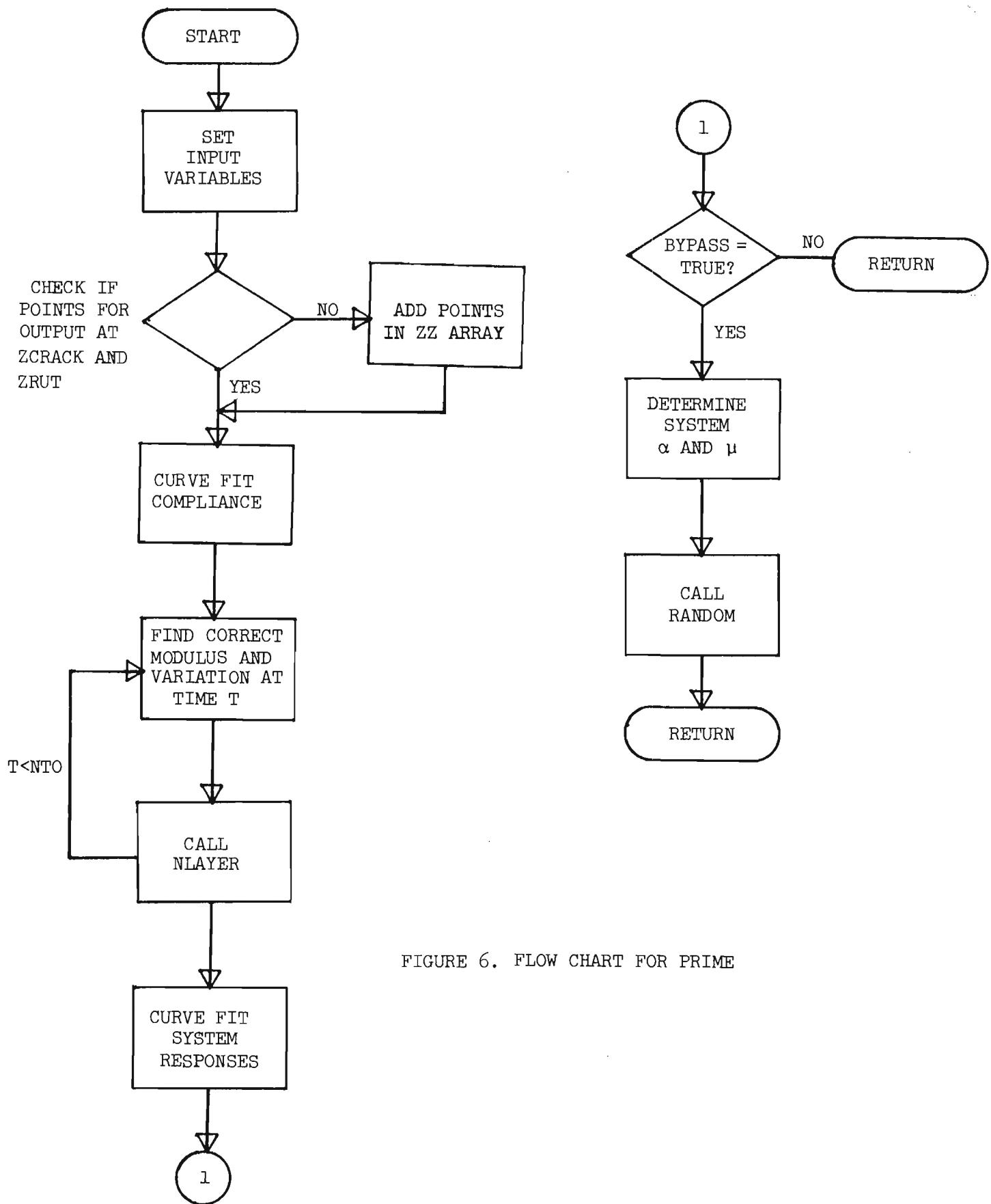


FIGURE 6. FLOW CHART FOR PRIME

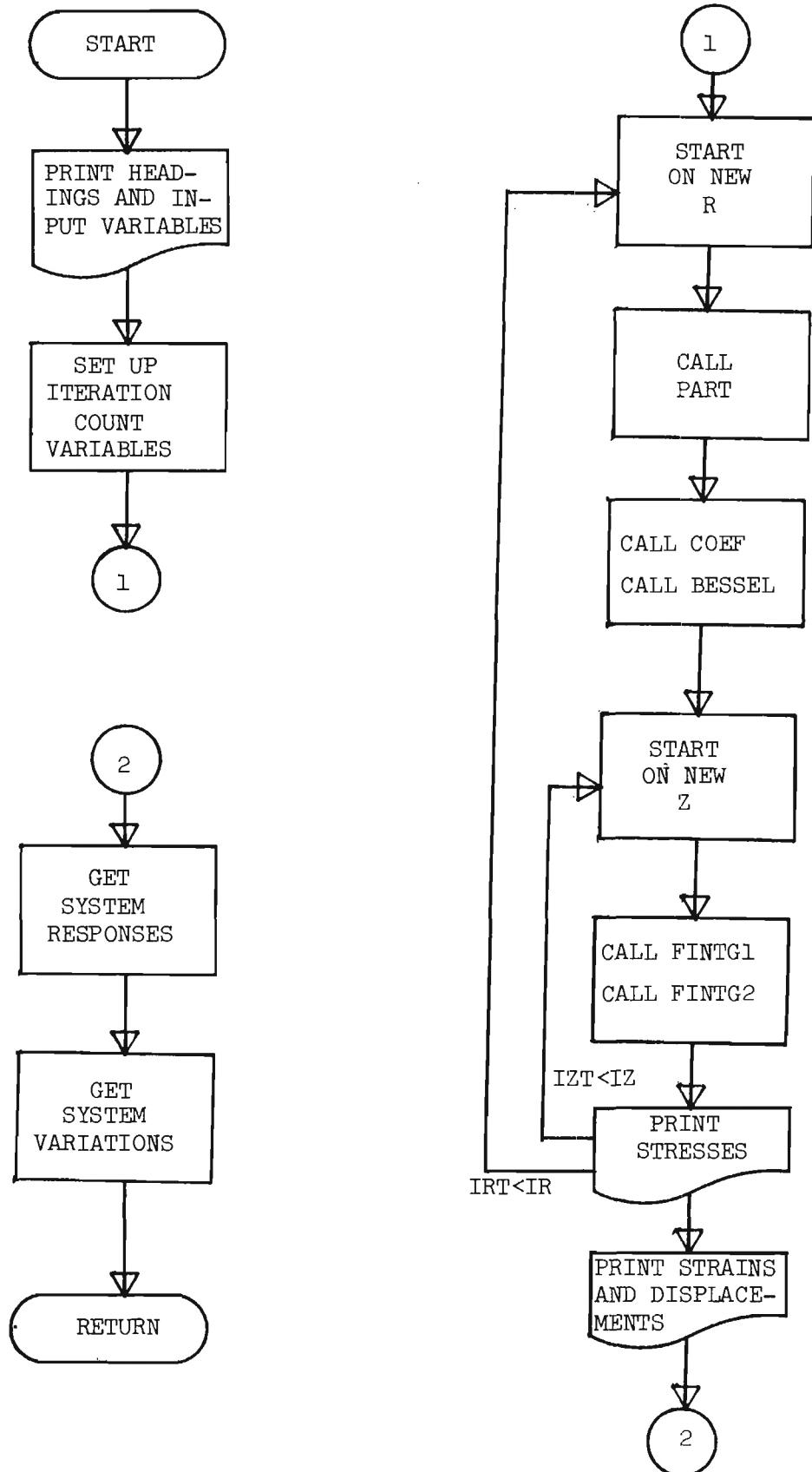


FIGURE 7. FLOW CHART FOR NLAYER

## CHAPTER 4

## SENSITIVITY ANALYSIS

Brief sensitivity analyses were performed on VESYS G. The objective of these analyses is to determine the accuracy of the system's response. Two types of sensitivity analyses were performed. The first type was to determine the accuracy of the VESYS G system's response as comparison with the current VESYS IIM system. The second type was to do a limited sensitivity analysis of the primary responses (mean and standard deviation) from the PRIME.

#### 4.1 Sensitivity Analysis of VESYS G

In order to compare the outputs from the VESYS G and the VESYS IIM, data from the original VESYS IIM design sample as included in the VESYS USER'S MANUAL was used as the inputs to the VESYS G program. The geometry of the pavement system for the design sample consists of 6 inches of surface course, and 8 inches of base course and a semi-infinite subgrade.

In the first run, the identical inputs from the VESYS IIM design samples were used. In the second run, the 6 inches surface course of the design sample was divided into two 3-inch layers and the 8 inch base course was divided into two 4-inch layers and the other inputs were the same. Thus, the second run could be considered as a 5-layer pavement system. The inputs and outputs of these two cases are included in Appendix 1 together with the outputs from the original VESYS IIM design sample.

The results from these three cases can be summarized as follows:

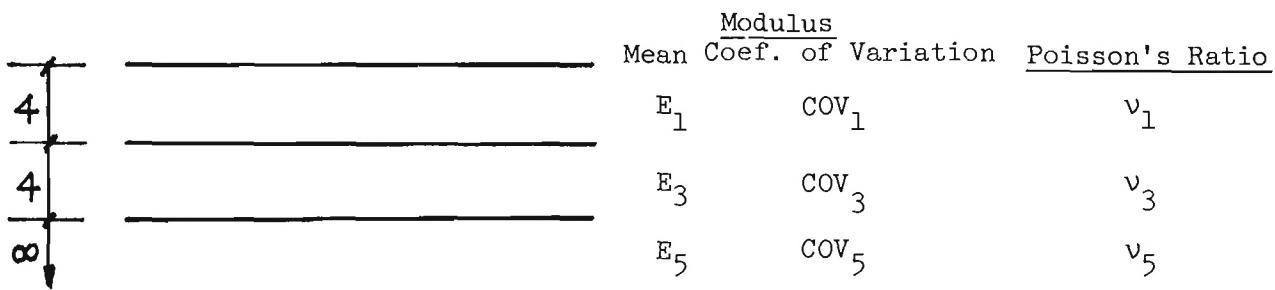
- (1) The outputs from the run 1 (3-layer) and run 2 (5-layer) are almost identical. The main differences are in the primary responses. This will be discussed in Section 4.2.
- (2) The primary responses between the VESYS G and VESYS IIM are very close. VESYS G tends to yield higher values (5% to 10% higher).
- (3) The rutting, cracking and slope variance predicted from VESYS G are very close to the values predicted by VESYS IIM.

#### 4.2 Sensitivity Analysis of PRIME

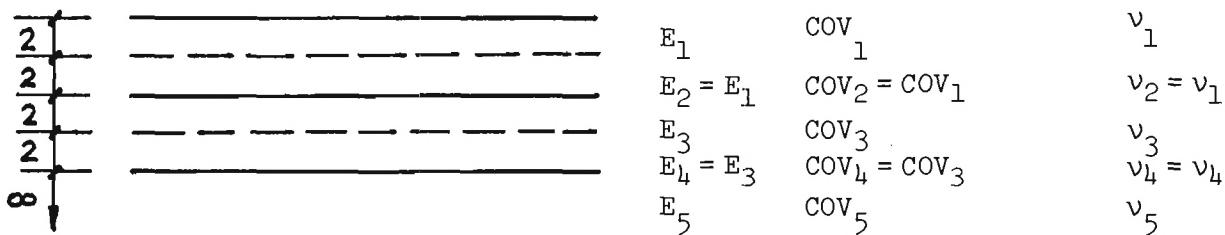
The objective of this part of sensitivity analysis is to determine the accuracy of the primary responses and the effect of the number of layers, and the probabilistic properties of layer materials on the probabilistic pavement responses. Four pavement systems as shown in Fig. 8 were used for the sensitivity analysis. The layer thickness and the material properties of each layer for these four pavement systems were so chosen such that Case B (5-layers), Case C (6-layers) and Case D (7-layers) are equivalent to Case A (3-layers). The additional layers other than the basic 3-layers can be considered as the imaginary ones. To determine the effect of the material variability on the pavement responses, different coefficients of variations (COV) of materials in each pavement layers were input to the program. The other input parameters including loading, and mean material properties were kept constant.

The input data for all cases other than the coefficient of variation are shown in the following:

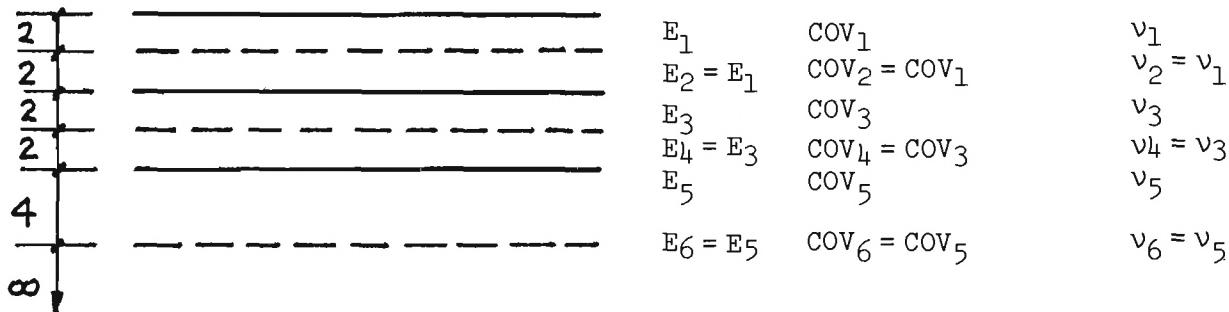
## CASE A



## CASE B



## CASE C



## CASE D

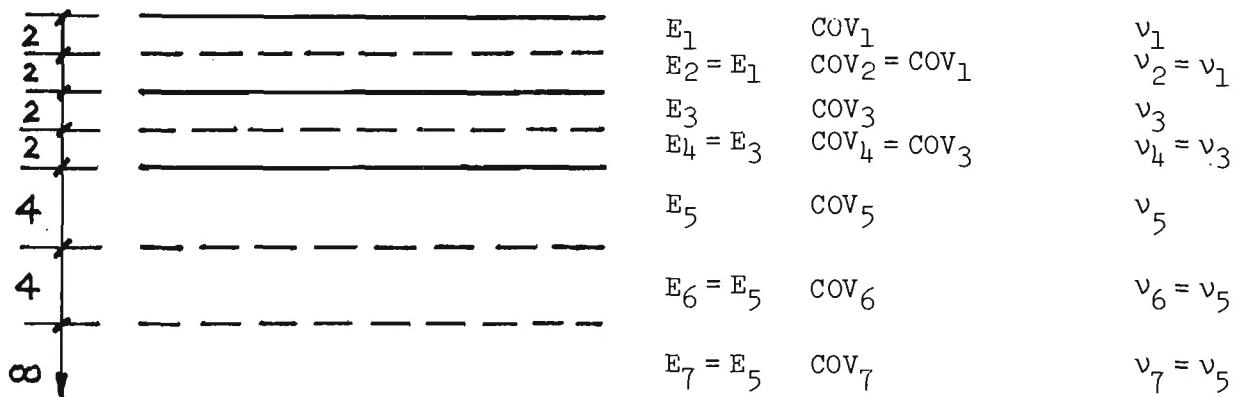


Figure 8. Pavement Systems for Sensitivity Analysis

Total Load (lbs.) = 5654.87

Tire Pressure (psi) = 50.00

Load Radius (in.) = 6.00

$E_1 = 27915$	$v_1 = 0.4$
$E_2 = 10,000$	$v_3 = 0.5$
$E_5 = 3225$	$v_5 = 0.5$

}

E's for different problem sets  
are slightly different than the  
values shown here. See  
printouts in Appendix II.

Four sample problem sets each with different combinations of coefficient of variation were run. The vertical surface deflections are summarized in Table 1. Detailed computer printout for all responses is included in Appendix II.

The expected value for surface deflections in all cases are almost the same. The standard deviations are significantly dependent on the input coefficients of variation (COV). For a given pavement system, for example Case A, the standard deviation of the surface deflection depends on the COV of all layer materials. Comparing the difference in the standard deviation between sample set 2 and 3 and sample set 3 and 4 from Table 1 indicates that COV in the subgrade has a more significant effect on the standard deviation of the surface deflection than the COV in the surface course.

On the other hand, results shown in Appendix II indicate that the standard deviations for the stresses and strains at a given position are most sensitive to the COV of the material occupying that position.

Another interesting finding worth to point out here is that for a given set of COV's, the mean responses, such as the surface deflection as shown in Table 1, are the same among four pavement systems. On the other hand, the standard deviations are different for different pavement

Table 1. Surface Deflection (Mean and Deviation)

Pavement System	Sample Problem 1		Sample Problem 2		Sample Problem 3		Sample Problem 4	
	Mean (inch)	Standard Deviation						
A	0.0785	0.00148	0.0783	0.00267	0.0767	0.00404	0.0733	0.00829
B	0.0785	0.00075	0.0783	0.00235	0.0767	0.00280	0.0733	0.00778
C	0.0785	0.00075	0.0783	0.00209	0.0767	0.00259	0.0733	0.00686
D	0.0785	0.00075	0.0783	0.00186	0.0767	0.00242	0.0733	0.00603

systems. Increasing number of layers, such as from Case A to Case D, will result in a smaller standard deviation for the surface deflection even though Case A-D are supposed to represent the same pavement system. This difference in the standard deviations can be attributed to the basic assumption made in deriving the closed-form probabilistic solution. In the original formulation of the solution, see Appendix 3, the properties of each layer were assumed to be independent. This assumption is reasonable for Case A in which the properties of the three different layers,  $E_1$ ,  $E_3$ , and  $E_5$  can be assumed mutually independent. On the other hand, for example in Case D,  $E_5$ ,  $E_6$  and  $E_7$  are actually the same layer, thus, the probabilistic properties of these three layers should be perfectly correlated instead of mutually independent. Therefore it is possible that the predicted standard deviation for Case D to be smaller than the standard deviation for Case A, due to the fact that some cross correlation terms which should be included in Case D were neglected.

CHAPTER 5  
RECOMMENDATIONS

It is not the intent of this chapter to discuss and recommend improvements for the predictability of VESYS system, although there are undoubtedly room for improvements in this respect. Rather, the recommendations presented herein are mainly for streamlining the operation of the VESYS G and for improving computer run time efficiency.

1. Determine minimum ITG (in subroutine NLAYER) currently 46 is used to reduce the computation time without sacrificing the accuracy of the outputs.
2. Reduce array dimensions to no more than necessary, and get rid of unused variables.
3. Inputting small tire pressures (PSI) and large modulus (small compliance) may cause overflow in subroutine TERMS. If the small values of PSI input is multiplied by a constant (say 50) and the output from that is divided by the same constant, it may help to avoid the overflow problem. This is especially helpful in IBM 360 systems, since range on such is very limited.
4. Output formats need improving.
5. Use more meaningful upper and lower bounds for input error check.

APPENDIX 1  
SAMPLE INPUT/OUTPUT

Sample Input - Run 1

Sample Output - Run 1

Sample Input - Run 2

Sample Output - Run 2

VESYS IIM Sample Problem Outputs

VESYS G SAMPLE PROBLEM 1

PROGRAM TITLE 3 LAYER INPUT    VERIFY 1/3    CHAR. SET  
 026  029  OTHER \_\_\_\_\_

NAME	CHARGE OR REFERENCE NUMBER	DATE			
DESIGN EXAMPLE	12-31-75				
TYPE					
LOADING	NT STATIC	N LAYER			
Z POINTS	2.	R POINTS	2.		
R POINTS					
0.0	6.0				
THICK					
6.0	8.0				
POSITION					
0.5	0.5	6.5			
Z POINTS					
0.0	6.0				
STATION					
0.0E-03	3.0E-03	1.0E-02	3.0E-02	1.0E-01	3.0E-01
1.0E-03	3.0E-03	1.0E-01	3.0E-01	1.0E-02	3.0E-02
LAYER					
3.7E-06	2.7E-06	1.52E-06	2.7E-06	8.6E-06	2.7E-06
1.45E-05	2.7E-05	2.5E-05	2.7E-05	4.0E-05	2.7E-05
6.2E-05	2.7E-05	8.6E-05	2.7E-05	1.1E-05	2.7E-05

PROGRAM TITLE 3 LAYER INPUT  VERIFY 2/3

148

2/3

CHAR. SET  
026 □ 0

02

5

023

9

911

18

7

**NAME** \_\_\_\_\_

— CHARGE OR REFERENCE NUMBER

- DATA

PROGRAM TITLE 3 LAYER INPUT  VERIFY 3/3  
NAME \_\_\_\_\_ CHARGE OR REFERENCE NUMBER \_\_\_\_\_  
CHAR. SET  
 028  079  OTHER \_\_\_\_\_  
DATE \_\_\_\_\_

1.4162E-04	1.728E-94	1.155E-103	1.727E-103	1.142E-102	1.121E-01
1.984E-01	1.293E-101	1.568E-102	1.932E-103	1.151E-03	1.657E-04
LAMPLN	1.105E-103	1.741E-103	1.044E-103	1.558E-103	1.044E-103
PROCUR	1.000E-01	1.000E-01	1.000E-01	1.000E-01	1.000E-01
INITIAL	1.6E-01	1.0E-01	1.3E-01	1.0E-01	1.0E-01
ALPHA	1.756E-01	1.055E-01	1.007E-01	1.000E-01	1.000E-01
GNV	1.311E-01	1.055E-01	1.007E-01	1.000E-01	1.000E-01
WTRENDOM	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
TRANSDOM	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
S	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
LAMBDA	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
NOELHAS	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
DELTA	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
ENDCIRUN	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01
ENDCJOB	1.0E-01	1.0E-01	1.0E-01	1.0E-01	1.0E-01

FIG. 4  
PREDICTIVE DESIGN PROCEDURES

VERB: A-AEROELASTIC PREDICTIVE MODEL  
(FLEXURE TESTER)

LAST REVISED: 11-10-76 TEC

DESIGN EXAMPLE: 11-11-76

INPUT DATA VALUES FOR TEC

>>> TYPE  
DATA

>>> L1010 = 100.0E+01

>>> R1010 = 64.0E+01

>>> L1010 = 100.0E+01

>>> R1010 = 64.0E+01

>>> L1010 = 100.0E+01

>>> R1010 = 64.0E+01

>>> Z1010 = 100.0E+01

>>> Z1010 = 64.0E+01

>>> M1010 = 100.0E+01

>>> T1010 = 100.0E+01

>>> D1010 = 100.0E+01

>>> E1010 = 100.0E+01

>>> ISTAT10 = 11

>>> ISTAT10 = 10000E+02

>>> ISTAT10 = 10000E+03

>>> ISTAT10 = 10000E+04

>>> ISTAT10 = 10000E+05

>>> ISTAT10 = 10000E+06

>>> ISTAT10 = 10000E+07

>>> ISTAT10 = 10000E+08

>>> ISTAT10 = 10000E+09

>>> ISTAT10 = 10000E+10

>>> ISTAT10 = 10000E+11

>>> ISTAT10 = 10000E+12

>>> ISTAT10 = 10000E+13

>>> ISTAT10 = 10000E+14

>>> ISTAT10 = 10000E+15

>>> ISTAT10 = 10000E+16

>>> ISTAT10 = 10000E+17

>>> ISTAT10 = 10000E+18

>>> ISTAT10 = 10000E+19

>>> ISTAT10 = 10000E+20





\*\*\*\*\* DESIGN EXAMPLE 12-31-11

LABORATORY DATA -- COFFEE CONCENTRATES

LAYER	TYPE	VALUE	VIBRATION
1	00E-02	0.37000E+00	0.27000E+00
2	00E-02	0.52000E+00	0.27000E+00
3	00E-02	0.85000E+00	0.27000E+00
4	00E-02	0.14500E+01	0.27000E+00
5	00E+00	0.25000E+01	0.27000E+00
6	00E+00	0.40000E+01	0.27000E+00
7	00E+00	0.62000E+01	0.27000E+00
8	00E+00	0.85000E+01	0.27000E+00
9	00E+00	0.12000E+02	0.27000E+00
10	00E+00	0.15500E+02	0.27000E+00
11	00E+00	0.13000E+02	0.27000E+00

111111 DATA      00000000000000000000000000000000

ENTERED DATA      DELETION      ERASE

LAYER	TIME	VALUE	VARIATION
1	1000E-02	0.22100E-04	0.32500E+00
1	1001E-02	0.22010E-04	0.32500E+00
1	1002E-02	0.22020E-04	0.32500E+00
1	1003E-02	0.22030E-04	0.32500E+00
1	1004E-02	0.22040E-04	0.32500E+00
1	1005E-02	0.22050E-04	0.32500E+00
1	1006E-02	0.22060E-04	0.32500E+00
1	1007E-02	0.22070E-04	0.32500E+00
1	1008E-02	0.22080E-04	0.32500E+00
1	1009E-02	0.22090E-04	0.32500E+00
1	1010E-02	0.22100E-04	0.32500E+00

..... DESIGN EXAMPLE ..... 12-31-79

PROFILE PARAMETERS

TOTAL LOAD (LBS) 120,000  
TIP PRESSURE (PSI) 1,000  
LAYER RADIUS (IN) 7.00  
LAYER 1 = INDOLES 400000 POISSON'S RATIO 0.20 THICKNESS (IN) 6.00  
LAYER 2 = NOLINS 400000 POISSON'S RATIO 0.20 THICKNESS (IN) 6.00  
LAYER 3 = XYLOLINE 400000 POISSON'S RATIO 0.20 SHOT-LINE(LAYER THICKNESS)

TIME = .01,000000 SECONDS

PROBABILISTIC PARAMETERS

LAYER 1=COEFFICIENT OF VARIATION OF MODULUS 0.70  
LAYER 2=COEFFICIENT OF VARIATION OF MODULUS 0.70  
LAYER 3=COEFFICIENT OF VARIATION OF MODULUS 0.70

STRESSES									
		VERTICAL	TANGENTIAL	RADIAL		SUPER			
		MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION
6.00	2.00	2.01E+01	1.187E-14	2.772E+01	1.177E-14	2.372E+01	1.438E-14	0.0	0.0
6.00	2.00	2.01E+01	1.187E-14	2.711E+01	1.148E-01	2.711E+01	1.541E-01	-1.03E-14	1.17E-14
6.00	2.00	2.01E+01	1.187E-14	2.690E+01	1.148E-01	2.690E+01	1.541E-01	-1.218E-14	1.67E-14

..... DESIGN EXAMPLE ..... 12-31-79

STRAIN COEFFICIENTS

		STRAINING			
		VERTICAL	RADIAL	RADIAL	TANGENTIAL
		MEAN	DEVIATION	MEAN	DEVIATION
6.00	1.00	1.02E-03	1.187E-03	0.0	1.187E-03
6.00	2.00	2.07E-03	1.187E-03	2.07E-03	1.187E-03

Digitized by srujanika@gmail.com

TOTAL LOAD /EPSI 128.68  
PERCENTAGE LOAD 100%  
LOAD DENSITY (L/D) 6.6

LAYER 1 = 0.0001 IN. 1007410. PRESSURE RATIO = 0.001 THICKNESS (IN.) = .000  
 LAYER 2 = 0.0001 IN. 1007410. PRESSURE RATIO = 0.001 THICKNESS (IN.) = .000  
 LAYER 3 = 0.0001 IN. 1007410. PRESSURE RATIO = 0.001 THICKNESS (IN.) = .000

**1111** = **85,100** - **82** = **85,018**

PARAPHRATIC LEXAMETICS

LAYER 1 = ECFE ECFE ECFE ECFE ECFE  
 LAYER 2 = ECFE ECFE ECFE ECFE ECFE  
 LAYER 3 = ECFE ECFE ECFE ECFE ECFE

		S T R E S S E S								
		V E R T I C A L			T A N G E N T I A L			R I G I D		S U P E R
R	Z	C O M P R E S S I O N		M E A N	E X V I T I O N		O F F S E T	C O M P R E S S I O N		E X V I T I O N
0.00	0.00	+0.12E+01	-0.55E-14	-0.34E+01	+0.24E+01	-0.34E+01	-0.27E+02	0.	0.	0.
0.00	0.00	+0.12E+01	-0.55E-14	-0.34E+01	+0.24E+01	-0.34E+01	-0.27E+02	0.	0.	0.
0.00	0.00	+0.12E+01	-0.55E-14	-0.34E+01	+0.24E+01	-0.34E+01	-0.27E+02	0.	0.	0.
0.00	0.00	+0.12E+01	-0.55E-14	-0.34E+01	+0.24E+01	-0.34E+01	-0.27E+02	0.	0.	0.

\*\*\*\*\* PRACTICE EXAMINEE 12-71-75

“ISLAMIC” FUND

		VERTICAL		FRONTAL		RADIAL		TANGENTIAL	
R	Z	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION
0.00	0.00	0.672E-04	0.183E-04	0.	0.	-0.618E-04	0.155E-04	-0.718E-06	0.166E-07
0.00	0.00	0.512E-04	0.144E-04	0.	0.	-0.790E-04	0.493E-04	-0.790E-06	0.316E-06
0.00	0.00	0.617E-04	0.135E-04	-0.717E-05	0.503E-06	-0.349E-06	0.415E-07	-0.529E-16	0.405E-17
0.00	0.00	0.417E-04	0.170E-04	-0.795E-05	0.752E-06	-0.416E-06	0.720E-07	-0.589E-16	0.128E-16

## \*\*\*\*\* DESIGN EXAMPLE \*\*\*\*\* 12-31-71

## PROBLEM PARAMETERS

TOTAL LOAD (KIPS) 120.00  
 TIE RATIO (EPR) 1.00  
 LAYER RATIO (LPR) 0.60

LAYER 1 = MODULUS 1144E+00 POISSON'S RATIO 0.00 THICKNESS (IN) 6.00  
 LAYER 2 = MODULUS 1144E+00 POISSON'S RATIO 0.00 THICKNESS (IN) 6.00  
 LAYER 3 = MODULUS 40000 POISSON'S RATIO 0.00 SEMI-INFINITE THICKNESS

TIME = .1000000 SECONDS

## PROBABILISTIC PARAMETERS

LAYER 1=COEFFICIENT OF VARIATION OF MODULUS 07.00  
 LAYER 2=COEFFICIENT OF VARIATION OF MODULUS 07.00  
 LAYER 3=COEFFICIENT OF VARIATION OF POISSON'S 07.00

		STRAIN				STRESS			
		VERTICAL DEVIATION		TANGENTIAL DEVIATION		RADIAL DEVIATION		SHEAR DEVIATION	
R	Z	DEPOT	DEVIATION	DEPOT	DEVIATION	DEPOT	DEVIATION	DEPOT	DEVIATION
6.00	2.000	2.010E+01	1.957E-14	2.020E+01	1.972E-01	2.000E+01	1.947E-02	0.0	0.0
6.00	6.000	2.000E+01	1.990E-14	2.045E+01	2.000E+00	2.022E+01	2.020E+00	1.413E-14	3.74E-16
6.00	6.000	2.010E+01	2.027E-01	2.054E+01	2.016E-01	2.011E+01	2.020E-01	1.423E-14	3.977E-16

## \*\*\*\*\* DESIGN EXAMPLE \*\*\*\*\* 12-31-71

## DISPLACEMENT

		STRAIN			
		RADIAL		TANGENTIAL	
R	Z	DEPOT	DEVIATION	DEPOT	DEVIATION
6.00	2.000	6.710E-04	1.167E-04	0.0	0.0
6.00	6.000	6.710E-04	1.161E-04	0.0	0.0
6.00	6.000	7.166E-04	1.151E-04	-1.431E-05	7.266E-06
6.00	6.000	7.166E-04	1.151E-04	-1.560E-05	9.944E-06

## \*\*\*\*\* DENTON EXAMPLE \*\*\*\*\*

12-71-71

## PROBLEM PARAMETERS

TOTAL LOAD (KIPS) 100.00  
 TIE PRESSURE (PSI) 1.00  
 LAYER RADIUS (IN) 0.00

LAYER 1 = MODULUS 60000, POISSONS RATIO 0.00 THICKNESS (IN) 6.00  
 LAYER 2 = MODULUS 60000, POISSONS RATIO 0.00 THICKNESS (IN) 1.00  
 LAYER 3 = MODULUS INFINITE, POISSONS RATIO 0.00 SEMI-INFINITE THICKNESS

TIME = .00000E+01 SECONDS

## DYNAMIC PLASTIC PARAMETERS

LAYER 1=COEFFICIENT OF VARIATION OF MODULUS 07.00  
 LAYER 2=COEFFICIENT OF VARIATION OF MODULUS 00.00  
 LAYER 3=COEFFICIENT OF VARIATION OF MODULUS 00.00

## STRAINS

		VERTICAL	TANGENTIAL	RADIAL		SHEAR	
#	Z	PEAK DEVIATION	MEAN DEVIATION	DEPARTURE DEVIATION	PEAK DEVIATION	MEAN DEVIATION	DEPARTURE DEVIATION
0.00	0.00	2.012E+1	0.169E+0	0.255E+1	0.200E+0	0.201E+01	0.424E+02
0.00	6.00	2.020E+1	0.420E+01	0.265E+1	0.440E+01	0.671E+01	0.114E+02
0.00	6.00	2.020E+1	0.724E+15	0.211E+01	0.211E+00	0.159E+01	0.197E+00
0.00	6.00	2.019E+03	0.247E+01	0.726E+01	0.210E+01	0.979E+01	0.212E+01

## \*\*\*\*\* DENTON EXAMPLE \*\*\*\*\*

12-71-71

## DISPLACEMENTS

## STRAINS

		DISPLACEMENT	ROTATIONAL	RADIAL	TRANSLATIONAL
#	Z	PEAK DEVIATION	MEAN DEVIATION	DEPARTURE DEVIATION	PEAK DEVIATION
0.00	0.00	2.020E+00	0.170E+00	0.00	-0.111E+00
0.00	6.00	2.020E+00	0.170E+04	0.00	-0.111E+00
0.00	6.00	2.020E+00	0.551E+00	0.266E+00	-0.577E+00
0.00	6.00	2.020E+00	0.756E+00	0.173E+00	-0.601E+00

.....

1941-1942 21-11-1942

Digitized by srujanika@gmail.com

$\text{Molar mass} = 100 \text{ g/mol}$

Call number 27  
Class number 201  
(201) 0217 101

Digitized by srujanika@gmail.com

• • • • • • • • • • • • • • •

## \*\*\*\*\* DESIGN EXAMPLE 12-71-71 \*\*\*\*\*

## PARAMETER PARAMETERS

TOTAL LOAD (LBS)	100.00
TIP DEFLECTION (INCH)	1.00
LAYER RADIUS (INCH)	6.00

LAYER 1 - MODULUS	247000	POISSON'S RATIO	.600	THICKNESS (IN)	6.00
LAYER 2 - MODULUS	51140	POISSON'S RATIO	.600	THICKNESS (IN)	8.00
LAYER 3 - MODULUS	44160	POISSON'S RATIO	.600	SUPER-INFINITE THICKNESS	

TIME = .03 SECONDS - CYCLES

## STRAIN RATE PARAMETERS

LAYER 1 COEFFICIENT OF VARIATION IN 1E MODULUS	27.00
LAYER 2 COEFFICIENT OF VARIATION IN 1E MODULUS	10.00

R	Z	STRAIN RATE			STRAIN RATE			STRAIN RATE		
		VERTICAL	RADIAL	TANGENTIAL	VERTICAL	RADIAL	TANGENTIAL	VERTICAL	RADIAL	TANGENTIAL
6.00	7.00	-1.00E-01	-1.97E-14	-1.79E-01	-1.15E-01	-1.76E-01	-5.95E-02	0.0	0.0	81.6%
6.00	6.00	-2.00E-02	-5.98E-16	-2.15E-01	-1.16E-01	-1.87E-01	-1.20E+00	-1.17E-14	-1.28E-15	81.6%

## \*\*\*\*\* DESIGN EXAMPLE 12-71-71 \*\*\*\*\*

## DISPLACEMENTS

## STRAINS

R	Z	DISPLACEMENT			STRAIN				
		VERTICAL	RADIAL	TANGENTIAL	VERTICAL	RADIAL	TANGENTIAL		
6.00	7.00	1.00E-01	1.96E-04	0.0	5.0	-1.17E-01	9.27E-06	-1.15E-05	3.91E-06
6.00	6.00	1.00E-02	1.87E-04	-1.79E-05	1.05E-02	-1.61E-02	1.43E-06	-1.24E-05	1.46E-06

.....  
.....  
.....

For the first time, the U.S. has issued a formal statement of its position on the issue.

DEPARTMENT OF DEFENSE

1. *Georgius Cyprianus* = 1000

199 (611) 2019-03-11  
11003 101004

S411126130.indd 1

12-18041 2700A 1.00 0000000000000000

## \*\*\*\*\* DESIGN EXAMPLE 12-71-71

## GEOTEXTILE PARAMETERS

TOTAL LOAD (kN) 120.00  
 TIP PRESSURE (kPa) 1.00  
 LAYER THICKNESS (cm) 7.00

LAYER 1 = SOILTYPE 11603: POREPRESSURE RATIO 0.00 THICKNESS (cm) 6.00  
 LAYER 2 = SOILTYPE 11604: POREPRESSURE RATIO 0.00 THICKNESS (cm) 6.00  
 LAYER 3 = SOILTYPE 11605: POREPRESSURE RATIO 0.00 TOTAL-DEFINITE THICKNESS

TIME = 0.000000E+00 SECONDS

## PLASTICITY PARAMETERS

LAYER 1=COEFFICIENT OF VARIATION OF MOULDR 17.00  
 LAYER 2=COEFFICIENT OF VARIATION OF MOULDR 19.00  
 LAYER 3=COEFFICIENT OF VARIATION OF MOULDR 15.00

## STRAINS

z	VERTICAL DEVIATION	TARGET TOTAL DEVIATION			TOTAL DEVIATION	SURFACE DEVIATION
		MEAN	DEVIATION	MIN		
0.00	0.00 0.00E+00	0.157E-01	0.430E-01	0.111E-01	0.158E-01	0.158E-01 0.00 0.00
6.00	6.00 0.00E+00	0.491E-01	0.121E-01	0.778E-01	0.827E-01	0.273E-01 0.161E-01 0.164E-01

## \*\*\*\*\* DESIGN EXAMPLE 12-71-71

## DISPLACEMENTS

## STRAINS

z	VERTICAL DEVIATION	RADIAL DEVIATION			TANGENTIAL DEVIATION	SURFACE DEVIATION
		MEAN	DEVIATION	MIN		
0.00	0.00 0.00E+00	0.231E-01	0.00 0.00	0.140E-01	0.912E-01 0.182E-05 0.122E-05	0.122E-05
6.00	6.00 0.00E+00	0.113E-01	0.187E-01 0.164E-01	0.164E-01 0.544E-01 0.217E-06 0.109E-05 0.315E-06	0.233E-07 0.262E-05 0.239E-06	

## \*\*\*\*\* DESIGN EXAMPLE \*\*\*\*\* 10-21-77

## SPECIFY PARAMETERS

TOTAL LOAD (KIPS) 100.00  
 TOTAL PRESSURE (PSI) 100.  
 LOAD PERTHUE (PSI) 100.  
 LAYER 1 = MODULUS 6E14, POISSON'S RATIO .300, THICKNESS (IN) .600  
 LAYER 2 = MODULUS 6E14, POISSON'S RATIO .300, THICKNESS (IN) .600  
 LAYER 3 = MODULUS 6E14, POISSON'S RATIO .300, CENTER-LINE FIBER THICKNESS

TIME = 0.000000000 SECONDS

## COMPUTED STRESS

LAYER 1=COEFFICIENT OF VARIATION OF MODULUS .07.  
 LAYER 2=COEFFICIENT OF VARIATION OF MODULUS .07.  
 LAYER 3=COEFFICIENT OF VARIATION OF MODULUS .07.

		VERTICAL	TRANSVERSE	RADIAL	SHEAR	
		DEVIATION	DEVIATION	DEVIATION	DEVIATION	MEAN DEVIATION
0.00	0.00	-6.172E-01	6.677E-15	6.117E+01	1.04E+01	-6.172E-01
6.00	6.00	-6.389E-01	6.737E-15	6.117E+01	1.05E+01	-6.389E-01

## \*\*\*\*\* DESIGN EXAMPLE \*\*\*\*\* 10-21-77

## DISPLACEMENTS

		VERTICAL	RADIAL	SHEAR	TRANSVERSE	ROTATIONAL
		DEVIATION	DEVIATION	DEVIATION	DEVIATION	DEVIATION
0.00	0.00	6.172E-01	6.677E-15	6.117E+01	1.04E+01	-6.172E-01
6.00	6.00	6.389E-01	6.737E-15	6.117E+01	1.05E+01	-6.389E-01

2 2 2 2 2 2 2 2 2 2 2 2 2

S. H. S. T. V.

www.nature.com/scientificreports/ | (2023) 13:103 | Article number: 103

3-10-13 7:38 A.M.

ANSWER: **10**

Digitized by srujanika@gmail.com

### Answers

1947-50 1948-50 1949-50 1950-52 1951-53 1952-54 1953-55 1954-56 1955-57 1956-58 1957-59 1958-60 1959-61 1960-62 1961-63 1962-64 1963-65 1964-66 1965-67 1966-68 1967-69 1968-70 1969-71 1970-72 1971-73 1972-74 1973-75 1974-76 1975-77 1976-78 1977-79 1978-80 1979-81 1980-82 1981-83 1982-84 1983-85 1984-86 1985-87 1986-88 1987-89 1988-90 1989-91 1990-92 1991-93 1992-94 1993-95 1994-96 1995-97 1996-98 1997-99 1998-2000 1999-2001 2000-2002 2001-2003 2002-2004 2003-2005 2004-2006 2005-2007 2006-2008 2007-2009 2008-2010 2009-2011 2010-2012 2011-2013 2012-2014 2013-2015 2014-2016 2015-2017 2016-2018 2017-2019 2018-2020 2019-2021 2020-2022 2021-2023 2022-2024 2023-2025 2024-2026 2025-2027 2026-2028 2027-2030

1.6.3  
16.1  
09.06.1  
16.1  
16.1  
09.06.1

سازمان اسناد و کتابخانه ملی

2010-01-21 2010-01-21 2010-01-21

2012 2013 2012 2013 2012 2013 2012 2013 2012 2013 2012 2013 2012 2013

.....  
.....  
.....  
.....  
.....

www.EasyEngineering.net

## SUMMARY AND CONCLUSION

663-2010-0077  
663-2010-0078  
663-2010-0079

Digitized by srujanika@gmail.com

101-101-21 374483 851036 0000000000000000

## \*\*\*\*\* CURVE-FITTED SYSTEM RESPONSES

DETERMINATIVE FACTOR FOR CRACKING IS RADIAL STRAIN AT 0.00 IN.

FITTED DATA	DELTA	G%
1.0000E+01	-0.21932E-01	
1.0000E+02	-0.25230E-02	
1.0000E+01	-0.25398E-01	
1.0000E+01	-0.25433E-01	
1.0000E+01	-0.25439E-01	
1.0000E+01	-0.25439E-01	
1.0000E+01	-0.25439E-01	

DETERMINATIVE FACTOR FOR CRACKING IS VERTICAL DISPLACEMENT AT 0.00 IN.

FITTED DATA	DELTA	G%
1.0000E+01	-0.26010E-01	
1.0000E+02	-0.14108E-01	
1.0000E+01	-0.25394E-01	
1.0000E+01	-0.25432E-01	
1.0000E+01	-0.25435E-01	
1.0000E+01	-0.25435E-01	
1.0000E+01	-0.25435E-01	

EXPERIMENTAL DEFORMATION  
SYSTEM RESPONSE FACTORS  
GIVEN BY: 1.24297E-03  
ALPHA E: 1.24297E-03

## RADIAL STRAIN CRACKING

RADIAL STRAIN PREDICTION  
DEFINITION OF CRACKING STRAIN

TEMPERATURE	STRAIN	MAP STRAIN	MAP	MAP STRAIN	K1	K2	GMAP
DEGREES-E	IN./IN.	IN./IN. X 10^10	CYCLES	CYCLES X 10^10	CYCLES	DIMENSIONLESS	DIMENSIONLESS
49.75	-0.274E-01	-0.77386E-10	-0.10725E+00	-0.35041E+17	-0.62020E-04	-0.26202E+01	-0.15285E+00
53.25	-0.2428E-01	-0.47414E-10	-0.10725E+00	-0.64161E+16	-0.72020E-04	-0.26150E+01	-0.12650E+01
57.50	-0.212E-01	-0.131E-10	-0.10725E+00	-0.1772E+17	-0.16020E-04	-0.67100E+01	-0.11271E+01
61.75	-0.1876E-01	-0.212E-10	-0.10725E+00	-0.17744E+17	-0.76022E-03	-0.26402E+01	-0.36610E+01
66.00	-0.1632E-01	-0.297E-10	-0.10725E+00	-0.21507E+18	-0.24202E-03	-0.26100E+01	-0.24605E+02
70.25	-0.1388E-01	-0.382E-10	-0.10725E+00	-0.25066E+18	-0.12801E-03	-0.26100E+01	-0.44745E+02
74.50	-0.1144E-01	-0.467E-10	-0.10725E+00	-0.28401E+18	-0.26402E-03	-0.26202E+01	-0.26202E+02
78.75	-0.1107E-01	-0.552E-10	-0.10725E+00	-0.79477E+18	-0.29302E-03	-0.26202E+01	-0.45125E+02
83.00	-0.1073E-01	-0.637E-10	-0.10725E+00	-0.80670E+17	-0.85022E-02	-0.26202E+01	-0.12132E+02
87.25	-0.1040E-01	-0.722E-10	-0.10725E+00	-0.1444E+17	-0.32022E-02	-0.26402E+01	-0.12714E+01
91.50	-0.1007E-01	-0.776E-10	-0.10725E+00	-0.12467E+17	-0.15102E-03	-0.26202E+01	-0.10154E+01
95.75	-0.974E-01	-0.77174E-10	-0.16571E+00	-0.65701E+04	-0.65701E-04	-0.26202E+01	-0.99477E+00

PRIMARY LOAD	MAP DAMAGE INDEX	AREA CRACKED	TIME
DIMENSIONLESS	DIMENSIONLESS	IN.^2/IN.^2	YEARS
0.7006E-02	-0.13720E-01	0.0	0.0
0.44728E-02	-0.15772E-01	0.0	1.0
0.71125E-02	-0.26907E-01	0.0	1.4
0.75082E-02	-0.15671E-01	0.0	2.0
0.14640E-01	-0.73321E-02	0.0	2.6
0.12955E-01	-0.17202E-01	0.0	3.0
0.82557E-01	-0.10161E-01	0.0	3.5
0.71779E-01	-0.15724E-01	0.0	12.0
0.61647E-01	-0.49881E-01	0.0	16.0
0.17416E-01	-0.74621E-01	0.0	20.0

卷之三

卷之三

प्राचीन विद्या के लिए अत्यधिक उत्तम विद्यालय है।

YEAR	TIME	YEN CLASSIC INVESTMENT PLANTATION (Year 10)
1	1.00	1.00
2	1.05	1.10
3	1.10	1.15
4	1.15	1.20
5	1.20	1.25
6	1.25	1.30
7	1.30	1.35
8	1.35	1.40
9	1.40	1.45
10	1.45	1.50
11	1.50	1.55
12	1.55	1.60
13	1.60	1.65
14	1.65	1.70
15	1.70	1.75
16	1.75	1.80
17	1.80	1.85
18	1.85	1.90
19	1.90	1.95
20	1.95	2.00
21	2.00	2.05
22	2.05	2.10
23	2.10	2.15
24	2.15	2.20
25	2.20	2.25
26	2.25	2.30
27	2.30	2.35
28	2.35	2.40
29	2.40	2.45
30	2.45	2.50
31	2.50	2.55
32	2.55	2.60
33	2.60	2.65
34	2.65	2.70
35	2.70	2.75
36	2.75	2.80
37	2.80	2.85
38	2.85	2.90
39	2.90	2.95
40	2.95	3.00
41	3.00	3.05
42	3.05	3.10
43	3.10	3.15
44	3.15	3.20
45	3.20	3.25
46	3.25	3.30
47	3.30	3.35
48	3.35	3.40
49	3.40	3.45
50	3.45	3.50
51	3.50	3.55
52	3.55	3.60
53	3.60	3.65
54	3.65	3.70
55	3.70	3.75
56	3.75	3.80
57	3.80	3.85
58	3.85	3.90
59	3.90	3.95
60	3.95	4.00
61	4.00	4.05
62	4.05	4.10
63	4.10	4.15
64	4.15	4.20
65	4.20	4.25
66	4.25	4.30
67	4.30	4.35
68	4.35	4.40
69	4.40	4.45
70	4.45	4.50
71	4.50	4.55
72	4.55	4.60
73	4.60	4.65
74	4.65	4.70
75	4.70	4.75
76	4.75	4.80
77	4.80	4.85
78	4.85	4.90
79	4.90	4.95
80	4.95	5.00
81	5.00	5.05
82	5.05	5.10
83	5.10	5.15
84	5.15	5.20
85	5.20	5.25
86	5.25	5.30
87	5.30	5.35
88	5.35	5.40
89	5.40	5.45
90	5.45	5.50
91	5.50	5.55
92	5.55	5.60
93	5.60	5.65
94	5.65	5.70
95	5.70	5.75
96	5.75	5.80
97	5.80	5.85
98	5.85	5.90
99	5.90	5.95
00	5.95	6.00
01	6.00	6.05
02	6.05	6.10
03	6.10	6.15
04	6.15	6.20
05	6.20	6.25
06	6.25	6.30
07	6.30	6.35
08	6.35	6.40
09	6.40	6.45
10	6.45	6.50
11	6.50	6.55
12	6.55	6.60
13	6.60	6.65
14	6.65	6.70
15	6.70	6.75
16	6.75	6.80
17	6.80	6.85
18	6.85	6.90
19	6.90	6.95
20	6.95	7.00
21	7.00	7.05
22	7.05	7.10
23	7.10	7.15
24	7.15	7.20
25	7.20	7.25
26	7.25	7.30
27	7.30	7.35
28	7.35	7.40
29	7.40	7.45
30	7.45	7.50
31	7.50	7.55
32	7.55	7.60
33	7.60	7.65
34	7.65	7.70
35	7.70	7.75
36	7.75	7.80
37	7.80	7.85
38	7.85	7.90
39	7.90	7.95
40	7.95	8.00
41	8.00	8.05
42	8.05	8.10
43	8.10	8.15
44	8.15	8.20
45	8.20	8.25
46	8.25	8.30
47	8.30	8.35
48	8.35	8.40
49	8.40	8.45
50	8.45	8.50
51	8.50	8.55
52	8.55	8.60
53	8.60	8.65
54	8.65	8.70
55	8.70	8.75
56	8.75	8.80
57	8.80	8.85
58	8.85	8.90
59	8.90	8.95
60	8.95	9.00
61	9.00	9.05
62	9.05	9.10
63	9.10	9.15
64	9.15	9.20
65	9.20	9.25
66	9.25	9.30
67	9.30	9.35
68	9.35	9.40
69	9.40	9.45
70	9.45	9.50
71	9.50	9.55
72	9.55	9.60
73	9.60	9.65
74	9.65	9.70
75	9.70	9.75
76	9.75	9.80
77	9.80	9.85
78	9.85	9.90
79	9.90	9.95
80	9.95	10.00
81	10.00	10.05
82	10.05	10.10
83	10.10	10.15
84	10.15	10.20
85	10.20	10.25
86	10.25	10.30
87	10.30	10.35
88	10.35	10.40
89	10.40	10.45
90	10.45	10.50
91	10.50	10.55
92	10.55	10.60
93	10.60	10.65
94	10.65	10.70
95	10.70	10.75
96	10.75	10.80
97	10.80	10.85
98	10.85	10.90
99	10.90	10.95
00	10.95	11.00
01	11.00	11.05
02	11.05	11.10
03	11.10	11.15
04	11.15	11.20
05	11.20	11.25
06	11.25	11.30
07	11.30	11.35
08	11.35	11.40
09	11.40	11.45
10	11.45	11.50
11	11.50	11.55
12	11.55	11.60
13	11.60	11.65
14	11.65	11.70
15	11.70	11.75
16	11.75	11.80
17	11.80	11.85
18	11.85	11.90
19	11.90	11.95
20	11.95	12.00
21	12.00	12.05
22	12.05	12.10
23	12.10	12.15
24	12.15	12.20
25	12.20	12.25
26	12.25	12.30
27	12.30	12.35
28	12.35	12.40
29	12.40	12.45
30	12.45	12.50
31	12.50	12.55
32	12.55	12.60
33	12.60	12.65
34	12.65	12.70
35	12.70	12.75
36	12.75	12.80
37	12.80	12.85
38	12.85	12.90
39	12.90	12.95
40	12.95	13.00
41	13.00	13.05
42	13.05	13.10
43	13.10	13.15
44	13.15	13.20
45	13.20	13.25
46	13.25	13.30
47	13.30	13.35
48	13.35	13.40
49	13.40	13.45
50	13.45	13.50
51	13.50	13.55
52	13.55	13.60
53	13.60	13.65
54	13.65	13.70
55	13.70	13.75
56	13.75	13.80
57	13.80	13.85
58	13.85	13.90
59	13.90	13.95
60	13.95	14.00
61	14.00	14.05
62	14.05	14.10
63	14.10	14.15
64	14.15	14.20
65	14.20	14.25
66	14.25	14.30
67	14.30	14.35
68	14.35	14.40
69	14.40	14.45
70	14.45	14.50
71	14.50	14.55
72	14.55	14.60
73	14.60	14.65
74	14.65	14.70
75	14.70	14.75
76	14.75	14.80
77	14.80	14.85
78	14.85	14.90
79	14.90	14.95
80	14.95	15.00
81	15.00	15.05
82	15.05	15.10
83	15.10	15.15
84	15.15	15.20
85	15.20	15.25
86	15.25	15.30
87	15.30	15.35
88	15.35	15.40
89	15.40	15.45
90	15.45	15.50
91	15.50	15.55
92	15.55	15.60
93	15.60	15.65
94	15.65	15.70
95	15.70	15.75
96	15.75	15.80
97	15.80	15.85
98	15.85	15.90
99	15.90	15.95
00	15.95	16.00
01	16.00	16.05
02	16.05	16.10
03	16.10	16.15
04	16.15	16.20
05	16.20	16.25
06	16.25	16.30
07	16.30	16.35
08	16.35	16.40
09	16.40	16.45
10	16.45	16.50
11	16.50	16.55
12	16.55	16.60
13	16.60	16.65
14	16.65	16.70
15	16.70	16.75
16	16.75	16.80
17	16.80	16.85
18	16.85	16.90
19	16.90	16.95
20	16.95	17.00
21	17.00	17.05
22	17.05	17.10
23	17.10	17.15
24	17.15	17.20
25	17.20	17.25
26	17.25	17.30
27	17.30	17.35
28	17.35	17.40
29	17.40	17.45
30	17.45	17.50
31	17.50	17.55
32	17.55	17.60
33	17.60	17.65
34	17.65	17.70
35	17.70	17.75
36	17.75	17.80
37	17.80	17.85
38	17.85	17.90
39	17.90	17.95
40	17.95	18.00
41	18.00	18.05
42	18.05	18.10
43	18.10	18.15
44	18.15	18.20
45	18.20	18.25
46	18.25	18.30
47	18.30	18.35
48	18.35	18.40
49	18.40	18.45
50	18.45	18.50
51	18.50	18.55
52	18.55	18.60
53	18.60	18.65
54	18.65	18.70
55	18.70	18.75
56	18.75	18.80
57	18.80	18.85
58	18.85	18.90
59	18.90	18.95
60	18.95	19.00
61	19.00	19.05
62	19.05	19.10
63	19.10	19.15
64	19.15	19.20
65	19.20	19.25
66	19.25	19.30
67	19.30	19.35
68	19.35	19.40
69	19.40	19.45
70	19.45	19.50
71	19.50	19.55
72	19.55	19.60
73	19.60	19.65
74	19.65	19.70
75	19.70	19.75
76	19.75	19.80
77	19.80	19.85
78	19.85	19.90
79	19.90	19.95
80	19.95	20.00
81	20.00	20.05
82	20.05	20.10
83	20.10	20.15
84	20.15	20.20
85	20.20	20.25
86	20.25	20.30
87	20.30	20.35
88	20.35	20.40
89	20.40	20.45
90	20.45	20.50
91	20.50	20.55
92	20.55	20.60
93	20.60	20.65
94	20.65	20.70
95	20.70	20.75
96	20.75	20.80
97	20.80	20.85
98	20.85	20.90
99	20.90	20.95
00	20.95	21.00
01	21.00	21.05
02	21.05	21.1

## PERFORMANCE

RUT - EPT	VAD RUT - EPTV	SLOPE VAD	VAD SLOPE VAD	CRACKING	VAD CRACKING	TIME
0.00000E+00	0.14700E+00	0.21000E+00	0.22100E+01	0.21000E+00	0.14700E+00	0.00
0.11700E+01	0.17400E+01	0.22100E+02	0.27100E+01	0.24700E+02	0.15772E+02	1.00
0.11700E+02	0.21100E+02	0.48700E+02	0.58100E+02	0.52100E+02	0.26720E+02	1.00
0.11700E+03	0.25100E+03	0.61970E+03	0.71170E+03	0.60458E+03	0.31147E+03	2.00
0.11700E+04	0.31100E+04	0.75000E+04	0.87170E+04	0.81419E+04	0.47320E+04	4.00
0.11700E+05	0.39100E+05	0.87000E+05	0.99170E+05	0.91786E+05	0.63700E+05	4.00
0.11700E+06	0.49100E+06	0.97000E+06	0.11200E+06	0.96256E+06	0.16100E+06	8.00
0.11700E+07	0.61700E+07	0.11200E+07	0.12600E+07	0.11719E+07	0.20772E+07	12.00
0.11700E+08	0.77400E+08	0.14676E+08	0.15441E+08	0.14914E+08	0.38916E+08	16.00
0.11700E+09	0.97100E+09	0.17776E+09	0.17500E+09	0.17410E+09	0.74002E+09	22.00

SERVICEABILITY	VARIANCE OF SERVICEABILITY	TIME
0.00000E+00	0.00000E+00	0.00
0.14700E+00	0.00000E+00	0.50
0.21100E+01	0.00000E+00	1.00
0.25100E+02	0.00000E+00	1.00
0.31100E+03	0.00000E+00	2.00
0.39100E+04	0.00000E+00	2.00
0.49100E+05	0.00000E+00	4.00
0.61700E+06	0.00000E+00	4.00
0.77400E+07	0.00000E+00	5.00
0.97100E+08	0.00000E+00	12.00
0.97100E+09	0.00000E+00	16.00
0.97100E+09	0.22700E+00	22.00

SERVICEABILITY FAILURE LEVEL IS 0.00000E+00

## MARGINAL STATE RELIABILITIES

STATE UP/DOWN 1 = 0.15471 0.674620 0.646772 0.617407 0.565560 0.62444 0.74333 0.66222 0.78111 2.65000

STATE DOWN/UP 2 = 0.174449 0.667722 0.611607 0.591556 0.605494 0.74333 0.66222 0.78111 2.650000 -0.15451

STATE1	STATE2	STATE3	STATE4	STATE5	STATE6	STATE7	STATE8	STATE9	STATE10	TIME	RELIABILITY
0.16000E+00	0.14977	0.10878	0.11620	0.11747	0.21117	0.66100	0.66000	0.66100	0.66100	0.0	1.00000
0.16010E+00	0.15197	0.12015	0.12648	0.12272	0.22174	0.66112	0.66049	0.66112	0.66112	0.5	0.99999
0.16011E+00	0.15215	0.12049	0.12682	0.12307	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99998
0.16012E+00	0.15233	0.12084	0.12713	0.12337	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99997
0.16013E+00	0.15252	0.12120	0.12744	0.12367	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99996
0.16014E+00	0.15271	0.12156	0.12775	0.12397	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99995
0.16015E+00	0.15290	0.12193	0.12806	0.12427	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99994
0.16016E+00	0.15309	0.12229	0.12837	0.12457	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99993
0.16017E+00	0.15327	0.12265	0.12868	0.12487	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99992
0.16018E+00	0.15346	0.12302	0.12900	0.12517	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99991
0.16019E+00	0.15364	0.12337	0.12931	0.12547	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99990
0.16020E+00	0.15383	0.12373	0.12962	0.12577	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99989
0.16021E+00	0.15401	0.12409	0.12993	0.12607	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99988
0.16022E+00	0.15419	0.12445	0.13024	0.12637	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99987
0.16023E+00	0.15437	0.12481	0.13055	0.12667	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99986
0.16024E+00	0.15455	0.12517	0.13086	0.12697	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99985
0.16025E+00	0.15473	0.12553	0.13117	0.12727	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99984
0.16026E+00	0.15491	0.12589	0.13148	0.12757	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99983
0.16027E+00	0.15509	0.12625	0.13179	0.12787	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99982
0.16028E+00	0.15527	0.12661	0.13210	0.12817	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99981
0.16029E+00	0.15545	0.12697	0.13241	0.12847	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99980
0.16030E+00	0.15563	0.12733	0.13272	0.12877	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99979
0.16031E+00	0.15581	0.12769	0.13303	0.12907	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99978
0.16032E+00	0.15600	0.12805	0.13334	0.12937	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99977
0.16033E+00	0.15618	0.12841	0.13365	0.12967	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99976
0.16034E+00	0.15636	0.12877	0.13396	0.12997	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99975
0.16035E+00	0.15654	0.12913	0.13427	0.13027	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99974
0.16036E+00	0.15672	0.12949	0.13458	0.13057	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99973
0.16037E+00	0.15690	0.12985	0.13489	0.13087	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99972
0.16038E+00	0.15708	0.13021	0.13520	0.13117	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99971
0.16039E+00	0.15726	0.13056	0.13551	0.13147	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99970
0.16040E+00	0.15744	0.13092	0.13582	0.13177	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99969
0.16041E+00	0.15762	0.13128	0.13613	0.13207	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99968
0.16042E+00	0.15780	0.13164	0.13644	0.13237	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99967
0.16043E+00	0.15798	0.13200	0.13675	0.13267	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99966
0.16044E+00	0.15816	0.13236	0.13706	0.13297	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99965
0.16045E+00	0.15834	0.13272	0.13737	0.13327	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99964
0.16046E+00	0.15852	0.13308	0.13768	0.13357	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99963
0.16047E+00	0.15870	0.13344	0.13800	0.13387	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99962
0.16048E+00	0.15888	0.13380	0.13830	0.13417	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99961
0.16049E+00	0.15906	0.13416	0.13861	0.13447	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99960
0.16050E+00	0.15924	0.13452	0.13892	0.13477	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99959
0.16051E+00	0.15942	0.13488	0.13923	0.13507	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99958
0.16052E+00	0.15960	0.13524	0.13954	0.13537	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99957
0.16053E+00	0.15978	0.13560	0.13985	0.13567	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99956
0.16054E+00	0.15996	0.13596	0.14015	0.13597	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99955
0.16055E+00	0.16014	0.13632	0.14046	0.13627	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99954
0.16056E+00	0.16032	0.13668	0.14076	0.13657	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99953
0.16057E+00	0.16050	0.13704	0.14107	0.13687	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99952
0.16058E+00	0.16068	0.13740	0.14138	0.13717	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99951
0.16059E+00	0.16086	0.13776	0.14168	0.13747	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99950
0.16060E+00	0.16104	0.13812	0.14200	0.13777	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99949
0.16061E+00	0.16122	0.13848	0.14230	0.13807	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99948
0.16062E+00	0.16140	0.13884	0.14261	0.13837	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99947
0.16063E+00	0.16158	0.13920	0.14291	0.13867	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99946
0.16064E+00	0.16176	0.13956	0.14322	0.13897	0.22174	0.66112	0.66049	0.66112	0.66112	1.0	0.99945
0.16065E+00	0.16194	0.13992	0.14352								

## VESYS G SAMPLE PROBLEM 2

PROGRAM TITLE 5 LAYER INPUT  VERIFY  
NAME \_\_\_\_\_ CHARGE OR REFERENCE NUMBER \_\_\_\_\_  
CHAR. SET  
026  029  OTHER   
DESIGN EXAMPLE 2-3-7ST  
TYPE  
LOADING 1.0 STATIC 1.1 WIRES 1.5 WIRE POINTS 1.6  
MAPLOT 1.7ST 2.0  
THICK  
3.0 3.0 4.0 4.0  
POSITION  
5.0 6.0 7.0 8.0 9.0  
MARK 10.0  
  
REMAINING CARDS SAME AS IN 3 LAYER CASES  
EXCEPT LAYER PROPERTIES ARE DUPLICATED  
AND THERE ARE 15 ALPHA'S AND 15 GND'S

FIGURE 4

PREDICTIVE DESIGN PROCEDURES

VESY: A PROBABILISTIC PREDICTIVE MODEL  
FOR RAILROAD DESIGN

LATEST REVISION: 2 APRIL 1977  
(CSC) TCI

INPUT EXAMPLE 10-31-77

INPUT DATA VALUES FOR FIGURE 1

>>> TYPE  
FILE&IN

>>> LOADINC .1E-001

RADIUS .64 E+01

>>> LAYER1 2

>>> ZERACK .60 E+01

>>> ROLLIN .6E-001

>>> ZROLLIN 0

>>> ZROLLOUT .6E-001

>>> ROLLOUT

>>> T-TCS .3E-001 .6E-001 .4E-001

>>> ROLLIN .6E-001 .6E-001 .6E-001

>>> ROLLOUT 11

>>> STATINC .3E-001 .6E-001

.1E-001 .6E-001

>>> LAYER1 .6E-001 .6E-001 .6E-001

.1E-001 .6E-001

>>> VERTSTAT .2700E+00 .2700E+00

.2700E+00 .2700E+00

>>> LAYER2 .5E00E+00 .9E-001

.1E-001 .4E-001

•1199E-02 •1199E-01 •6199E+00



\*\*\*\*\* DESIGN EXAMPLE 12-71-7E \*\*\*\*\*

LITERATURE DATA -- CREEP COMPLIANCES

Layer 1	Time	Value	Variation
1.0000E+00	3.7630E+00	+2.7000E+00	
1.0000E+00	5.2500E+00	+2.7000E+00	
1.0000E+00	8.6740E+00	+2.7000E+00	
1.0000E+00	1.1910E+01	+2.7000E+00	
1.0000E+00	2.0500E+01	+2.7000E+00	
1.0000E+00	4.4300E+01	+2.7000E+00	
1.0000E+00	6.6200E+01	+2.7000E+00	
1.0000E+00	8.6600E+01	+2.7000E+00	
1.0000E+00	1.1200E+02	+2.7000E+00	
1.0000E+00	1.6000E+02	+2.7000E+00	

LITERATURE DATA	DELT	GES
1.00000E+03	-0.5550340E-06	
1.00000E+03	-0.170160E-07	
1.00000E+03	-0.379010E-08	
1.00000E+03	-0.664330E-09	
1.00000E+03	-0.107870E-09	
1.00000E+03	-0.193000E-09	

Layer 2	Time	Value	Variation
1.0000E+00	3.7010E+00	+2.7000E+00	
1.0000E+00	5.2100E+00	+2.7000E+00	
1.0000E+00	8.6300E+00	+2.7000E+00	
1.0000E+00	1.1900E+01	+2.7000E+00	
1.0000E+00	2.0500E+01	+2.7000E+00	
1.0000E+00	4.4200E+01	+2.7000E+00	
1.0000E+00	6.6100E+01	+2.7000E+00	
1.0000E+00	8.6500E+01	+2.7000E+00	
1.0000E+00	1.1200E+02	+2.7000E+00	
1.0000E+00	1.6000E+02	+2.7000E+00	

LITERATURE DATA	DELT	GES
1.00000E+03	-0.117210E-06	
1.00000E+03	-0.117214E-07	
1.00000E+03	-0.117215E-08	
1.00000E+03	-0.117216E-09	
1.00000E+03	-0.117217E-09	
1.00000E+03	-0.117218E-09	

LAYER 3	TIME	VALUE	VARIATION
1	100E-02	•170000E+00	•210000E+00
1	100E-01	•170000E+00	•210000E+00
1	100E+00	•170000E+00	•210000E+00
1	100E+01	•170000E+00	•210000E+00
1	100E+02	•170000E+00	•210000E+00
1	100E+03	•170000E+00	•210000E+00

FITTED DATA	DELTA	0.00
•170000E+00	0.	
•170000E+01	0.	
•170000E+02	0.	
•170000E+03	0.	
		•170000E+04

LAYER 4	TIME	VALUE	VARIATION
1	100E-02	•170000E+00	•200000E+00
1	100E-01	•170000E+00	•200000E+00
1	100E+00	•170000E+00	•200000E+00
1	100E+01	•170000E+00	•200000E+00
1	100E+02	•170000E+00	•200000E+00
1	100E+03	•170000E+00	•200000E+00

FITTED DATA	DELTA	0.00
•170000E+00	0.	
•170000E+01	0.	
•170000E+02	0.	
•170000E+03	0.	
		•170000E+04

LAYER 5	TIME	VALUE	VEGETATION
1	100E-02	•221000E+00	•321000E+00
1	100E-01	•221000E+00	•321000E+00
1	100E+00	•221000E+00	•321000E+00
1	100E+01	•221000E+00	•321000E+00
1	100E+02	•221000E+00	•321000E+00
1	100E+03	•221000E+00	•321000E+00

FITTED DATA	DELTA	0.00
•221000E+00	0.	
•221000E+01	0.	
•221000E+02	0.	
•221000E+03	0.	
		•221000E+04

## \*\*\*\*\* DESIGN EXAMPLE 12-31-78

## PROBLEM PARAMETERS

TOTAL LOAD (KIPS) 120.00  
 TIP PRESSURE (PSI) 1.0  
 LOAD RADIUS (IN) 6.43  
 LAYER 1 - MODULUS (EPRATIO) POISONS RATIO .500 THICKNESS (IN) 3.00  
 LAYER 2 - MODULUS (EPRATIO) POISONS RATIO .500 THICKNESS (IN) 3.00  
 LAYER 3 - MODULUS (EPRATIO) POISONS RATIO .500 THICKNESS (IN) 4.00  
 LAYER 4 - MODULUS (EPRATIO) POISONS RATIO .500 THICKNESS (IN) 4.00  
 LAYER 5 - MODULUS 45426. POISONS RATIO .500 SEMI-INFINITE THICKNESS

TIME = 0.1 IN. SEC. SECNDS

## ES PARALLELIC PARAMETERS

LAYER 1--COEFFICIENT OF VARIATION OF MODULUS 27.00  
 LAYER 2--COEFFICIENT OF VARIATION OF MODULUS 27.00  
 LAYER 3--COEFFICIENT OF VARIATION OF MODULUS 20.00  
 LAYER 4--COEFFICIENT OF VARIATION OF MODULUS 20.00  
 LAYER 5--COEFFICIENT OF VARIATION OF MODULUS 12.00

STRAINS									
		VERTICAL		TANGENTIAL		RADIAL		SHEAR	
R	Z	MEAN DEVIATION							
0.00	0.00	-0.100E+1	-0.30E-14	-0.3720E-01	-0.497E-01	-0.370E-01	-0.157E-01	0.0	0.0
0.00	0.00	-0.100E+1	-0.160E-01	-0.694E-01	-0.216E-01	-0.594E-01	-0.365E-03	0.0	0.0
0.00	0.00	-0.992E+0	-0.795E-15	-0.711E+01	-0.370E+01	-0.282E+01	-0.327E+00	-0.627E-19	-0.173E-14
0.00	0.00	-0.101E+0	-0.134E-01	-0.600E-01	-0.167E-01	-0.507E-01	-0.158E-01	-0.214E-01	-0.137E-02

## \*\*\*\*\* DESIGN EXAMPLE 12-31-78

## DISPLACEMENTS

DISPLACEMENTS									
		RADIAL		RADIAL		TANGENTIAL			
R	Z	MEAN DEVIATION							
0.00	0.00	-0.127E-04	-0.127E-04	0.0	0.0	-0.500E-06	-0.159E-06	-0.602E-06	-0.117E-06
0.00	0.00	-0.542E-04	-0.121E-04	-0.262E-05	-0.377E-05	-0.312E-06	-0.331E-07	-0.674E-06	-0.208E-06
0.00	0.00	-0.541E-04	-0.121E-04	-0.318E-05	-0.324E-05	-0.359E-06	-0.347E-07	-0.629E-06	-0.235E-07

6.1.12869 Date: 04-03

34-18245 - 43-18245 - 34-18245

THE CHICKENNESS  
SOCIETY OF THE UNITED STATES

1-49  
1-51  
1-50-1  
1-51  
1-52-1

Sudanese Arabic

Digitized by srujanika@gmail.com

Digitized by srujanika@gmail.com

2. **APRIL 11, 1941** **TELEGRAM** **TO THE SECRETARY OF STATE**  
**RE BRIEFING ON THE SITUATION IN AFRICA**

$$S_{\text{solid}} = 10 \cdot 100 \text{ J}^\circ = 1000$$

Sustained until

\*\*\*\*\* DESIGN EXAMPLE 12-31-71

DRONE FLIGHT PLANNING

TIME = 10000.00 SECONDS

THE PAPERS OF THE PRESIDENT

Digitized by srujanika@gmail.com

P	T	VERTICAL		TANGENTIAL		RADIAL		SHEAR	
		MEAN DEVIATION	STDEV						
8.00	8.00	-0.12E+01	0.34E-15	-0.25E+01	0.37E+00	-0.25E+01	0.35E+01	-0.15E+01	0.30E+00
	6.00	-0.24E+00	0.50E-01	-0.25E+01	0.36E+01	-0.25E+01	0.35E+03	-0.15E+01	0.30E+00
6.00	8.00	-0.89E-05	0.63E-15	-0.21E+01	0.24E+00	-0.19E+01	0.20E+00	-0.17E-14	0.17E-14
	6.00	-0.12E+00	0.17E+01	-0.27E+01	0.23E+01	-0.29E+01	0.22E+01	-0.20E-14	0.21E-22

\*\*\*\*\* FOREIGN EXAMPLE 12-21-70

• ISPLACES

		VERTICAL		RADIAL		RADIAL		TANGENTIAL	
e	r	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION
0.00	0.00	0.00E+00	0.16E-04	0.	0.	0.11E-07	0.12E-06	0.11E-05	0.35E-06
0.00	F.00	0.07E+00	0.171E-04	0.	0.	0.18E-06	0.22E-06	0.15E-05	0.49E-06
1.00	0.00	0.79E+00	0.16E-04	-0.55E+00	0.63E-06	-0.57E-06	0.77E-07	-0.91E-06	0.62E-07
1.00	F.00	0.77E+00	0.16E-04	-0.76E+00	0.59E-06	-0.61E-06	0.80E-07	-0.12E-05	0.74E-07

## \*\*\*\*\* DESIGN EXAMPLE 12-71-7F \*\*\*\*\*

## PROBLEM PARAMETERS

TOTAL LENGTH (L25)	125.00				
TIP DEFLECTION (DST1)	1.0				
LEAD RADIUS (R1)	0.4				
LAYER 1 - MODULUS	410000	POISSON'S RATIO	0.30	THICKNESS (IN)	3.00
LAYER 2 - MODULUS	410000	POISSON'S RATIO	0.30	THICKNESS (IN)	3.00
LAYER 3 - MODULUS	50000	POISSON'S RATIO	0.30	THICKNESS (IN)	4.00
LAYER 4 - MODULUS	50000	POISSON'S RATIO	0.30	THICKNESS (IN)	4.00
LAYER 5 - MODULUS	40415	POISSON'S RATIO	0.30	SHELL-INFINITE THICKNESS	

TIME = 0.0000 SEC SECONDS

## PROBABILISTIC PARAMETERS

LAYER 1--COEFFICIENT OF VARIATION OF MODULUS	27.00
LAYER 2--COEFFICIENT OF VARIATION OF MODULUS	27.00
LAYER 3--COEFFICIENT OF VARIATION OF MODULUS	20.00
LAYER 4--COEFFICIENT OF VARIATION OF MODULUS	20.00
LAYER 5--COEFFICIENT OF VARIATION OF MODULUS	12.00

## STRAINS

R	Z	VERTICAL		TANGENTIAL		RADIAL		SHEAR	
		MEAN DEVIATION	SD DEVIATION						
0.00	0.00	-0.13E+01	-0.42E-15	-0.212E+01	-0.316E-01	-0.215E+01	-0.185E-01	0.	0.
0.00	4.00	-0.36E+01	-0.274E-01	-0.109E+00	-0.410E-01	-0.168E+01	-0.107E-02	0.	0.
4.00	0.00	-0.26E+01	-0.331E-15	-0.177E+01	-0.192E-00	-0.165E+01	-0.150E-01	-0.691E-14	-0.141E-14
4.00	4.00	-0.342E+01	-0.160E-01	-0.774E-01	-0.212E-01	-0.117E+02	-0.238E-01	-0.658E-01	-0.72E-02

## \*\*\*\*\* DESIGN EXAMPLE 12-71-7F \*\*\*\*\*

## STRAINS

R	Z	DISPLACEMENTS		RADIAL		TANGENTIAL	
		MEAN DEVIATION	SD DEVIATION	MEAN DEVIATION	SD DEVIATION	MEAN DEVIATION	SD DEVIATION
0.00	0.00	-0.106E+01	-0.180E-04	0.	0.	-0.111E-05	-0.587E-06
0.00	4.00	-0.598E+01	-0.180E-04	0.	0.	-0.211E-06	-0.542E-06
4.00	0.00	-0.821E+01	-0.178E-04	-0.876E-05	-0.654E-06	-0.107E-06	-0.113E-05
4.00	4.00	-0.683E+01	-0.178E-04	-0.102E-04	-0.579E-06	-0.817E-07	-0.160E-05

• 44345-76 • 50765-76 • 50766-76 • 50767-76 • 50768-76 • 50769-76 • 50770-76 • 50771-76 • 50772-76 • 50773-76

SHIBAIS 1-11-61 3700X4000

...PRAECLARUM  
...SNEZHOV  
...TAVOLI  
...TAVOLI  
...TAVOLI  
...TAVOLI

.....  
S 3 S S - 1 1 5

Digitized by srujanika@gmail.com

199 (all) 50114-0207  
201 (1st) 50114-0211  
202 (all) 50114-0214

Solutions and 80

12-71-7  
MATERIAL TESTED

17315

三

• 51 •

.....

24-34990 0-0-08610 10-34910 9-0-08610 10-34910 10-34910 10-34910 0-0-08610 10-34910 0-0-08610 30-9 30-9

Digitized by srujanika@gmail.com

Sundays 16. May 19 = 241

69° 69° 69°  
69° 69° 69°  
69° 69° 69°

Digitized by srujanika@gmail.com

SL-14-61 37187A3 1.01.31 100000000000000

## \*\*\*\*\* DESIGN EXAMPLE 12-71-75 \*\*\*\*\*

## PROFILE PARAMETERS

TOTAL LOAD (KIPS) 124.61  
 TIRE PRESSURE (PSIA) 1.21  
 LOAD FACTOR (1/10) 6.4

LAYER 1 - "M" ELLUS 116E-03 POISSON'S RATIO .500 THICKNESS (IN) 3.00  
 LAYER 2 - "M" ELLUS 117E-03 POISSON'S RATIO .500 THICKNESS (IN) 3.00  
 LAYER 3 - "M" ELLUS 118E-03 POISSON'S RATIO .500 THICKNESS (IN) 4.00  
 LAYER 4 - "M" ELLUS 119E-03 POISSON'S RATIO .500 THICKNESS (IN) 4.00  
 LAYER 5 - "M" ELLUS 404E-05 POISSON'S RATIO .700 SEMI-INFINITE THICKNESS

TYPE = 27 DEG/61 SECONDS

## PROBABILISTIC PARAMETERS

LAYER 1--COEFFICIENT OF VARIATION OF MODULUS .270  
 LAYER 2--COEFFICIENT OF VARIATION OF MODULUS .270  
 LAYER 3--COEFFICIENT OF VARIATION OF MODULUS .200  
 LAYER 4--COEFFICIENT OF VARIATION OF MODULUS .200  
 LAYER 5--COEFFICIENT OF VARIATION OF MODULUS .320

## STRESSES

R	Z	VERTICAL		TANGENTIAL		RADIAL		SHEAR		SIGMA
		MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION	MEAN	DEVIATION	
0.00	0.00	2.142E+01	2.207E-15	2.133E+01	2.013E-01	1.93E+00	2.133E+01	1.17E-01	0.	0.
0.00	6.00	2.572E+01	2.182E-01	2.561E+01	2.013E-01	2.131E+00	2.131E+01	7.95E-03	0.	0.
6.00	0.00	2.855E+01	2.177E-15	2.121E+01	2.000E-01	1.80E+00	2.117E+01	7.64E-01	-2.73E-16	51.05
6.00	6.00	3.375E+01	2.764E-02	3.250E+01	2.750E-02	2.25E+00	2.125E+01	1.30E-01	-1.58E+00	51.05

## \*\*\*\*\* DESIGN EXAMPLE 12-71-75 \*\*\*\*\*

## DISPLACEMENTS

## STRAINS

R	Z	MEAN DEVIATION		MEAN DEVIATION		MEAN DEVIATION		MEAN DEVIATION		SIGMA
		VERTICAL	RADIAL	TANGENTIAL	RADIAL	TANGENTIAL	MEAN	DEVIATION	MEAN	
0.00	0.00	2.146E-01	2.027E-04	0.	0.	0.	2.146E-01	9.65E-06	-2.140E-05	2.119E-05
0.00	6.00	2.125E-01	2.027E-04	0.	0.	0.	2.125E-01	3.20E-06	-2.375E-05	2.215E-06
6.00	0.00	2.113E-01	2.000E-04	-2.755E-06	2.133E-06	-2.548E-07	2.100E-01	-1.99E-06	-2.225E-05	2.022E-06
6.00	6.00	2.103E-01	2.027E-04	-2.169E-06	2.024E-06	-2.029E-07	2.050E-01	-2.225E-07	-2.225E-05	2.179E-07

\*\*\*\*\* DESIGN EXAMPLE 12-31-77

ΕΡΓΑ ΛΕΠΤΑ ΕΙΓΡΑΣ

TOTAL LOAD (LBS) 108.4  
TIRE PRESSURE (PSI) 1.6  
LOAD PER TIRE (LBS) 6.4

LAYER 1 = POLYIMIDE 0.5142, POISSON'S RATIO .300 THICKNESS (.1IN) 3.00  
 LAYER 2 = POLYIMIDE 0.5142, POISSON'S RATIO .300 THICKNESS (.1IN) 3.00  
 LAYER 3 = POLYIMIDE 0.5142, POISSON'S RATIO .300 THICKNESS (.1IN) 3.00  
 LAYER 4 = POLYIMIDE 0.5142, POISSON'S RATIO .300 THICKNESS (.1IN) 3.00

TIME = -01 000 002 01000 005

#### OPERATIONAL PARAMETERS

LAYER 1--COFFEE CUP T OF VEGETATION OF MOLLUSCS 27.  
 LAYER 2--COFFEE CUP T OF VEGETATION OF MOLLUSCS 27.  
 LAYER 3--COFFEE CUP T OF VEGETATION OF MOLLUSCS 28.  
 LAYER 4--COFFEE CUP T OF VEGETATION OF MOLLUSCS 28.

STRATEGIES

VERTICAL PREDICTIVE DIA

$\mu$	$\sigma$	MEAN DEVIATION							
0.00	0.00	$-0.15E-01$	$-0.10E-15$	$-0.117E-01$	$-0.15E-06$	$-0.175E-01$	$-0.16E-01$	0.	0.
0.00	0.00	$-0.15E-01$	$-0.147E-01$	$-0.135E-01$	$-0.11E-01$	$-0.138E-01$	$-0.164E-01$	0.	0.
0.00	0.00	$-0.15E-01$	$-0.227E-01$	$-0.111E-01$	$-0.741E-01$	$-0.125E-01$	$-0.527E-01$	$-0.118E-14$	$-0.433E-15$
0.00	0.00	$-0.15E-01$	$-0.393E-02$	$-0.111E-01$	$-0.11E-01$	$-0.125E-01$	$-0.122E-01$	$-0.178E-01$	$-0.500E-12$

\*\*\*\*\* DESIGN EXAMPLE 12-31-70

198 LACERTES

.....  
VERTICAL

8.66 2015-1615-13 2015-16 8.

S T R A I N

CENTRAL TANGENTIAL

	MEAN	STDEV	MEAN	STDEV
REF-05	-1020E-05	-1000E-05	174E-05	174E-05
172-05	1249E-06	1417E-05	153E-06	
145-05	1885E-06	-785E-16	524E-06	

## \*\*\*\*\* DITION EXAMPLE 12-31-74

## PROLEM PARAMETERS

TOTAL LOAD (KPS)	120.00				
TIE COEFFICIENT (PSI)	1.00				
LOAD RATIO (PSI)	6.00				
LAYER 1 - MODULUS	62435.	POISSON'S RATIO	.500	THICKNESS (IN)	3.00
LAYER 2 - MODULUS	62435.	POISSON'S RATIO	.500	THICKNESS (IN)	3.00
LAYER 3 - MODULUS	51824.	POISSON'S RATIO	.500	THICKNESS (IN)	4.00
LAYER 4 - MODULUS	50924.	POISSON'S RATIO	.500	THICKNESS (IN)	4.00
LAYER 5 - MODULUS	49445.	POISSON'S RATIO	.500	DEFINITE THICKNESS	

TIME = 1.31200E-01 SECONDS

## PROBLEM STIFF PARAMETERS

LAYER 1-COEFFICIENT OF VARIATION OF MODULUS	27.0%
LAYER 2-COEFFICIENT OF VARIATION OF MODULUS	27.0%
LAYER 3-COEFFICIENT OF VARIATION OF MODULUS	26.0%
LAYER 4-COEFFICIENT OF VARIATION OF MODULUS	26.0%
LAYER 5-COEFFICIENT OF VARIATION OF MODULUS	22.0%

S T R E S S E S									
		V E R T I C A L		T A N G E N T I A L		R A D I A L		S H E A R	
E Z		M A P S D E V I A T I O N		M E M D E V I A T I O N		M F A C D E V I A T I O N		M P E D D E V I A T I O N	
		M A P S D E V I A T I O N		M E M D E V I A T I O N		M F A C D E V I A T I O N		M P E D D E V I A T I O N	
0.00	2.00	-1.145E+01	-1.34E-15	-1.134E+01	-1.78E+01	-1.125E+01	-97E-02	0.	0.
0.00	6.00	-6.633E+01	-1.03E-01	-6.134E+00	-1.77E-01	-6.124E+01	-265E-03	0.	0.
6.00	0.00	-6.398E+01	-1.34E-15	-6.134E+01	-6.122E-01	-6.133E+01	-400E-01	-345E-14	-1.00E-15
6.00	6.00	-3.345E+01	-1.61E-02	-6.075E+01	-9.12E-02	-6.158E+01	-842E-02	-1.94E+00	-2.94E-02

## \*\*\*\*\* DITION EXAMPLE 12-31-74

S T R E S S E S									
		V E R T I C A L		R A D I A L		R A D I A L		T A N G E N T I A L	
E Z		M A P S D E V I A T I O N		M E M D E V I A T I O N		M F A C D E V I A T I O N		M P E D D E V I A T I O N	
		M A P S D E V I A T I O N		M E M D E V I A T I O N		M F A C D E V I A T I O N		M P E D D E V I A T I O N	
0.00	2.00	-1.75E-01	-2.14E-06	0.	0.	-2.012E-06	-1.02E-06	-6.23E-06	-2.09E-06
0.00	6.00	-1.20E-01	-1.24E-04	-2.03E-06	-3.74E-05	-2.23E-06	-2.06E-06	-3.38E-06	-6.61E-06
6.00	0.00	-1.15E-01	-2.07E-04	-2.00E-04	-8.98E-07	-1.62E-05	-5.67E-07	-3.34E-05	-2.10E-07

## \*\*\*\*\* DESIGN EXAMPLE 12-71-7 \*\*\*\*\*

## PROFILE PARAMETERS

TOTAL LOAD (LBS)	100.00				
TIRE PRESSURE (PSI)	1.0				
LOAD RADIUS (IN)	6.0				
LAYER 1 - MODULUS	63E+7	POISSON'S RATIO	.500	THICKNESS (IN)	3.00
LAYER 2 - MODULUS	55E+7	POISSON'S RATIO	.500	THICKNESS (IN)	3.00
LAYER 3 - MODULUS	50E+7	POISSON'S RATIO	.500	THICKNESS (IN)	4.00
LAYER 4 - MODULUS	52E+7	POISSON'S RATIO	.500	THICKNESS (IN)	4.00
LAYER 5 - MODULUS	45E+7	POISSON'S RATIO	.500	SEMI-INFINITE THICKNESS	

TIME = .107 E+03 SECONDS

## PRE-ELASTIC PARAMETERS

LAYER 1-COEFFICIENT OF VARIATION OF MODULUS	27.0
LAYER 2-COEFFICIENT OF VARIATION OF MODULUS	27.0
LAYER 3-COEFFICIENT OF VARIATION OF MODULUS	20.0
LAYER 4-COEFFICIENT OF VARIATION OF MODULUS	20.0
LAYER 5-COEFFICIENT OF VARIATION OF MODULUS	20.0

## SITE EFFECTS

		VERTICAL	TRANSMISSION	RADIAL	SHEAR					
#	Z	MEAN DEVIATION	MEAN DEVIATION	MEAN DEVIATION	MEAN DEVIATION					
0.00	2.00	-2.102E+01	-45.7E-17	-2.995E+00	1.22E+00	-2.955E+02	0.	0.	0.	SLOP
0.00	6.00	-2.643E+01	-2.80E-02	-2.136E+00	3.17E-01	-2.134E+02	1.07E-03	0.	0.	SLOP
6.00	2.00	-2.029E+01	-2.54E-15	-2.120E+01	3.56E-01	-2.121E+01	3.41E-01	-4.24E-14	2.71E-16	SLOP
6.00	6.00	-2.376E+01	-2.11E-02	-2.654E+01	3.91E-02	-2.161E+00	2.17E-02	-2.22E+00	3.526E-12	SLOP

## \*\*\*\*\* DESIGN EXAMPLE 12-71-7 \*\*\*\*\*

## DISPLACEMENTS

## DEFORMATIONS

		TRANSMISSION	RADIAL	TEMPERATURE					
#	Z	MEAN DEVIATION	MEAN DEVIATION	MEAN DEVIATION					
0.00	2.00	2.10E+04	2.10E+04	0.	0.	0.53E+17	1.01E+05	4.31E+07	2.71E+05
0.00	6.00	2.17E+03	2.10E+04	0.	0.	4.77E+15	1.55E+04	4.67E+05	1.75E+04
6.00	2.00	2.14E+03	2.10E+04	-5.62E+07	4.43E+05	-1.11E+17	2.11E+06	2.02E+07	1.05E+06
6.00	6.00	2.14E+03	2.08E+04	-2.207E+06	4.91E+05	-1.20E+17	2.22E+07	2.34E+06	1.08E+07

## \*\*\*\*\* CLOUD COMPUTING SYSTEM ARCHITECTURE

DETERMINING FACTOR FOR CRICKING IS RADIAL STRAIN AT 4,00 IN.

FITTED DATA	DELTAS	RES
-1.00000E+ 3	-0.204692E-06	
-1.00000E+ 2	-0.501290E-07	
-1.00000E+ 1	-0.250648E-06	
-1.00000E+ 0	-0.975155E-07	
-1.00000E- 1	-0.250350E-07	
-1.00000E- 2	-0.744382E-06	
-1.00000E- 3	-0.120054E-06	

DETERMINING FACTOR FOR PUTTING IN VERTICAL DISPLACEMENT AT 0.00 IN.

PERMANENT DEFORMATION  
SYSTEM RESPONSE FACTORS  
GPII = -0.7325 ± 0.08  
ALPHA = -24295 ± 10

F A T I M E C R A F T I N G

## GENERAL STRAIN RESPONSE DUE TO ERIV LOADING (HAWKES)

DEFENSE INDEX	VAR. DEFENSE INDEX	AREA	CRACKED	TIME
DEFENSE INDEX	DEFENSE INDEX	604 YDS/180000 YR'S		YEARS
124702	-22	121145	5	1970
124144	-22	121211	5	1970
127516	-22	125347	5	1970
124210	-22	124220	5	1970
127245	-21	125530	5	1970
127251	-21	125545	5	1970
125739	-21	125501	5	1970
125216	-21	125511	5	1970
127217	-21	127211	5	1970
125233	-20	127221	4	1970

RUT DEFLECTION

PERMANENT DEFORMATION  
SYSTEM DEFLECTION FACTORS  
GAMMA = .65385 ± .01  
ALPHA = .24215 ± .01

GENERAL DEFLECTIONS PREDICTED  
DUE TO PEAK LOADING (KIPS/ESTIMATED)

TEMPERATURE	DEFLECTION	VARIANCE DEFLECTION
DEGREES F	INCHES	INCHES <sup>2</sup>
40.070	.4423E-.02	.2281E-.05
40.030	.4428E-.02	.2249E-.05
40.000	.4421E-.02	.2177E-.05
40.000	.4411E-.02	.2154E-.05
39.930	.4412E-.02	.2140E-.05
39.900	.4413E-.02	.2126E-.05
39.850	.4414E-.02	.2112E-.05
39.800	.4414E-.02	.2098E-.05
39.750	.4414E-.02	.2084E-.05
39.700	.4414E-.02	.2070E-.05
39.650	.4414E-.02	.2056E-.05
39.600	.4414E-.02	.2042E-.05
39.550	.4414E-.02	.2028E-.05
39.500	.4414E-.02	.2014E-.05
39.450	.4414E-.02	.2000E-.05
39.400	.4414E-.02	.1986E-.05
39.350	.4414E-.02	.1972E-.05
39.300	.4414E-.02	.1958E-.05
39.250	.4414E-.02	.1944E-.05
39.200	.4414E-.02	.1930E-.05
39.150	.4414E-.02	.1916E-.05
39.100	.4414E-.02	.1902E-.05
39.050	.4414E-.02	.1888E-.05
39.000	.4414E-.02	.1874E-.05
38.950	.4414E-.02	.1860E-.05
38.900	.4414E-.02	.1846E-.05
38.850	.4414E-.02	.1832E-.05
38.800	.4414E-.02	.1818E-.05
38.750	.4414E-.02	.1804E-.05
38.700	.4414E-.02	.1790E-.05
38.650	.4414E-.02	.1776E-.05
38.600	.4414E-.02	.1762E-.05
38.550	.4414E-.02	.1748E-.05
38.500	.4414E-.02	.1734E-.05
38.450	.4414E-.02	.1720E-.05
38.400	.4414E-.02	.1706E-.05
38.350	.4414E-.02	.1692E-.05
38.300	.4414E-.02	.1678E-.05
38.250	.4414E-.02	.1664E-.05
38.200	.4414E-.02	.1650E-.05
38.150	.4414E-.02	.1636E-.05
38.100	.4414E-.02	.1622E-.05
38.050	.4414E-.02	.1608E-.05
38.000	.4414E-.02	.1594E-.05
37.950	.4414E-.02	.1580E-.05
37.900	.4414E-.02	.1566E-.05
37.850	.4414E-.02	.1552E-.05
37.800	.4414E-.02	.1538E-.05
37.750	.4414E-.02	.1524E-.05
37.700	.4414E-.02	.1510E-.05
37.650	.4414E-.02	.1496E-.05
37.600	.4414E-.02	.1482E-.05
37.550	.4414E-.02	.1468E-.05
37.500	.4414E-.02	.1454E-.05
37.450	.4414E-.02	.1440E-.05
37.400	.4414E-.02	.1426E-.05
37.350	.4414E-.02	.1412E-.05
37.300	.4414E-.02	.1398E-.05
37.250	.4414E-.02	.1384E-.05
37.200	.4414E-.02	.1370E-.05
37.150	.4414E-.02	.1356E-.05
37.100	.4414E-.02	.1342E-.05
37.050	.4414E-.02	.1328E-.05
37.000	.4414E-.02	.1314E-.05
36.950	.4414E-.02	.1300E-.05
36.900	.4414E-.02	.1286E-.05
36.850	.4414E-.02	.1272E-.05
36.800	.4414E-.02	.1258E-.05
36.750	.4414E-.02	.1244E-.05
36.700	.4414E-.02	.1230E-.05
36.650	.4414E-.02	.1216E-.05
36.600	.4414E-.02	.1202E-.05
36.550	.4414E-.02	.1188E-.05
36.500	.4414E-.02	.1174E-.05
36.450	.4414E-.02	.1160E-.05
36.400	.4414E-.02	.1146E-.05
36.350	.4414E-.02	.1132E-.05
36.300	.4414E-.02	.1118E-.05
36.250	.4414E-.02	.1104E-.05
36.200	.4414E-.02	.1090E-.05
36.150	.4414E-.02	.1076E-.05
36.100	.4414E-.02	.1062E-.05
36.050	.4414E-.02	.1048E-.05
36.000	.4414E-.02	.1034E-.05
35.950	.4414E-.02	.1020E-.05
35.900	.4414E-.02	.1006E-.05
35.850	.4414E-.02	.992E-.05
35.800	.4414E-.02	.978E-.05
35.750	.4414E-.02	.964E-.05
35.700	.4414E-.02	.950E-.05
35.650	.4414E-.02	.936E-.05
35.600	.4414E-.02	.922E-.05
35.550	.4414E-.02	.908E-.05
35.500	.4414E-.02	.894E-.05
35.450	.4414E-.02	.880E-.05
35.400	.4414E-.02	.866E-.05
35.350	.4414E-.02	.852E-.05
35.300	.4414E-.02	.838E-.05
35.250	.4414E-.02	.824E-.05
35.200	.4414E-.02	.810E-.05
35.150	.4414E-.02	.796E-.05
35.100	.4414E-.02	.782E-.05
35.050	.4414E-.02	.768E-.05
35.000	.4414E-.02	.754E-.05
34.950	.4414E-.02	.740E-.05
34.900	.4414E-.02	.726E-.05
34.850	.4414E-.02	.712E-.05
34.800	.4414E-.02	.698E-.05
34.750	.4414E-.02	.684E-.05
34.700	.4414E-.02	.670E-.05
34.650	.4414E-.02	.656E-.05
34.600	.4414E-.02	.642E-.05
34.550	.4414E-.02	.628E-.05
34.500	.4414E-.02	.614E-.05
34.450	.4414E-.02	.600E-.05
34.400	.4414E-.02	.586E-.05
34.350	.4414E-.02	.572E-.05
34.300	.4414E-.02	.558E-.05
34.250	.4414E-.02	.544E-.05
34.200	.4414E-.02	.530E-.05
34.150	.4414E-.02	.516E-.05
34.100	.4414E-.02	.502E-.05
34.050	.4414E-.02	.488E-.05
34.000	.4414E-.02	.474E-.05
33.950	.4414E-.02	.460E-.05
33.900	.4414E-.02	.446E-.05
33.850	.4414E-.02	.432E-.05
33.800	.4414E-.02	.418E-.05
33.750	.4414E-.02	.404E-.05
33.700	.4414E-.02	.390E-.05
33.650	.4414E-.02	.376E-.05
33.600	.4414E-.02	.362E-.05
33.550	.4414E-.02	.348E-.05
33.500	.4414E-.02	.334E-.05
33.450	.4414E-.02	.320E-.05
33.400	.4414E-.02	.306E-.05
33.350	.4414E-.02	.292E-.05
33.300	.4414E-.02	.278E-.05
33.250	.4414E-.02	.264E-.05
33.200	.4414E-.02	.250E-.05
33.150	.4414E-.02	.236E-.05
33.100	.4414E-.02	.222E-.05
33.050	.4414E-.02	.208E-.05
33.000	.4414E-.02	.194E-.05
32.950	.4414E-.02	.180E-.05
32.900	.4414E-.02	.166E-.05
32.850	.4414E-.02	.152E-.05
32.800	.4414E-.02	.138E-.05
32.750	.4414E-.02	.124E-.05
32.700	.4414E-.02	.110E-.05
32.650	.4414E-.02	.96E-.05
32.600	.4414E-.02	.82E-.05
32.550	.4414E-.02	.68E-.05
32.500	.4414E-.02	.54E-.05
32.450	.4414E-.02	.40E-.05
32.400	.4414E-.02	.26E-.05
32.350	.4414E-.02	.12E-.05
32.300	.4414E-.02	.08E-.05
32.250	.4414E-.02	.04E-.05
32.200	.4414E-.02	.00E-.05

RUT DEFLECTION	VAR RUT DEFLECTION	TIME
INCHES	INCHES <sup>2</sup>	YEARS
.1111E-1	.1112E-02	.1000
.1111E-1	.1172E-02	.1020
.1111E-1	.1232E-02	.1040
.1111E-1	.1292E-02	.1060
.1111E-1	.1352E-02	.1080
.1111E-1	.1412E-02	.1100
.1111E-1	.1472E-02	.1120
.1111E-1	.1532E-02	.1140
.1111E-1	.1592E-02	.1160
.1111E-1	.1652E-02	.1180
.1111E-1	.1712E-02	.1200
.1111E-1	.1772E-02	.1220
.1111E-1	.1832E-02	.1240
.1111E-1	.1892E-02	.1260
.1111E-1	.1952E-02	.1280
.1111E-1	.2012E-02	.1300
.1111E-1	.2072E-02	.1320
.1111E-1	.2132E-02	.1340
.1111E-1	.2192E-02	.1360
.1111E-1	.2252E-02	.1380
.1111E-1	.2312E-02	.1400
.1111E-1	.2372E-02	.1420
.1111E-1	.2432E-02	.1440
.1111E-1	.2492E-02	.1460
.1111E-1	.2552E-02	.1480
.1111E-1	.2612E-02	.1500
.1111E-1	.2672E-02	.1520
.1111E-1	.2732E-02	.1540
.1111E-1	.2792E-02	.1560
.1111E-1	.2852E-02	.1580
.1111E-1	.2912E-02	.1600
.1111E-1	.2972E-02	.1620
.1111E-1	.3032E-02	.1640
.1111E-1	.3092E-02	.1660
.1111E-1	.3152E-02	.1680
.1111E-1	.3212E-02	.1700
.1111E-1	.3272E-02	.1720
.1111E-1	.3332E-02	.1740
.1111E-1	.3392E-02	.1760
.1111E-1	.3452E-02	.1780
.1111E-1	.3512E-02	.1800
.1111E-1	.3572E-02	.1820
.1111E-1	.3632E-02	.1840
.1111E-1	.3692E-02	.1860
.1111E-1	.3752E-02	.1880
.1111E-1	.3812E-02	.1900
.1111E-1	.3872E-02	.1920
.1111E-1	.3932E-02	.1940
.1111E-1	.3992E-02	.1960
.1111E-1	.4052E-02	.1980
.1111E-1	.4112E-02	.2000
.1111E-1	.4172E-02	.2020
.1111E-1	.4232E-02	.2040
.1111E-1	.4292E-02	.2060
.1111E-1	.4352E-02	.2080
.1111E-1	.4412E-02	.2100
.1111E-1	.4472E-02	.2120
.1111E-1	.4532E-02	.2140
.1111E-1	.4592E-02	.2160
.1111E-1	.4652E-02	.2180
.1111E-1	.4712E-02	.2200
.1111E-1	.4772E-02	.2220
.1111E-1	.4832E-02	.2240
.1111E-1	.4892E-02	.2260
.1111E-1	.4952E-02	.2280
.1111E-1	.5012E-02	.2300
.1111E-1	.5072E-02	.2320
.1111E-1	.5132E-02	.2340
.1111E-1	.5192E-02	.2360
.1111E-1	.5252E-02	.2380
.1111E-1	.5312E-02	.2400
.1111E-1	.5372E-02	.2420
.1111E-1	.5432E-02	.2440
.1111E-1	.5492E-02	.2460
.1111E-1	.5552E-02	.2480
.1111E-1	.5612E-02	.2500
.1111E-1	.5672E-02	.2520
.1111E-1	.5732E-02	.2540
.1111E-1	.5792E-02	.2560
.1111E-1	.5852E-02	.2580
.1111E-1	.5912E-02	.2600
.1111E-1	.5972E-02	.2620
.1111E-1	.6032E-02	.2640
.1111E-1	.6092E-02	.2660
.1111E-1	.6152E-02	.2680
.1111E-1	.6212E-02	.2700
.1111E-1	.6272E-02	.2720
.1111E-1	.6332E-02	.2740
.1111E-1	.6392E-02	.2760
.1111E-1	.6452E-02	.2780
.1111E-1	.6512E-02	.2800
.1111E-1	.6572E-02	.2820
.1111E-1	.6632E-02	.2840
.1111E-1	.6692E-02	.2860
.1111E-1	.6752E-02	.2880
.1111E-1	.6812E-02	.2900
.1111E-1	.6872E-02	.2920
.1111E-1	.6932E-02	.2940
.1111E-1	.6992E-02	.2960
.1111E-1	.7052E-02	.2980
.1111E-1	.7112E-02	.3000
.1111E-1	.7172E-02	.3020
.1111E-1	.7232E-02	.3040
.1111E-1	.7292E-02	.3060
.1111E-1	.7352E-02	.3080
.1111E-1	.7412E-02	.3100
.1111E-1		

## PERFORMANCE

PUT DEPTH	VAR PUT DEPTH	SLOPE VAR	VAR SLOPE VAR	CRACKING	VAR CRACKING	TIME
.57725E+01	.16138E+02	.21517E+03	.17751E+01	.26742E+02	.10136E+05	.50
.11125E+00	.17135E+02	.34570E+03	.52247E+01	.44117E+02	.15211E+05	1.50
.17125E+00	.19144E+02	.39551E+03	.68721E+01	.70886E+02	.25747E+05	1.50
.23125E+00	.24155E+02	.69425E+03	.10981E+01	.98248E+02	.39422E+05	2.00
.29125E+00	.32165E+02	.89226E+03	.15215E+01	.13243E+02	.45174E+05	3.00
.35125E+00	.38175E+02	.67018E+03	.25135E+01	.17658E+01	.60445E+05	4.00
.41125E+00	.44185E+02	.17156E+03	.44227E+01	.22719E+01	.15591E+04	8.00
.47125E+00	.50195E+02	.12180E+03	.72121E+01	.36621E+01	.22924E+04	12.00
.53125E+00	.56205E+02	.15527E+03	.12175E+01	.57711E+01	.47215E+04	16.00
.59125E+00	.62215E+02	.18550E+03	.18742E+01	.13243E+00	.72253E+04	20.00

SERVICEABILITY	VARIANCE OF SERVICEABILITY	TIME
4.54724E	.000054	.50
4.54713E	.01312	1.20
4.51550E	.07064	1.50
4.45674E	.01143	2.00
4.41713E	.01615	3.00
4.36267E	.03443	4.00
4.31216E	.01017	8.00
4.26931E	.01817	12.00
4.22330E	.01156	16.00
4.01969E	.02137	20.00

SERVICEABILITY FAILURE LEVEL IS .000000

## RELIABILITY STATE PROBABILITIES

STATE UPPER BOUNDS = .41E+51 .46791E0 .46467E0 .46156E7 .36215E0 .36204E0 .36347E3 .36162E2 .36781E1 .36500E0

STATE LOWER BOUNDS = .42742E0 .46067E0 .46128E7 .36945E0 .36004E0 .36343E3 .36002E2 .36791E1 .36500E0 -.41E+51

STATE	STATE1	STATE2	STATE3	STATE4	STATE5	STATE6	STATE7	STATE8	STATE9	STATE10	TIME	RELIABILITY
.50117E	.51427	.10896	.01623	.00149	.01477	.00291	.00080	.01949	.00200	.00000	.0	.100000
.50328E	.51428	.21497	.05923	.02729	.01482	.00461	.00063	.02298	.00200	.00000	.5	.999993
.50539E	.51112	.27124	.01264	.02134	.01274	.00461	.00066	.03326	.00200	.00000	1.0	.999993
.50750E	.51124	.28221	.02651	.02142	.01323	.00518	.00061	.03301	.00200	.00000	1.5	.999993
.50961E	.51125	.28626	.04653	.02142	.01323	.00518	.00062	.03302	.00200	.00000	2.0	.999993
.51172E	.51647	.29167	.01263	.02149	.01476	.00299	.00070	.02078	.00200	.00000	2.5	.999993
.51383E	.51139	.29744	.02644	.02174	.01213	.00126	.00118	.02371	.00200	.00000	4.0	.999993
.51594E	.51140	.29837	.02521	.02170	.01262	.00179	.00165	.02017	.00200	.00000	8.0	.999993
.51805E	.51251	.22513	.02647	.02148	.00946	.00305	.00169	.02292	.00200	.00000	12.0	.999993
.52016E	.51141	.19106	.024647	.02142	.01261	.00198	.00162	.02127	.00200	.00000	16.0	.999993
.52227E	.51141	.16327	.02327	.022761	.01861	.00743	.00247	.00576	.00185	.00000	20.0	.999993

WARNING: RELIABILITY EXCEEDS THE SPECIFIED TOLERANCE THROUGHOUT THE PERIOD OF THIS ANALYSIS. IF AN ACCURATE ESTIMATE OF SERVICE LIFE IS REQUIRED, THEN THIS PLOT SHOULD BE REINTERPTED WITH ADDITIONAL TERMINATION POINTS, EXTENDING THE ANALYSIS BEYOND THE TIME AT WHICH RELIABILITY DIPS BELOW THE MINIMUM ACCEPTABLE LEVEL.

## VESYS IIM SAMPLE PROBLEM

— 5 —

~~D-E-L-C-I-S-T-E-R~~

**VESYS II-M : A FLOW PLASTIC PREDICTIVE MODEL  
FOR GROUTING DESIGN**

LATEST REVISIORS: 20 JANUARY 1976  
FEDERAL HIGHWAY ADMINISTRATION

LESION EXAMPLE 3-1-16

INPUT DATA VALUE: Freq = 1

TYPE 1  
TITLES ENTERTAINMENT  
URLS

3333 10147-1

>>> TSTAT1  
0.1000E+01 - 0.3000E+01 - 0.1000E+01 0.1000E+01 - 0.3000E+01 - 0.1000E+01 0.1000E+01 - 0.3000E+01 - 0.1000E+01 0.1000E+01 - 0.3000E+01 - 0.1000E+01

>>> LAYER1  
0.1/100-06 0.5/100F-06 0.1/100UL-06 0.1/145.0F-06 0.1/2500F-06 0.1/4000F-06 0.1/6000F-06 0.1/8500F-06 0.1/11000F-06

```
>>> LAYERs  
_0.1700E-04 -0.1700E-04 -0.1700E-04 -0.1700E-04 -0.1700E-04 -0.1700E-04 -0.1700E-04 -0.1700E-04
```

>>> LAYER1  
-0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04 -0.2200E-04

~~~~~ VACUUM 0.010E-04  
~~~~~ VACUUM 0.020E-10  
~~~~~ VACUUM 0.030E-10

...>>> 11 111111

-2222 ELLIAS - 7 -



CURRENT ESTIMATE

FITTED CLEP CONVERGENCE FUNCTIONS - SPATIAL SERIES

1 = 0.011 LAYER-1 6.61 E-LAYER-2 6.41 E-LAYER-3

DELTA(1)

|   |            |             |            |            |
|---|------------|-------------|------------|------------|
| 1 | -0.124E-04 | 0.1         | 1.0        | 0.3000E-03 |
| 2 | -0.124E-05 | 0.1         | 1.0        | 0.3100E-02 |
| 3 | -0.124E-05 | 0.1         | 1.0        | 0.3000E-01 |
| 4 | -0.124E-05 | 0.1         | 1.0        | 0.1000E-01 |
| 5 | -0.124E-05 | 0.1         | 1.0        | 0.1000E-00 |
| 6 | -0.124E-05 | 0.1         | 1.0        | 0.2000E-01 |
| 7 | -0.124E-04 | 0.17000E-04 | 1.2200E-04 | 1.0        |

COEFFICIENTS

LE VARIATION: 0.200E 00 0.200E 00 0.325E 00

VALUES OF CLEP COEFFICIENTS WITH RESIDUALS

| LIST<br>(SECONDS) | LAYER 1     |             |             | LAYER 2     |             |             | LAYER 3     |             |     |
|-------------------|-------------|-------------|-------------|-------------|-------------|-------------|-------------|-------------|-----|
|                   | PREDICTED   | % ERK       | MEASURED    | PREDICTED   | % ERK       | MEASURED    | PREDICTED   | % ERK       |     |
| 0.10E-02          | 0.37000E-04 | 0.37122E-06 | -0.49       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.1 |
| 0.30E-02          | 0.37000E-04 | 0.16441E-06 | 1.44        | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.2 |
| 0.10E-01          | 0.37000E-04 | 1.73065E-06 | -1.53       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0 |
| 0.30E-01          | 0.37000E-04 | 0.14464E-05 | 0.66        | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.1 |
| 0.10E-00          | 0.37000E-04 | 0.24911E-05 | 0.16        | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0 |
| 0.30E-00          | 0.37000E-04 | 0.60374E-05 | -0.96       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0 |
| 0.10E-01          | 0.37000E-04 | 0.61376E-05 | 1.61        | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0 |
| 0.30E-01          | 0.37000E-04 | 0.61391E-05 | -0.23       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.1 |
| 0.10E-12          | 0.12125E-04 | -0.24       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0         |     |
| 0.30E-12          | 0.11136E-04 | 0.16019E-04 | -0.14       | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.0 |
| 0.10E-03          | 0.12125E-04 | 0.18969E-04 | 1.74        | 0.17000E-04 | 0.17000E-04 | 0.0         | 0.22000E-04 | 0.22000E-04 | 0.1 |

PRIMARY RAYLEIGH SYSTEM

LAYER 1 THICKNESS = 0.1 INCHES  
LAYER 2 THICKNESS = 0.1 INCHES  
RAYLEIGH VELOCITY = 0.1 INCHES  
VERTICAL DISTANCE = 0.04 INCHES  
LAYER OF INTEREST = 1  
LOADING IN PSF = 1

PRIMARY RAYLEIGH SYSTEM FREQUENCY

| TIME (SEC) | WAVE NUMBER | COST OF WAVE |
|------------|-------------|--------------|
| -0.101-02  | 0.000000000 | 0.221900000  |
| -0.099-02  | 0.000000000 | 0.221900000  |
| -0.101-01  | 0.000000000 | 0.177050000  |
| -0.099-01  | 0.000000000 | 0.177050000  |
| -0.101-00  | 0.000000000 | 0.177050000  |
| -0.099-00  | 0.000000000 | 0.177050000  |
| -0.101-01  | 0.148137000 | 0.150075000  |
| -0.099-01  | 0.148137000 | 0.150075000  |
| -0.101-02  | 0.148137000 | 0.150075000  |
| -0.099-02  | 0.148137000 | 0.150075000  |
| -0.101-00  | 0.148137000 | 0.150075000  |
| -0.099-00  | 0.148137000 | 0.150075000  |

CLAY-LIME SYSTEM WAVE NUMBER WAVE NUMBER

| I | CLAY SYSTEM  | LIME SYSTEM |
|---|--------------|-------------|
| 1 | -0.255851-06 | 0.100000000 |
| 2 | -0.10071-05  | 0.100000000 |
| 3 | -0.250000-06 | 0.100000000 |
| 4 | -0.222201-04 | 0.100000000 |
| 5 | -0.20011-04  | 0.100000000 |
| 6 | -0.175712-04 | 0.100000000 |
| 7 | -0.170702-03 | 0.100000000 |

PERMANENT DEFLECTION

SYSTEM RESPONSE FACTOR = 1.000000000  
CRO = 0.500000000  
ALENA = 0.000000000

PERFORATION AND CYLINDER

LAYER 1 THICKNESS = 0.1 INCHES  
LAYER 2 THICKNESS = 0.1 INCHES  
RADIAL RESISTANCE = 0.01 INCHES  
VERTICAL RESISTANCE = 0.01 INCHES  
LAYER OF INTERFAC = 1  
LOADING IN TENSILE = 1

CLOSED-UP CYLINDER STRESS

ccc

| TIME (SEC) | STRESS (PSI) | CLOSED UP VAL |
|------------|--------------|---------------|
| 0.10E-02   | 0.11700E+00  | 0.229062E-00  |
| 0.30E-02   | 0.12400E+00  | 0.124447E-00  |
| 0.10E-01   | 0.14700E+00  | 0.095640E-01  |
| 0.30E-01   | 0.15400E+00  | 0.011200E-01  |
| 0.10E-02   | 0.13700E+00  | 0.055611E-01  |
| 0.30E-02   | 0.12700E+00  | 0.097505E-01  |
| 0.10E-01   | 0.10000E+00  | 0.113571E-00  |
| 0.30E-01   | 0.10000E+00  | 0.120457E-00  |
| 0.10E-02   | 0.13700E+00  | 0.14631E-00   |
| 0.30E-02   | 0.14800E+00  | 0.158604E-01  |
| 0.10E-03   | 0.16000E+00  | 0.157391E-01  |

CLOSED-UP CYLINDER STRESS - CYLINDER STRESS

1 - CYL SYSTEM - CYL STRESS

|   |              |              |
|---|--------------|--------------|
| 1 | -0.63640E-02 | 0.100000E-03 |
| 2 | -0.13970E-02 | 0.180000E-02 |
| 3 | -0.11620E-02 | 0.200000E-02 |
| 4 | -0.96400E-02 | 0.100000E-11 |
| 5 | -0.71611E-02 | 0.100000E-04 |
| 6 | -0.39390E-02 | 0.200000E-11 |
| 7 | -0.44262E-02 | 0.01         |

FEAT 1-600-124 Cyclic

CENTRAL STRAIN CYCLING  
DUE TO FLOOR SWAYING DIMENSIONLESS

| TEMPERATURE | STRAIN      | VAR STRAIN                | NFAIL        | VAR NFAIL              | K1            | K2            | GMAX          |
|-------------|-------------|---------------------------|--------------|------------------------|---------------|---------------|---------------|
| DEGREES F   | 1.0E-1%     | (IN./IN.) <sup>0.02</sup> | CYCLES       | CYCLES <sup>0.02</sup> | CYCLES        | DIMENSIONLESS | DIMENSIONLESS |
| 49.00       | 0.00000E+00 | 3.21587E-09               | 0.32453E-09  | 0.26025E-15            | -0.245200E-04 | 0.257700E-01  | 0.33215E-02   |
| 53.50       | 0.00000E+00 | 0.27309E-09               | 0.43567E-08  | 0.45804E-15            | 0.72800E-04   | 0.25800E-01   | 0.12902E-01   |
| 58.00       | 0.00000E+00 | -0.31674E-04              | -0.56672E-08 | -0.74725E-15           | -0.15500E-03  | -0.26700E-01  | 0.55033E-01   |
| 61.00       | 0.00000E+00 | 1.11759E-08               | 0.39283E-08  | 0.37183E-15            | 0.79900E-03   | 0.26400E-01   | 0.59570E-00   |
| 75.00       | 0.00000E+00 | 1.26145E-06               | 0.33631E-08  | 0.27182E-15            | -0.24229E-02  | 0.257700E-01  | 0.13337E-01   |
| 81.00       | 0.00000E+00 | -0.53442E-06              | 0.49302E-08  | 0.44565E-15            | 0.12400E-01   | 0.25100E-01   | 0.20655E-02   |
| 84.00       | 0.00000E+00 | -0.74155E-08              | -0.92178E-08 | -0.20422E-16           | -0.28400E-01  | -0.25000E-01  | 0.46547E-02   |
| 84.70       | 0.00000E+00 | 0.19713E-08               | 0.94196E-08  | 0.21320E-16            | 0.25300E-01   | 0.26300E-01   | 0.45525E-02   |
| 78.00       | 0.00000E+00 | -0.45536E-08              | -0.40680E-08 | -0.40205E-15           | -0.55870E-02  | -0.258700E-01 | 0.10133E-02   |
| 71.00       | 0.00000E+00 | 0.16702E-06               | 0.34272E-06  | 0.20291E-15            | 0.93200E-03   | 0.25400E-01   | 0.10265E-01   |
| 59.00       | 0.00000E+00 | 0.20411E-06               | 0.56732E-08  | -0.17594E-15           | -0.45100E-03  | -0.26700E-01  | 0.28557E-01   |
| 52.00       | 0.00000E+00 | 0.22111E-06               | 0.44582E-08  | 0.47920E-15            | 0.65700E-04   | 0.26300E-01   | 0.39777E-02   |

| DAMAGE INDEX  | VAR 1.0E-1% INDEX | AREA CRACKED     | TIME  |
|---------------|-------------------|------------------|-------|
| DIMENSIONLESS | DIMENSIONLESS     | S.YRS/1000CYCLES | YEARS |
| -0.14868E-01  | 0.17577E-06       | 0.0              | 0.50  |
| -0.26433E-01  | 0.21759E-06       | 0.0              | 1.00  |
| -0.41121E-01  | 0.26130E-06       | 0.0              | 1.50  |
| -0.52805E-01  | 0.31654E-06       | 0.0              | 2.00  |
| -0.75926E-01  | 0.412075E-05      | 0.0              | 3.00  |
| -0.10573E-00  | 0.47000E-03       | 0.0              | 4.00  |
| -0.23789E-00  | 0.56335E-03       | 0.0              | 8.00  |
| -0.39649E-00  | 0.67216E-03       | 0.0              | 12.00 |
| -0.58151E-00  | 0.15623E-02       | 0.0              | 16.00 |
| -0.79297E-00  | 0.30433E-02       | 0.0              | 20.00 |

BUILT-IN EXPANSION

DEVIATION COEFFICIENTS

SYSTEM RESPONSE COEFFICIENTS

GLO = 0.5451E-04

ALPHA = 0.1455E-04

GENERAL VIBRATION TEST

LINEAR STAR TRADING LINEARISATION

TEMPERATURE DEVIATION = VARIATION OF  
LEAKAGE F 1000 °C IN %/K

|         |            |            |
|---------|------------|------------|
| -49.70  | 0.2110E+01 | 0.2262E+05 |
| -53.30  | 0.1920E+01 | 0.2302E+05 |
| -59.50  | 0.1510E+01 | 0.2593E+05 |
| -66.60  | 0.1050E+01 | 0.4570E+05 |
| -75.20  | 0.7100E+00 | 0.6202E+05 |
| -84.40  | 0.4100E-01 | 0.4144E+05 |
| -94.70  | 0.1800E-01 | 0.1170E+05 |
| -105.90 | 0.7000E-02 | 0.3715E+05 |
| -118.10 | 0.2000E-02 | 0.5951E+05 |
| -59.10  | 0.1100E-02 | 0.2550E+05 |
| -52.20  | 0.5000E-02 | 0.2743E+05 |

DEVIATION = VARIATION OF LEAKAGE  
IN CM<sup>3</sup>/S IN %/K

|             |            |       |
|-------------|------------|-------|
| 0.82944E-01 | 0.1000E+02 | 0.00  |
| 0.11425E-01 | 0.1000E+02 | 1.00  |
| 0.11690E-01 | 0.1000E+02 | 1.00  |
| 0.12941E-01 | 0.1000E+02 | 2.00  |
| 0.14216E-01 | 0.1000E+02 | 3.00  |
| 0.15324E-01 | 0.1000E+02 | 4.00  |
| 0.16675E-01 | 0.1000E+02 | 6.00  |
| 0.21154E-01 | 0.1000E+02 | 14.00 |
| 0.23226E-01 | 0.1000E+02 | 16.00 |
| 0.25052E-01 | 0.1000E+02 | 20.00 |

8-24-CAR-1-A-6

|             |            |            |
|-------------|------------|------------|
| STUFF       | AMOUNT     | ITEM #     |
| FACTURES    | 100        | 88133861-2 |
| .23063      | .01000 -11 | 0.00       |
| .40521 AM   | .01000 -01 | 1.00       |
| .02591 AM   | .01000 -01 | 1.00       |
| .56161      | .01000 -01 | 0.00       |
| .03448      | .01000 -00 | 0.00       |
| .767621 AM  | .01000 -00 | 4.00       |
| .0116921 01 | .01000 -00 | 0.00       |
| .015361 01  | .01000 -01 | 12.00      |
| .10520 01   | .01000 -01 | 10.00      |
| 21.520      | .01000 -01 | 0.00       |

~~— f f & f : b B = 1 c i —~~

| REL. EPTH  | VAR. E. T. EPTH | SLOPE VAR    | VAR SLOPE VAR | CRAC-ING     | VAR CRACKING | TIME  |
|------------|-----------------|--------------|---------------|--------------|--------------|-------|
| 0.8204E-01 | -0.1118E-02     | 0.2306E-06   | 0.24109E-01   | 0.14655E-01  | 0.25567E-04  | 0.51  |
| 0.1000E-00 | -0.1176E-02     | -0.4069E-06  | -0.72704E-01  | -0.2643E-01  | 0.43249E-04  | 1.02  |
| 0.1100E-01 | -0.1155E-02     | 0.4500E-06   | 0.45506E-01   | 0.41100E-01  | 0.68816E-04  | 1.51  |
| 0.1274E-01 | 0.10364E-02     | -0.5616E-06  | -0.14267E-00  | -0.52855E-01 | -0.95494E-04 | 2.00  |
| 0.1427E-01 | 0.10397E-02     | 0.6649E-06   | 0.21234E-00   | 0.72925E-01  | 0.12975E-03  | 3.03  |
| 0.1550E-01 | 0.10434E-02     | -0.7474E-06  | -0.28113E-00  | -0.41657E-01 | -0.17500E-03 | 3.62  |
| 0.1677E-01 | 0.10471E-02     | 0.82012E-06  | 0.23711E-01   | 0.44330E-02  | 0.209        | 3.09  |
| 0.1811E-01 | 0.10508E-02     | -0.15096E-01 | -0.10209E-01  | -0.55581E-01 | -0.33264E-03 | 4.00  |
| 0.2055E-01 | 0.10545E-02     | 0.16091E-01  | 0.14035E-01   | 0.56161E-02  | 0.15623E-02  | 15.01 |
| 0.2350E-01 | 0.10582E-02     | -0.21096E-01 | -0.20001E-01  | -0.19223E-02 | -0.29532E-02 | 20.00 |

| SERVICEABILITY | VARIANCE IN SERVICEABILITY | TIME  |
|----------------|----------------------------|-------|
| 0.21386        | 0.00111                    | 0.50  |
| 0.21211        | 0.10571                    | 1.00  |
| 0.21352        | 0.11518                    | 1.50  |
| 0.21272        | 0.12476                    | 2.00  |
| 0.21119        | 0.15107                    | 3.00  |
| 0.21173        | 0.14525                    | 4.00  |
| 0.21176        | 0.17612                    | 5.00  |
| 0.21050        | 0.19570                    | 15.00 |
| 0.21056        | 0.21671                    | 16.00 |
| 0.21057        | 0.23197                    | 20.00 |

SERVICEABILITY TESTS AND DESIGN

#### **SASOLIC SULFURIC ACID**

STATE OF CALIFORNIA - 2013-51 - 0-174859 - 6-19-11 - 6-13367-3-90155 - 3-67461 - 3-36333 - 3-06222 - 6-78111 - 2-50000

STATEMENT OF ESTATE - 4.03220 4.05775 4.12667 3.05555 3.06244 3.06354 3.06222 2.76111 2.05111 2.01111 1.51111

| STATE1  | STATE2  | STATE3  | STATE4  | STATE5  | STATE6  | STATE7  | STATE8  | STATE9  | STATE10 | STATE11 | STATE12 |
|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|
| 0.31959 | 0.32447 | 0.12212 | 0.32123 | 0.32323 | 0.30003 | 0.30003 | 0.0     | 0.00700 | 0.0     | 1.01100 |         |
| 0.34221 | 0.32447 | 0.0     | 0.0     | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.01100 |         |
| 0.24210 | 0.21100 | 0.20217 | 0.11100 | 0.02200 | 0.00223 | 0.00001 | 0.00001 | 0.00000 | 0.00000 | 1.01100 |         |
| 0.21716 | 0.11100 | 0.10100 | 0.06079 | 0.00000 | 0.00079 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.01100 |         |
| 0.19122 | 0.11100 | 0.11100 | 0.11100 | 0.01100 | 0.01100 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.00000 |         |
| 0.14971 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.00000 |         |
| 0.12679 | 0.21100 | 0.21100 | 0.21100 | 0.10564 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.01100 |         |
| 0.79113 | 0.11100 | 0.00000 | 0.00000 | 0.126   | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.00000 |         |
| 0.54947 | 0.11100 | 0.00000 | 0.00000 | 0.06756 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 1.00000 |         |
| 0.04043 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.15031 | 0.07134 | 0.12375 | 0.10055 | 0.10029 | 1.01100 |         |
| 0.17661 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.15032 | 0.07134 | 0.12375 | 0.10055 | 0.10029 | 1.01100 |         |

~~WARNING~~ PERIODICALLY EXCEETS THE SERVICE LIFE OF THE EQUIPMENT. THIS EQUIPMENT HAS A DESIGNATED SERVICE LIFE AS INDICATED, BUT THIS FOR THE USE AS INDICATED. THE ADDITIONAL THERMOMETER POINTS, EXISTING IN THE EQUIPMENT, ARE NOT FOR INDICATING TEMPERATURES BELOW THE SERVICE LIFE.

APPENDIX 2  
OUTPUT FOR SENSITIVITY ANALYSIS OF PRIME

Sample Problem Set 1

Sample Problem Set 2

Sample Problem Set 3

Sample Problem Set 4

NOTE: This part is included in a separate volume.

APPENDIX 3  
VISCOELASTIC CLOSED-FORM PROBABILISTIC  
SOLUTION FOR PRIME

## APPENDIX 3

## Viscoelastic Closed Form Probabilistic Solution for PRIME

1. Introduction

The primary response model PRIME for the VESYS G is a viscoelastic closed form probabilistic solution for N-layered pavement systems.

The geometrical model is a multi-layered, semi-infinite half-space, see Fig. A2-1. Each layer has distinct material properties characterized as linear elastic or linear viscoelastic. The material properties can be random (mean and deviation) or deterministic (mean value only). The loading is considered to be uniform, normal to the surface and acting over a circular area. The responses of the pavement system are the stresses (normal stress, tangential stress, radial stress and shear stress), strains (normal, tangential and gradial) and vertical deflection at any specified radial position and vertical position. As pointed out in Chapter 3, the viscoelastic solution for the primary responses (stresses, strains and deflections) was obtained using the "quasi-elastic" solution. This method involves replacing elastic moduli in a N-layered elastic solution by instantaneous values of the relaxation moduli (or creep compliances). The method is based upon the work of Schapery (1) and is applicable to viscoelastic problems in which the derivative of the time-dependent solution with respect to logarithmic times is a slowly varying function of  $\log t$ .

---

(1) Schapery, R. A., "A Method of Viscoelastic Stress Analysis Using Elastic Solutions", J. Franklin Institute, Vol. 279, pp. 268-289, 1965.

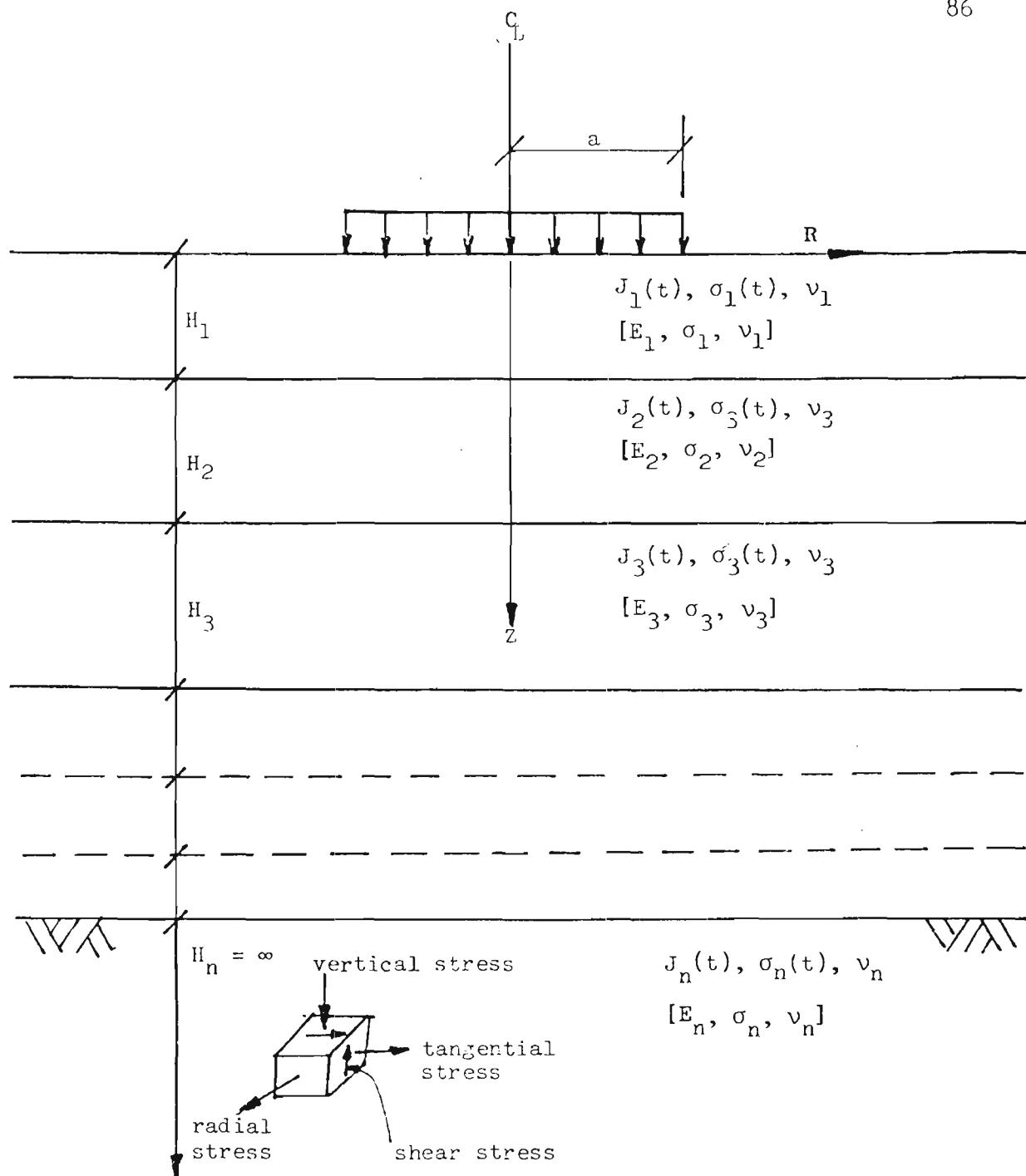


Figure 3-1. Schematic of N-Layered Viscoelastic System

Thus, the viscoelastic solution consists of a series of elastic solutions, with each elastic solution corresponding to the value of the viscoelastic solution at that specific time.

The quasi elastic solution consists of two parts, determining the expected values for the primary responses, and the deviations for the primary responses. CHEV5L (2) program was used in the first part to determine the expected values for the primary responses. Modifications were made on CHEV5L such that the program can handle any number of layers instead of a maximum of five layers.

In the following the formulations for determining the expected values and deviations of the primary responses for a N-layered pavement system are presented.

## 2. Primary Responses in an Elastic Layered System

The formulation presented in this section are based on the work of Warren and Dieckman (2). The reader is referred to the original work for more detailed derivations.

In an axially symmetric, cylindrical coordinate system, the stresses and displacements in each layer may be written in matrix form

$$S_i(z) = \begin{bmatrix} \sigma_z^i \\ \tau_{rz}^i \\ u_z^i \\ w^i \end{bmatrix} = K(v_i, E_i) M(z, v_i) D(z) \begin{bmatrix} A_i \\ B_i \\ C_i \\ D_i \end{bmatrix} \quad (1)$$

(2) Warren, H. and Dieckman, W. L., "Numerical Computation of Stresses and Strains in a Multiple Layer Asphalt Pavement System", Chevron Research Corp., Unpublished Internal Report, 1963.

where  $\sigma_z^i$ ,  $\tau_{rz}^i$ ,  $u^i$ , and  $w^i$  refer to normal stress, shear stress, radial displacement and vertical displacement in the  $i^{\text{th}}$  layer, and

$$K(v, E) = \begin{bmatrix} -m J_0(mr) & -- & -- & -- \\ -- & m^2 J_1(mr) & -- & -- \\ -- & -- & \frac{1+v}{E} m J_1(mr) & -- \\ -- & -- & -- & -\frac{1+v}{E} m J_0(mr) \end{bmatrix} \quad (2)$$

$$M(z, v) = \begin{bmatrix} 1 & mz + 2v - 1 & -1 & -mz + 2v - 1 \\ 1 & mz + 2v & 1 & mz - 2v \\ 1 & mz + 1 & -1 & -mz + 1 \\ 1 & mz + 4v - 2 & 1 & mz - 4v + 2 \end{bmatrix} \quad (3)$$

$$D(z) = \begin{bmatrix} me^{mz} & -- & -- & -- \\ -- & e^{mz} & -- & -- \\ -- & -- & me^{-mz} & -- \\ -- & -- & -- & e^{-mz} \end{bmatrix} \quad (4)$$

and  $A_i$ ,  $B_i$ ,  $C_i$ ,  $D_i$  are constants which are determined from boundary and continuity conditions between layers:

and  $A_i$ ,  $B_i$ ,  $C_i$ ,  $D_i$  are constants which are determined from boundary and continuity conditions between layers:

$$\begin{bmatrix} A_i \\ B_i \\ C_i \\ D_i \end{bmatrix} = \frac{D^{-1}(h_i) X_i D(h_i)}{4(v_i - 1)} \begin{bmatrix} A_{i+1} \\ B_{i+1} \\ C_{i+1} \\ D_{i+1} \end{bmatrix}, \quad i = 1, 2, \dots, n-1$$

$$= \sum_{j=i}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ C_n \\ D_n \end{bmatrix} \quad (5)$$

$$\begin{bmatrix} 0 \\ K(m) \end{bmatrix} = Q \sum_{j=1}^{n-1} \frac{\pi^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ C_n \\ D_n \end{bmatrix} \quad (6)$$

where  $K(M) = 1/m^2$ , and

$$Q = \begin{bmatrix} m & 2v_1 & m & -2v_1 \\ m & 2v_1 - 1 & -m & 2v_i - 1 \end{bmatrix} \quad (7)$$

$$X_i = 4M^{-1}(h_i, v_i) K^{-1}(v_i, E_i) K(v_{i+1}, E_{i+1}) M(h_i, v_{i+1}) (v_i - 1) \quad (8)$$

In the above equations  $m$  is a dummy variable;  $J_0(mr)$ ,  $J_1(mr)$  are Bessel functions,  $v_i$  and  $E_i$  are Poisson's ratio, and the Elastic Modulus of the  $i^{\text{th}}$  layer; and  $\sigma_z^i$ ,  $\tau_{rz}^i$ ,  $u^i$ ,  $w^i$  are the normal stress, shear stress, radial displacement and vertical displacement of the  $i^{\text{th}}$  layer. Thus, given the number of layers  $n$ ,  $E_i$ ,  $v_i$  ( $i = 1, \dots, n$ ) and with a unit load acting on the surface point, Equations (5) and (6) can be used to determine the coefficients  $A_i$ ,  $B_i$ ,  $C_i$ ,  $D_i$  ( $i = 1, \dots, n$ ). Then substituting into Equation (1), the stresses and displacements in each layer can be determined.

For a circular load, acting over radius as with intensity  $p$ , the stresses and displacements are given by

$$\begin{aligned} \tilde{\sigma}_z^i &= a \int_0^\infty J_1(ma) \sigma_z^i dm \\ \tilde{\tau}_{rz}^i &= a \int_0^\infty J_1(ma) \tau_{rz}^i dm \\ \tilde{u}^i &= a \int_0^\infty J_1(ma) u^i dm \\ \tilde{w}^i &= a \int_0^\infty J_1(ma) w^i dm \end{aligned} \quad (9)$$

### 3. Probabilistic Closed Form Solution

The partial derivative method was used for developing closed form probabilistic solutions for the stationary load program.

The method involves expanding the desired function (stress, strain or deflection) in a Taylor's Series expansion about the mean and neglecting all moments greater than second order. For example, letting  $S_i$  represent the desired stress or displacement at the  $i^{\text{th}}$  layer of the pavement system and  $E_1, E_2, \dots, E_n$  the instantaneous values of the relaxation modulus of  $N$ -layers of different materials at the loading time of interest, one can write

$$S_i = g_i(E_1, E_2, \dots, E_n) \quad (10)$$

The expected value of  $S_i$  may be approximated by the expression

$$E[S_i] = g_i(\bar{E}_1, \bar{E}_2, \dots, \bar{E}_n) + \frac{1}{2} \sum_{j=1}^n \left. \frac{\partial^2 g_i}{\partial E_j^2} \right|_{\bar{E}_j} \sigma_{E_j}^2 \quad (11)$$

The variance of  $S_i$  is given by

$$\text{Var}[S_i] = \sum_{j=1}^n \left( \left. \frac{\partial g_i}{\partial E_j} \right|_{\bar{E}_j} \right)^2 \sigma_{E_j}^2 \quad (12)$$

The term  $g_i(\bar{E}_1, \bar{E}_2, \dots, \bar{E}_n)$  represents the mean value of  $S_i$ , i.e., the value of  $S_i$  obtained using mean moduli values  $\bar{E}_i$ . The terms

$$\left. \frac{\partial^2 g_i}{\partial E_j^2} \right|_{\bar{E}_j} \sigma_{E_j}^2 \quad \text{and} \quad \left( \left. \frac{\partial g_i}{\partial E_j} \right|_{\bar{E}_j} \right)^2 \sigma_{E_j}^2$$

are the 2<sup>nd</sup> and 1<sup>st</sup> partial derivatives of  $S_i$  with respect to  $E_j$  evaluated at  $\bar{E}_j$  and multiplied by  $\sigma_{E_j}^2$  respectively. The  $\bar{E}_j$  and  $\sigma_{E_j}^2$  are assumed to

have been determined from an experimental characterization program.

Equation (11) and (12) were used assuming that  $E_i$  are mutually independent.

From Eq. (1)

$$\frac{\partial g_i}{\partial E_j} = \frac{\partial S_i}{\partial E_j} = \frac{\partial K(v_i, E_i)}{\partial E_j} M(z_i, v_i) D(z) \begin{bmatrix} A_i \\ B_i \\ C_i \\ D_i \end{bmatrix}$$

$$+ K(v_i, E_i) M(z_i, v_i) D(z) \begin{bmatrix} \frac{\partial A_i}{\partial E_j} \\ \frac{\partial B_i}{\partial E_j} \\ \frac{\partial C_i}{\partial E_j} \\ \frac{\partial D_i}{\partial E_j} \end{bmatrix} \quad (13)$$

$$\frac{\partial^2 g_i}{\partial E_j^2} = \frac{\partial^2 S_i}{\partial E_j^2} = \frac{\partial^2 K(v_i, E_i)}{\partial E_j^2} M(z_i, v_i) D(z) \begin{bmatrix} A_i \\ B_i \\ C_i \\ D_i \end{bmatrix}$$

$$+ 2 \frac{\partial K(v_i, E_i)}{\partial E_j} M(z_i, v_i) D(z) \begin{bmatrix} \frac{\partial A_i}{\partial E_j} \\ \frac{\partial B_i}{\partial E_j} \\ \frac{\partial C_i}{\partial E_j} \\ \frac{\partial D_i}{\partial E_j} \end{bmatrix}$$

$$+ K(v_i, E_i) M(z_i, v_i) D(z) \begin{bmatrix} \frac{\partial^2 A_i}{\partial E_j^2} \\ \frac{\partial^2 B_i}{\partial E_j^2} \\ \frac{\partial^2 C_i}{\partial E_j^2} \\ \frac{\partial^2 D_i}{\partial E_j^2} \end{bmatrix} \quad (14)$$

where

$$\frac{\partial K(v_i, E_i)}{\partial E_j} = \begin{bmatrix} 0 & \dots & \dots & \dots \\ \dots & 0 & \dots & \dots \\ \dots & \dots & \frac{-(1+\nu)}{E_j^2} \delta_{ij} m J_1(mr) & \dots \\ \dots & \dots & \dots & \frac{+(1+\nu)}{E_j^2} \delta_{ij} m J_0(mr) \end{bmatrix} \quad (15)$$

$$\frac{\partial^2 K(v_i, E_i)}{\partial E_j^2} = \begin{bmatrix} 0 & \dots & \dots & \dots \\ \dots & 0 & \dots & \dots \\ \dots & \dots & \frac{+2(1+\nu)}{E_j^3} \delta_{ij} m J_1(mr) & \dots \\ \dots & \dots & \dots & \frac{-2(1+\nu)}{E_j^3} \delta_{ij} m J_0(mr) \end{bmatrix} \quad (16)$$

and

$$\delta_{ij} = \begin{cases} 1 & \text{when } i=j \\ 0 & \text{when } i \neq j \end{cases}$$

Differentiating Equation (6) and rearranging the terms

$$\begin{aligned}
 & \sum_{j=1}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ \frac{\partial C_n}{\partial E_j} \\ \frac{\partial D_n}{\partial E_j} \end{bmatrix} = \\
 & \sum_{\ell=1}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial X_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)} \prod_{\substack{j=1 \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ C_n \\ D_n \end{bmatrix} \quad (17)
 \end{aligned}$$

where  $\frac{\partial X_i}{\partial E_j}$  is defined in Equation (18)

$$\frac{\partial x_i}{\partial E_j} = \begin{bmatrix} -\frac{\partial L_i}{\partial E_j} & \frac{\partial L_i}{\partial E_j} \cdot q(-h_i, v_{i+1}, v_i) & -(2mh_i + 4v_i - 1) \frac{\partial L_i}{\partial E_j} & -\frac{\partial L_i}{\partial E_j} q(-h_i, v_{i+1}, v_i) \\ 0 & \frac{\partial L_i}{\partial E_j} (4v_{i+1} - 3) & 2 \frac{\partial L_i}{\partial E_j} & (2mh_i - 4v_{i+1} + 1) \frac{\partial L_i}{\partial E_j} \\ (2mh_i - 4v_i + 1) \frac{\partial L_i}{\partial E_j} & \frac{\partial L_i}{\partial E_j} p(h_i, v_{i+1}, v_i) & -\frac{\partial L_i}{\partial E_j} & -\frac{\partial L_i}{\partial E_j} q(h_i, v_{i+1}, v_i) \\ -2 \frac{\partial L_i}{\partial E_j} & -(2mh_i + 4v_{i+1} - 1) \frac{\partial L_i}{\partial E_j} & 0 & (4v_{i+1} - 3) \frac{\partial L_i}{\partial E_j} \end{bmatrix} \quad (18)$$

where

$$\frac{\partial L_i}{\partial E_j} = \frac{\partial}{\partial E_j} \left[ \frac{\frac{E_i}{E_{i+1}} \frac{1+v_{i+1}}{1+v_i}}{\frac{\partial E_j}{\partial E_j}} \right] = \left( \frac{1+v_{i+1}}{1+v_i} \right) \left[ \frac{1}{E_{i+1}} \delta_{ij} - \frac{E_i}{E_{i+1}^2} \delta_{\alpha j} \right] \quad ; \alpha = i+1$$

$$p(h, v, \mu) = m^2(2h^2) + 4mh(v-\mu) + (1-8v\mu + 2\mu)$$

$$q(h, v, \mu) = 2mh(2v-1) - (1 + 8v\mu - 6\mu)$$

From Equation (5) we obtain

$$\begin{bmatrix}
 \frac{\partial A_i}{\partial E_j} \\
 \frac{\partial B_i}{\partial E_j} \\
 \frac{\partial C_i}{\partial E_j} \\
 \frac{\partial D_i}{\partial E_j}
 \end{bmatrix} = \frac{n-1}{\prod_{j=1}^n \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)}} \begin{bmatrix}
 0 \\
 0 \\
 \frac{\partial C_n}{\partial E_j} \\
 \frac{\partial D_n}{\partial E_j}
 \end{bmatrix} \\
 + \frac{n-1}{\sum_{\ell=i}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial x_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)}} \sum_{\substack{j=i \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)} \begin{bmatrix}
 0 \\
 0 \\
 C_n \\
 D_n
 \end{bmatrix} \quad (19)$$

Since  $C_n, D_n, A_i, B_i, C_i, D_i$  are determined using the mean values,  $\bar{E}_i, \frac{\partial C_n}{\partial E_j}, \frac{\partial D_n}{\partial E_j}, \frac{\partial A_i}{\partial E_j}, \frac{\partial B_i}{\partial E_j}, \frac{\partial C_i}{\partial E_j}, \frac{\partial D_i}{\partial E_j}$  ( $i = 1, \dots, n-1$ ) may be determined using Equations (17) and (19). The second partial derivatives of  $S_i$  with respect to  $E_j$  are obtained from Equations (17) and (19).

$$Q \frac{n-1}{\prod_{j=1}^n \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)}} \begin{bmatrix}
 0 \\
 0 \\
 \frac{\partial^2 C_n}{\partial E_j^2} \\
 \frac{\partial^2 D_n}{\partial E_j^2}
 \end{bmatrix}$$

$$= - 2 \sum_{\ell=1}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial x_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{j=1 \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ \frac{\partial C_n}{\partial E_j} \\ \frac{D_n}{E_j} \end{bmatrix}$$

$$- \sum_{\ell=1}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial^2 x}{\partial E_j^2} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{j=1 \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ C_n \\ D_n \end{bmatrix}$$

$$- \sum_{\ell=1}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial x_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{k=1 \\ k \neq \ell}}^{n-1} \frac{D^{-1}(h_k) \frac{\partial x_k}{\partial E_j} D(h_k)}{4(v_k - 1)}$$

$$\sum_{\substack{j=1 \\ j \neq k \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) x_j D(h_j)}{4(v_j - 1)} \begin{bmatrix} 0 \\ 0 \\ C_n \\ D_n \end{bmatrix} \quad (20)$$

and

$$\begin{aligned}
& \left[ \begin{array}{c} \frac{\partial^2 A_i}{\partial E_j^2} \\ \frac{\partial^2 B_i}{\partial E_j^2} \\ \frac{\partial^2 C_i}{\partial E_j^2} \\ \frac{\partial^2 D_i}{\partial E_j^2} \end{array} \right] = \sum_{j=1}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \left[ \begin{array}{c} 0 \\ 0 \\ \frac{\partial^2 C_n}{\partial E_j^2} \\ \frac{\partial^2 D_n}{\partial E_j^2} \end{array} \right] \\
& + 2 \sum_{\ell=i}^{n-1} \frac{D(h_\ell) \frac{\partial X_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{j=i \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \left[ \begin{array}{c} 0 \\ 0 \\ \frac{\partial C_n}{\partial E_j} \\ \frac{\partial D_n}{\partial E_j} \end{array} \right] \\
& + \sum_{\ell=i}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial^2 X_\ell}{\partial E_j^2} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{j=i \\ j \neq \ell}}^{n-1} \frac{D^{-1}(h_j) X_j D(h_j)}{4(v_j - 1)} \left[ \begin{array}{c} 0 \\ 0 \\ C_n \\ D_n \end{array} \right] \\
& + \sum_{\ell=i}^{n-1} \frac{D^{-1}(h_\ell) \frac{\partial X_\ell}{\partial E_j} D(h_\ell)}{4(v_\ell - 1)} \sum_{\substack{k=i \\ k \neq \ell}}^{n-1} \frac{D^{-1}(h_k) \frac{\partial X_k}{\partial E_j} D(h_k)}{4(v_k - 1)} \left[ \begin{array}{c} 0 \\ 0 \\ C_n \\ D_n \end{array} \right]
\end{aligned}
\tag{21}$$

where

$$\begin{aligned}
 \frac{\partial^2 \chi_i}{\partial E_j^2} = & \begin{bmatrix}
 -\frac{\partial^2 L_i}{\partial E_j^2} & \frac{\partial^2 L_i}{\partial E_j^2} q(-h_i, v_{i+1}, v_i) & -\frac{\partial^2 L_i}{\partial E_j^2} & -\frac{\partial^2 L_i}{\partial E_j^2} (-h_i, v_{i+1}, v_i) \\
 0 & \frac{\partial^2 L_i}{\partial E_j^2} (4v_{i+1} - 3) & 2 \frac{\partial^2 L_i}{\partial E_j^2} & (2mh_i - 4v_{i+1} + 1) \cdot \frac{\partial^2 L_i}{\partial E_j^2} \\
 (2mh_i - 4v_i + 1) \frac{\partial^2 L_i}{\partial E_j^2} & \frac{\partial^2 L_i}{\partial E_j^2} p(h_i, v_{i+1}, v_i) & -\frac{\partial^2 L_i}{\partial E_j^2} & -\frac{\partial^2 L_i}{\partial E_j^2} q(h_i, v_{i+1}, v_i) \\
 -2 \frac{\partial^2 L_i}{\partial E_j^2} & -(2mh_i + 4v_{i+1} - 1) \frac{\partial^2 L_i}{\partial E_j^2} & 0 & \frac{\partial^2 L_i}{\partial E_j^2}
 \end{bmatrix} \quad (22)
 \end{aligned}$$

and  $\frac{\partial^2 L_i}{\partial E_j^2} = 2 \left( \frac{1 + v_{i+1}}{1 + v_i} \right) \left[ \frac{-1}{E_{i+1}^2} \delta_{ij} + \frac{E_i}{E_{i+1}^3} \delta_{\alpha j} \right] ; \quad \alpha = i+1$

Now that  $\frac{\partial A_i}{\partial E_j}$ ,  $\frac{\partial B_i}{\partial E_j}$ ,  $\frac{\partial^2 C_i}{\partial E_j^2}$ ,  $\frac{\partial^2 D_i}{\partial E_j^2}$  are known, these may be substituted into

Equations (13) and (14) and then (11) and (12) to obtain  $E[S_i]$  and  $\text{Var}[S_i]$ .

To obtain the solution for a circular load, the stresses or displacement must be integrated from zero to infinity with respect to  $m$  and multiplied by  $a$ . Hence

$$\begin{aligned}
 E[\tilde{S}_i] &= a E \left[ \int_0^\infty J_1(ma) S_i dm \right] \\
 &\approx a E \left[ \sum_{j=1}^N J_1(ma)_j S_{ij} \Delta(m_j) \right] \\
 &\approx a \left[ \sum_{j=1}^m J_1(ma)_j E[S_i]_j \Delta(m_j) \right] \\
 &= a \int_0^\infty J_1(ma) E[S_i] dm \tag{23}
 \end{aligned}$$

where  $J_1(ma)_j$ ,  $S_{ij}$  denote  $J_1(ma)$ ,  $S_i$  evaluated at  $m = m_j$ .

Similarly,

$$\begin{aligned}
 \text{VAR} [\tilde{S}_i] &= \sum_{j=1}^N (\sigma_j)^2 a^2 \left[ \frac{\partial}{\partial E_j} \int_0^\infty J_1(ma) S_i(E_j, m) dm \right]^2 \\
 &= \sum_{j=1}^N (\sigma_j)^2 a^2 \left[ \int_0^\infty J_1(ma) \frac{\partial S_i(E_j, m)}{\partial E_j} dm \right]^2 \tag{24}
 \end{aligned}$$

APPENDIX 4  
PROGRAM LISTING

NOTE: This part is included in a separate volume.