

ACCURACY OF A NEW METHOD FOR MEASUREMENT OF
AN ACOUSTIC IMPEDANCE

A THESIS

Presented to

The Faculty of the Graduate Division

by

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In Partial Fulfillment

of the Requirements for the Degree

Master of Science

In the School of Aerospace Engineering

Georgia Institute of Technology

August, 1970

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Date approved by Chairman: 11/5/70

ACKNOWLEDGMENTS

The author would like to express here his appreciation to his thesis advisor, Dr. Warren C. Strahle, for his guidance and understanding, and for his encouragement in the difficult periods of the research.

Also appreciated is the work of Mr. Brady R. Daniel who was instrumental in the construction and operation of the experimental apparatus. He generously donated many free hours of time toward the completion of this project. The contribution of Mr. Charlie R. Lord of the electronics laboratory is also acknowledged.

Dr. Ben T. Zinn and Dr. James E. Hubbartt are due a special note of thanks for their sincere interest and constructive comments as readers.

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SUMMARY

Two possible experimental procedures are described for the measurement of the acoustic impedance of a vaporizing liquid. Theoretical investigation of the first procedure shows that the use of a vaporizing spray in an impedance tube would require experimental precision far beyond present capabilities. Consequently this approach is rejected. The second procedure involves an impedance tube terminated by a plane vaporizing surface. The design of this experiment is described and initial experiments to determine the impedance of the end plate assembly without mean air flow and without a vaporizing fluid are performed. It is found that the experimental precision of the impedance measurement is rather poor. Reasons for the imprecise results are deduced and recommendations for experiment improvement are given.

NOTATION

A	complex constant
B	complex constant
c	mean speed of sound
D	diffusion coefficient
D_1	diameter of a droplet
f	frequency
h	height of the plexiglass wall
i	imaginary number $\sqrt{-1}$
K	proportional to the wave number, defined by K_r and K_i
K_r	dimensionless circular frequency, $K_r = \frac{\omega L}{c}$
K_i	decay factor
L	length of the tube
L_v	heat of vaporization
\tilde{m}	instantaneous mass flow rate
\bar{m}	mean mass flow rate
m	perturbation of the mass flow rate, $m = \tilde{m} - \bar{m}$
M	mach number
\dot{M}	mass flow rate vaporizing per droplet
\bar{P}	ambient pressure
\tilde{p}	instantaneous pressure at a point
p	perturbation of the pressure at a point, or acoustic pressure, $p = \tilde{p} - \bar{P}$

P	dimensionless acoustic pressure, $P = p/\bar{p}$
P'	number defined by $P' = P e^{i\psi}$
Re	Reynolds number
Sc	Schmidt number
t	time
T	temperature
\tilde{u}	instantaneous speed of the gas
u	perturbation of the speed of the gas
U	dimensionless perturbation of the speed of the gas, $U = u/c$
x	abscissa
X	dimensionless abscissa, $X = x/L$
Y	number defined by $Y = \frac{P_1' + P_3'}{2P_2'} = \frac{p_1 + p_3}{2p_2}$
z	acoustic impedance, $z = p/u$
Z	dimensionless acoustic impedance, $Z = \frac{P}{u} = \frac{zc}{\bar{p}}$
zz	quantity defined by equation (9)
γ	ratio of specific heats
μ	viscosity
ψ	phase of P , taking the phase of P_1 (at X_1) as zero
Ψ	complex constant
$\tilde{\rho}$	instantaneous density
$\bar{\rho}$	mean density
ρ	perturbation of the density
ω	circular frequency
Ω	dimensionless circular frequency, $\Omega = \frac{\omega L}{c}$

Subscripts:

- v vaporization
- j index related to the three transducers used in the
 experiment

CHAPTER I

INTRODUCTION

Several theoretical treatments have appeared in the literature¹⁻⁴ concerning the change of vaporization rate of a liquid due to acoustic waves in the ambient fluid. However, no experiments have been successfully designed to measure this effect. This thesis explores two methods.

Both methods use an impedance tube, which is an enclosure where a system of one dimensional standing waves is produced; then measurements of the pressures of various locations enable the acoustic impedance to be computed at any point of this tube. In particular, if a liquid vaporizes at a given point of the tube in a mean one dimensional air flow, it is possible to compute the acoustic impedance at this point for any given vaporization rate.

The vaporization process of interest is one which occurs in flight vehicle engines in the form of a vaporizing spray. An initial attempt is made to model this situation experimentally. It is found theoretically in the second chapter that the influence of the spray on the acoustic wave pattern is too small to be evaluated. This method is therefore to be rejected.

A second approach explored is to use an impedance tube

terminated by a flat porous plate through which a vaporizing fluid is allowed to pass. It can be shown that the presence of a vaporization process induces a small but noticeable change of the acoustic impedance of the flat porous plate. It is therefore important to be able to determine this impedance with precision, even without vaporization.

The third chapter of this thesis is concerned with the design of the experiment, including the vaporization devices. The fourth and fifth chapters present measurements of the end plate impedance without mean air flow in the tube, and without the flow of a vaporizing fluid. The measurements are taken using a new method similar to the one investigated in Ref. 5. Several fixed microphones are employed, instead of the classical single traversing microphone. The fifth chapter presents results of this acoustic experiment, and remarks on their accuracy.

CHAPTER II

EVALUATION OF THE FIRST EXPERIMENT

The first method investigated is depicted in Figure 1. A low speed air flow (240 in/sec) passes horizontally between two plexiglass walls. Perpendicular to the flow a spray of droplets is introduced by a vibrating hypodermic needle⁶. The droplets vaporize, as they fall due to gravity, and the flow passage is divided, therefore, into region I containing air and region II containing air and vapor. Acoustic waves are produced by an acoustic driver located at the left hand end.

The feasibility of determining the effect of the vaporizing spray may be determined theoretically. Since $\tilde{m}_2 = \tilde{\rho}_2 \tilde{u}_2$ at point 2 downstream of the droplets, if small amplitude disturbances are allowed

$$m_2 = \rho_2 \tilde{u}_2 + \bar{\rho}_2 u_2 \quad (1)$$

This equation may be transformed, using three relations. Assume isentropic oscillations $P_2/\bar{P} = \gamma(\rho_2/\bar{\rho}_2)$. Note that the acoustic pressure is assumed the same on both sides of the droplets, since 1). The momentum equation applied to a small control surface containing the droplets and their vaporization process⁸ shows that the change in pressure across

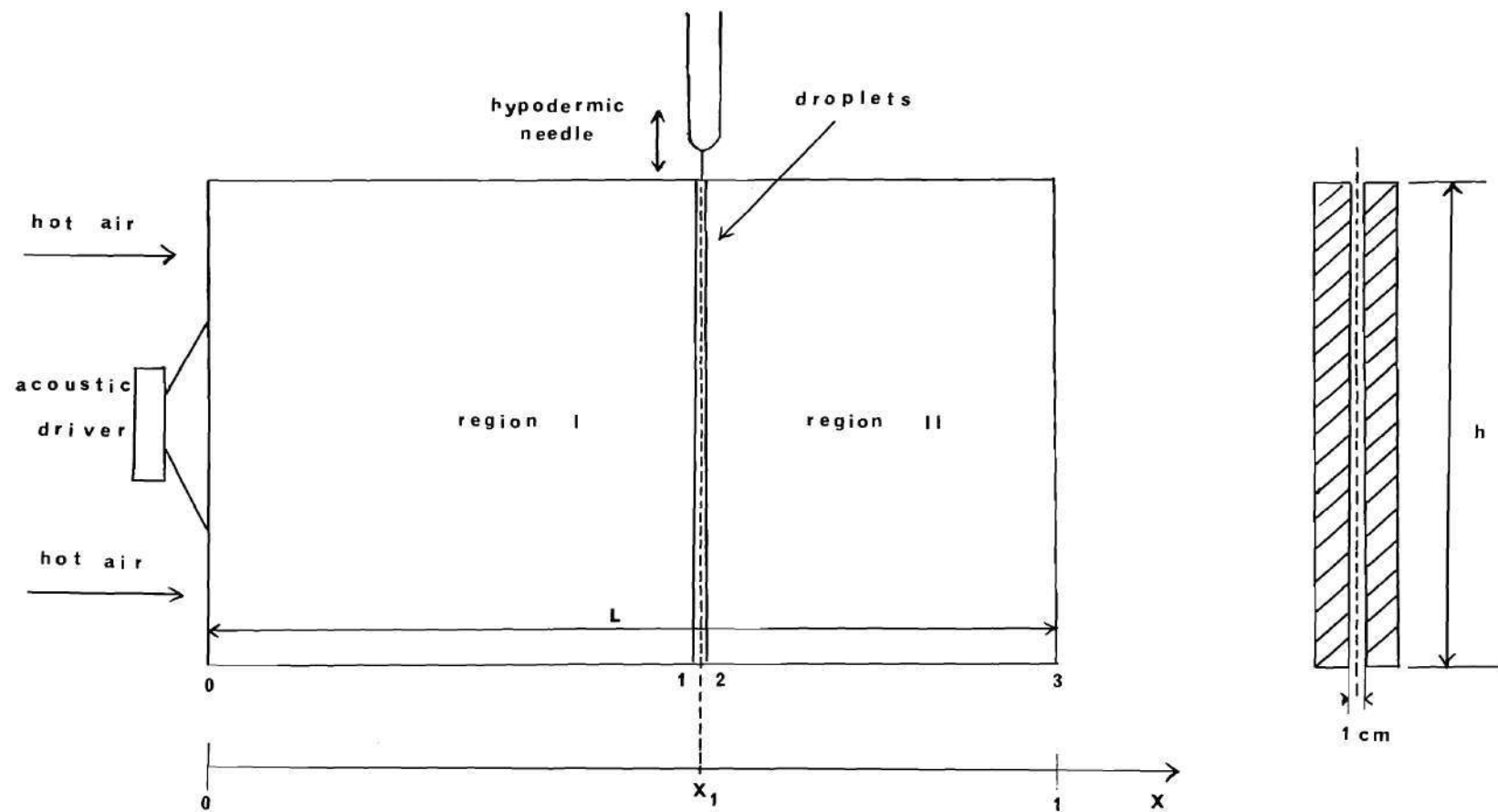


Figure 1. First Experiment

the droplets depends on the change in Mach number. This change being very low, as shown later, a strong expansion wave does not exist; 2). The wavelength of the traveling wave is very large compared to the control surface considered so that there are no wave propagation phenomena in the control volume. Furthermore, wave scattering from the droplets is negligible. Thus, $p_{II} \approx p_I$. Finally, use the definition of the dimensionless acoustic impedance, $Z = \frac{p/\bar{p}}{u/\bar{c}}$. Equation (1) then becomes

$$\frac{m_v/\bar{m}_v}{p/\bar{p}} = \frac{1}{\gamma} + \frac{\bar{m}_2}{\bar{m}_v} \frac{1}{M_2 Z_2} - \frac{\bar{m}_1}{\bar{m}_v} \frac{1}{M_1 Z_1} \quad (2)$$

The quantity on the left hand side of Equation (2) is the quantity of interest. From theory¹ it is dependent upon frequency and is a quantity of the order of unity. Measurable quantities on the right hand side of Equation (2) are Z_1 and Z_2 .

Assuming a pressure wave in region I of the form

$$P_I = A_I \sin(\Omega_I X - \Psi_I) e^{i\omega t}$$

where P_I , A_I , Ψ_I are complex numbers, and a corresponding waveform in region II, consider the determination of Z_1 and Z_2 . The acoustic impedance of the end of the tube determines the value of Ψ_{II} . For convenience in computation, assume that the end of the tube behaves as an ideal open end with a zero

impedance. Then, at $X = 1$, $P_{II} = 0$ or $\sin(\Omega_{II} - \Psi_{II}) = 0$. Consequently, $\Psi_{II} = \Omega_{II}$ or $\Psi_{II} = \Omega_{II} + \pi$. At $X = X_1$, the relation $P_{II} = P_I$ proved earlier gives A_{II} . Finally,

$$P_{II} = \frac{A_I \sin[\Omega_I X_1 - \Psi_I]}{\sin[\Omega_{II}(X_1 - 1)]} \sin[\Omega_{II}(X - 1)] e^{i\omega t}$$

Using the acoustic equation⁷ $\partial P / \partial X = -\bar{\rho} \partial u / \partial t$

$$\frac{u}{c} = - \frac{\bar{P} A}{i \bar{\rho} c^2} \cos[\Omega X - \Psi] e^{i\omega t}$$

so that, from the definition of the dimensionless acoustic impedance, $Z = -i\gamma \tan(\Omega X - \Psi)$. Consequently, Equation (2) becomes

$$\frac{m_v / \bar{m}_v}{P / \bar{P}} = \frac{1}{\gamma} \left\{ 1 + \frac{i \bar{m}_2}{\bar{m}_v M_2} \left[\frac{1}{\tan[\Omega_{II}(X_1 - 1)]} - \frac{c_2}{c_1} \frac{1}{\tan[\Omega_I X_1 - \Psi_I]} \right] \right\} \quad (3)$$

It now remains to be seen whether normal experimental errors would yield an acceptable computation of the ratio $\frac{m_v / \bar{m}_v}{P / \bar{P}}$. The measurable acoustic quantities in Equation (3) are c_1 , c_2 , Ω_I , Ω_{II} , and Ψ_I . To evaluate the ratio \bar{m}_v / \bar{m}_2 , consider the following development.

Consider first droplets of 750 microns in diameter since Dabora⁶ has determined that droplets of this size remain equally spaced during their downward travel. That is, there

is no risk of droplet coalescence. For computational simplicity assume the liquid is water at 212°F and the air is at 900°R and 1 atm. The formula for the vaporization rate in dry air is⁸

$$\dot{M} = 2\eta D_\ell \mu \ln \left[1 + \frac{c_p (T - T_\ell)}{L_v} \right] (1 + 0.276 \text{ Re}^{1/2} \text{ Sc}^{1/2}) \quad (4)$$

The following numerical values are chosen from Reference 9.

$$\begin{aligned} \mu \text{ (water vapor)} &= 4.42 \times 10^{-2} \text{ lbm/ft/hr} \\ \mu \text{ (air)} &= 6.6 \times 10^{-2} \text{ lbm/ft/hr} \\ L_v &= 1205.5 \text{ BTU/lbm} \\ c_p &= 6007 \text{ BTU/lbm/°F} \\ \rho_{\text{air}} &= 0.113 \times 10^{-4} \text{ lbm/ft}^3 \\ D_{\text{vapor-air}} &= 1.337 \text{ ft}^2/\text{hr} \end{aligned}$$

Consequently, $\text{Sc} = 1.117$, $\text{Re} = 19.5$ and, using Equation (4), $\dot{M} = 7 \times 10^{-8} \text{ lbm/sec}$. This is the initial vaporization rate. Assuming the droplet velocity in the Reynolds number in Equation (4) to remain unchanged and noting $\dot{M} = -\rho_L \frac{d}{dt}(\pi D_\ell^3/6)$ an integration for the lifetime of the droplet yields 14.1 sec. A simpler formula for the lifetime results if the Reynolds number is assumed constant in Equation (4). In this case the lifetime is 9.68 sec which is of the same order of magnitude as the more precise estimate. According to Dabora⁶ the liquid velocity at the needle exit should equal the Stoke's velocity (drag = weight). This velocity of 36.7 ft/sec

can be obtained pressurizing the fluid ahead of the orifice. Integrating Stokes formula to account for the changing droplet size it is found that for a fall of 2 inches the time required is 0.00446 sec and the droplet size at the end of the fall is $D_1 = 749.8$ microns. For a fall of 40 in (1 m) the time is 0.0896 sec and the diameter has only decreased to 746.5 microns. In these computations the air velocity has been assumed small in comparison with the droplet velocity. To produce these droplets by a vibrating needle requires a vibration frequency of 1100 cycles per second⁶, which is also the droplet number production rate. For a 40 in/sec (1 m/sec) air flow it is therefore determined that $\bar{m}_v/\bar{m}_1 = 4.8 \times 10^{-4}$ if the distance between the plexiglass walls is about 0.4 in (1 cm), a reasonable minimum distance to prevent splashing of the spray on the walls. A higher speed airflow would result in downstream displacement of the droplets and destruction of the one dimensional situation, although it would increase the vapor production. Using multiple needles would be possible but there are experimental difficulties in making them all perform properly⁶. It therefore appears that the experiment is constrained to operate with small mass flow ratios, say $\bar{m}_v/\bar{m}_1 \approx \bar{m}_v/\bar{m}_2 < 10^{-3}$.

Repeating the calculation for 100 μ droplets yields $\bar{m}_v/\bar{m}_2 = 8.3 \times 10^{-4}$ where in this case vaporization is complete after a fall of only 1.07 in. It is concluded that there is no experimental arrangement of this type to produce \bar{m}_v/\bar{m}_2

greater than roughly 10^{-3} .

Returning to Equation (3) and noting that $M_2 \approx 0.003$, it is seen that the term in braces

$$\left[\frac{1}{\tan[\Omega_{II}(X_1-1)]} - \frac{c_2}{c_1} \frac{1}{\tan[\Omega_I X_1 - \psi_1]} \right]$$

must be small, of the order of 10^{-5} in order that the vaporization ratio $\frac{m_v/\bar{m}_v}{P/\bar{P}}$ be of order unity¹. The temperature is not important in this evaluation: Since $\frac{\bar{m}_2}{\bar{m}_v M_2} = \frac{\bar{p}_2 c_2}{\bar{m}_v}$ which is inversely proportional to the square root of the temperature T_2 , assuming a perfect adiabatic gas, and inversely proportional to \bar{m}_v which is a logarithmic function of T_2 .

The small term in braces is the difference of two terms of order unity which contain numerical errors due to the experimental error. It is concluded that this method is not satisfactory for the measurement of the change in vaporization rate.

CHAPTER III

DESIGN OF THE SECOND EXPERIMENT

The first experiment was shown to be impractical, due to the too low ratio of vaporized mass flow rate to air flow rate. This ratio can be raised to the order of 10^{-1} if the configuration of Figure 2 is used. Here the vaporizing surface which covers the end plate of an impedance tube. Air flows downward toward the plate which is constructed of porous metal through which the vaporizing fluid is forced. Escape of the air and vapor is provided by a cylindrical passage between the tube wall and the end plate. In this second experiment, it can be shown that the presence of a vaporization process induces a small but noticeable change of the acoustic impedance of the flat porous plate. This chapter is concerned with the design of the experiment, including the vaporization devices and the acoustic apparatus. The following chapters will discuss the acoustic problem and measurements.

In order to match the Reynolds number of droplets in a rocket engine, a constraint on the design was selected as $10 \leq Re \leq 100$, where the reference length is the end plate radius, selected as 3 in. For ambient air the velocity in the tube approaching the end plate varies from 0.16 to 1.6

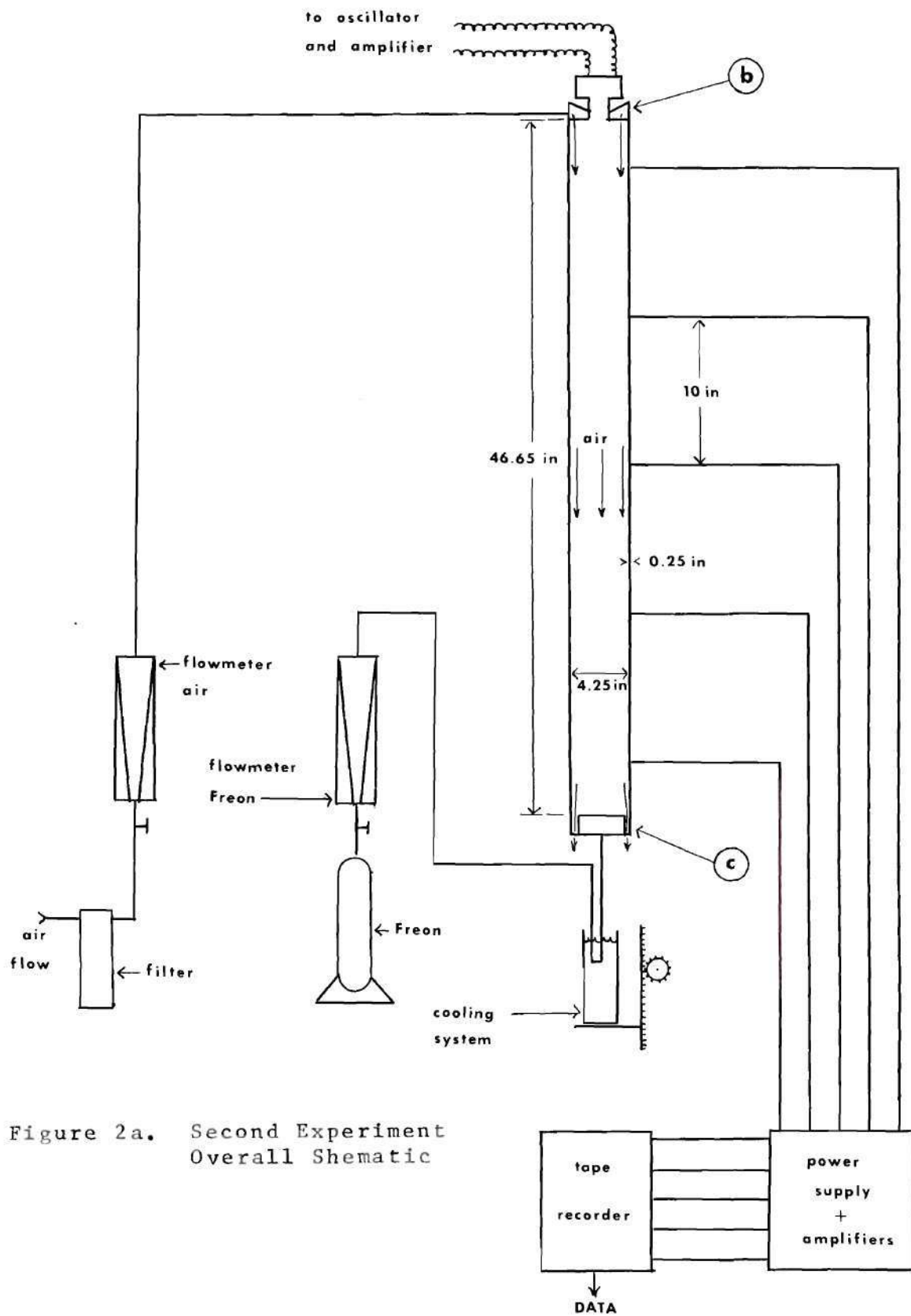


Figure 2a. Second Experiment
Overall Schematic

in/sec as the Reynolds number goes from 10 to 100. Obviously, the velocities occurring in a rocket chamber are not matched. The maximum air flow rate is $0.77 \text{ ft}^3/\text{min}$.

Freon 114 was selected as the most promising non-toxic, non flammable fluid as the vaporizing substance. Freon 114 has the lowest storage pressure at ambient temperature of the Freon family. It has a boiling point at one atmosphere of 38.4°F and a vapor pressure at 72°F of 2 atm. Further data on Freon 114 are given in Reference 10. The storage pressure in standard Freon bottles is slightly above the vapor pressure at the ambient temperature and this vapor pressure is used to pump the Freon. Before the expansion of the Freon from the storage bottle to the atmosphere, it is necessary that the Freon be cooled below its boiling point, as shown in Figure 3. The path consists of cooling at constant pressure, Then expansion of the liquid to 1 atm, by heat transfer from the experimental apparatus. A schematic of the cooling system (at constant pressure) is shown in Figure 4.

The flow rate required can be determined using the method of Reference 11, in which it is assumed that the end plate surface behaves as a stagnation point. Using this method, the estimated flow rate is $0.106 \text{ cm}^3/\text{min}$ at an air flow Reynolds number of 10, and $0.335 \text{ cm}^3/\text{min}$ at an air flow Reynolds number of 100 (Recall that $10 \leq \text{Re} \leq 100$ to match the Reynolds number of droplets in a rocket engine). The flow-

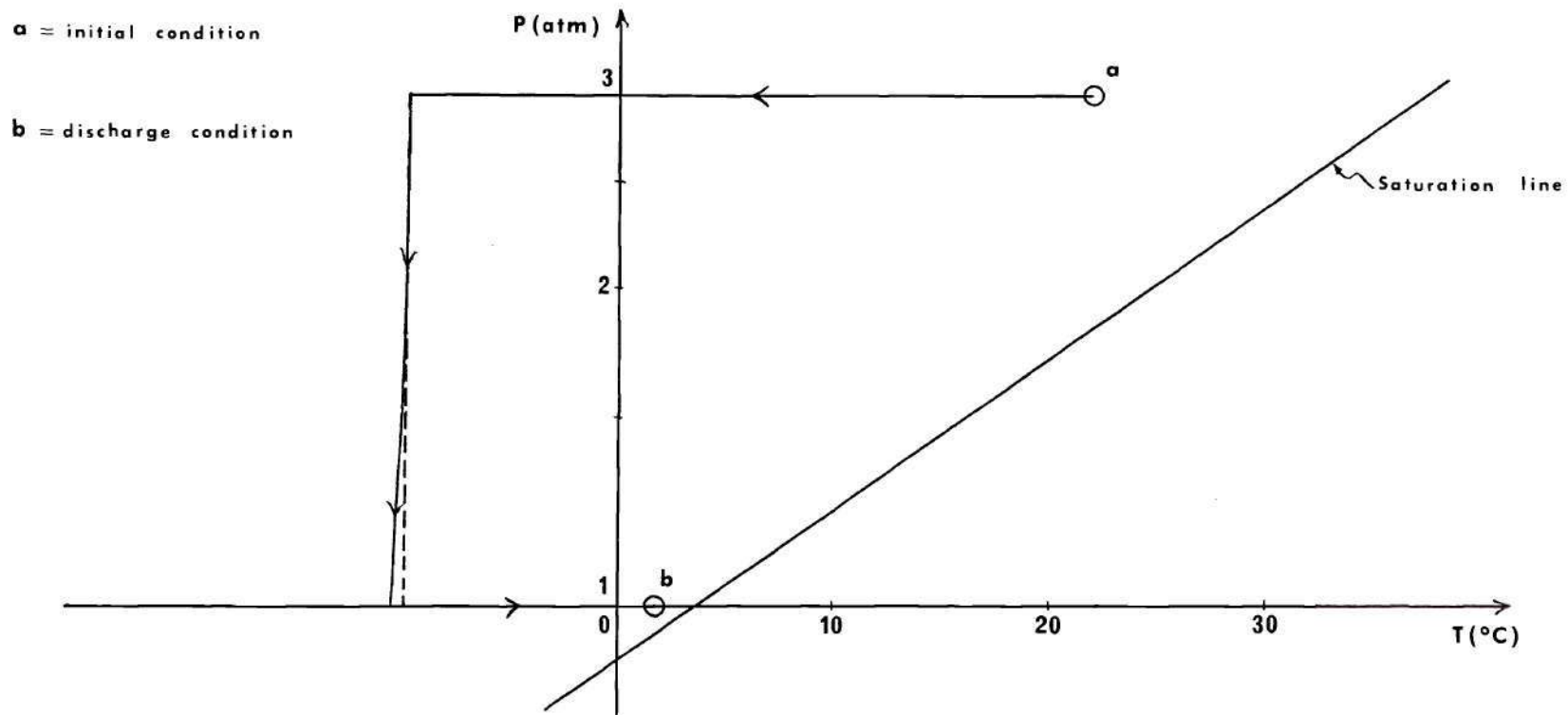


Figure 3. Temperature vs. Pressure for Freon 114

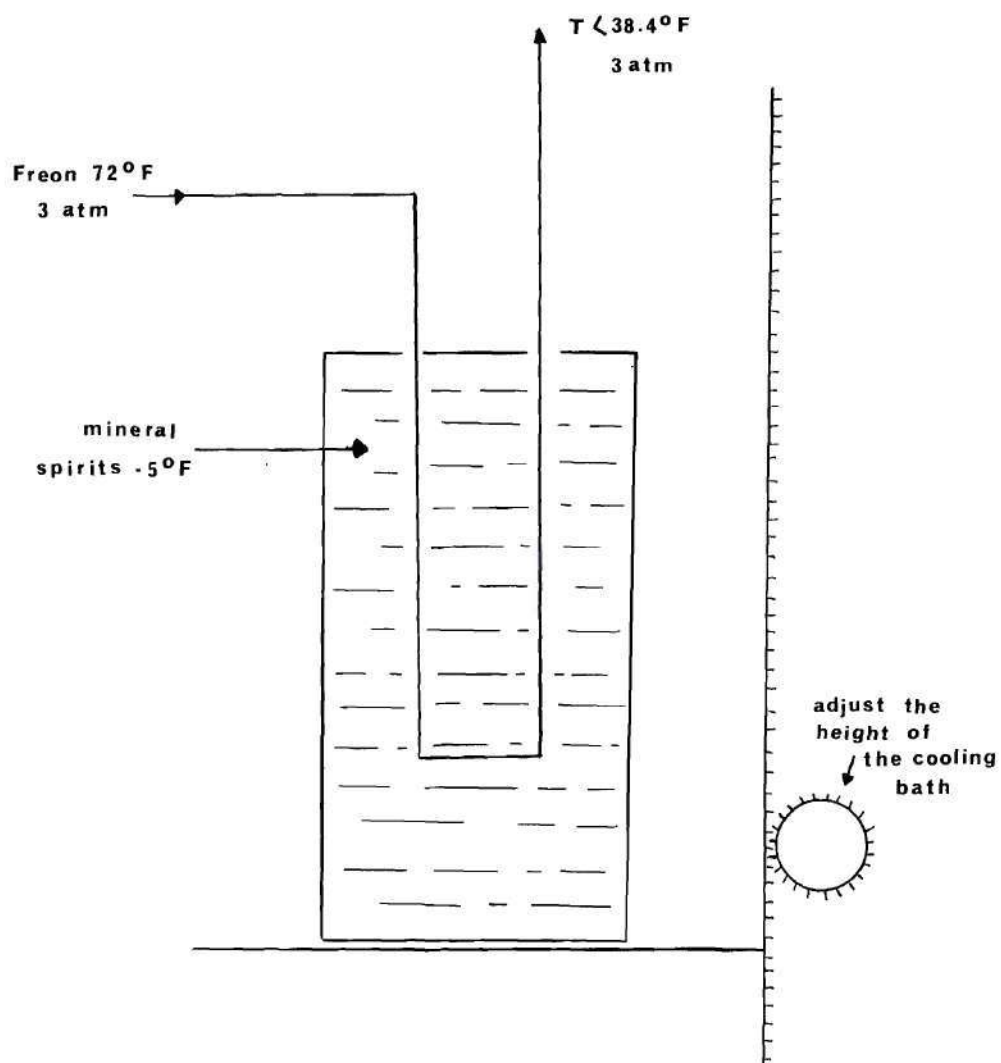


Figure 4. Design of the Cooling System

meter required for the Freon can be roughly sized using these figures.

Actual construction of the air and Freon flow systems were not included in the scope of this thesis. However, the acoustic system was completed. Sound generation is performed by a University 50 watt, 16 ohm driver which yields a satisfactory signal above 100 cps. A Dynasciences Corporation, Model PC 125, oscillator is used to generate a sine wave of a selected frequency and the signal is fed to a Krohn-Hite amplifier, Model DCA, 50R. To adapt the amplifier impedance to the drive impedance a Krohn-Hite matching transformer, Model MT-56, is used. As shown later, the acoustic pressure is measured at five points as located in Figure 2. Bruel and Kjaer condenser microphones, Type 4135, are used. Calibration of the transducers is accomplished with a Bruel and Kjaer Microphone Amplifier, Type 2604. The 5 power supplies and amplifiers for the pressure transducers are grouped into a compact single unit. Each channel has seven attenuation settings ranging from gains of 1 to 1000, in increments of a factor of $\sqrt{10}$. To record the signals an Ampex, Model FR 1300, 14 channel tape recorder is used. A two channel Tektronix, Model 535A, oscilloscope is used to determine appropriate amplifier gain settings, determine resonance frequencies, check signal distortion and check phase differences between two signals.

From the recorded signals it is necessary to determine

the amplitudes and phases of the five channels used. The output voltage is read by a digital voltmeter, Data Technology Corporation, Model 360B. The phase difference between two channels is determined using an Aerometrics, Model PM 730, phasemeter. The error in phase angle is roughly $\pm 0.7^\circ$. The primary error in the amplitude arises from a $\pm 2\%$ error in the tape recorder.

Calibration of the acoustic system is accomplished by introducing a known acoustic source to each microphone. The signals used ranged from 110 to 160 db re 2×10^{-4} μ bar. 180 db is the maximum signal rating for the transducers and below 110 db the signal is not strong enough for the tape recorder. Each microphone is provided a correction factor (K) to add to the readings on the Bruel and Kjaer microphone amplifier, which is used as a standard. It was checked with another acoustic standard, a Dynasciences Corporation, Model PC 125, acoustic calibrator and some differences were noted. The K factors were modified in accordance with the results in Appendix I.

To calibrate the channel amplifiers a microphone is placed near the acoustic driver and a db level is determined on the B&K microphone amplifier. The microphone signal is then put through a channel on the five channel box. The output voltage is recorded for all attenuation settings, several db settings, and several frequencies. The frequencies of interest were determined as the resonant frequencies of the

acoustic tube 186, 312, 443, 585, 882, and 1025 cps. The results of this calibration are given in Appendix II. The assumption is then made that the output voltage is linearly transformed to the tape recorder.

A photograph of the experimental setup is shown in Figure 5.

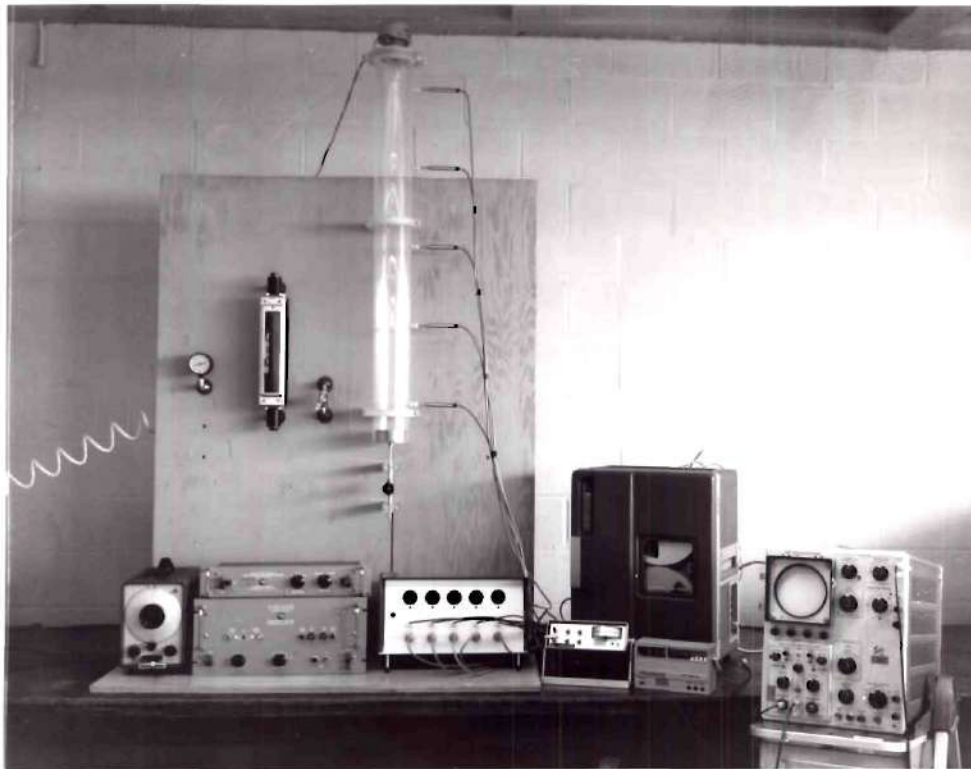


Figure 5. Experimental Setup

CHAPTER IV

THEORY OF THE IMPEDANCE MEASUREMENT

In the second experiment, the presence of a vaporization process induces a small but noticeable change of the acoustic impedance of the flat porous plate. It is therefore important to be able to determine this impedance with precision, even without vaporization. This chapter presents the theory of this determination.

The classical method of measuring the acoustic impedance at the end of a tube uses a traversing microphone⁷. This approach is undesirable in the present program because of the presence of a mean flow and a vaporization in its future version, and the consequent mechanical complexity of the device. An attempt is therefore made to measure to impedance by five fixed transducers.

In the future version, the presence of a mean flow will affect the acoustics of the system⁷, and in particular the fundamental mode in which we are interested. However, the effect is proportional to the Mach number. Since $M \approx 0.003$, the effect is negligible in this experiment. The standing wave in the tube will consist of two traveling waves, the wave produced by the driver and the wave reflected from the tube end⁷. These waves are damped due to wall effects. Using complex notation the general expression for the standing

waveform is given by

$$P = (Ae^{iKX} + Be^{-iKX})e^{i\omega t}$$

with

$$K = K_r + iK_i$$

A and B are also complex numbers. The objective is the measurement of the three unknowns A, B, and K. Note that K_r is unknown because c is unknown in the absence of a temperature measurement.

Let us write the acoustic pressure as $P = |P|e^{i(\omega t + \psi)}$. The time origin may be chosen so that, at $X = X_1$, the location of the top transducer, $\psi_1 = 0$. Thus, measurement of the pressure amplitude $|P|$ and phase relative to transducer number 1 yields a complex number P' . Three such measurements at three different points yield three equations

$$P'_j = Ae^{iKX_j} + Be^{-iKX_j} \quad j = 1, 2, 3 \quad (5)$$

for A, B, and K. Solving for A and B using $j = 1$ and 2,

$$A = \frac{P'_1 e^{-iKX_2} - P'_2 e^{-iKX_1}}{2i \sin K(X_1 - X_2)}$$

$$B = \frac{P'_2 e^{iKX_1} - P'_1 e^{iKX_2}}{2i \sin K(X_1 - X_2)} \quad (6)$$

Placing Equations (6) into $j = 3$ of Equations (5)

$$\begin{aligned}
& P_1' \sin K(X_2 - X_3) + P_2' \sin K(X_3 - X_1) \\
& + P_3' \sin K(X_1 - X_2) = 0
\end{aligned} \tag{7}$$

In general Equation (7) cannot be solved analytically.

However, for equal transducer spacings with $\Delta X = X_3 - X_2 = X_2 - X_1$, there results as a nontrivial solution

$$Y = \cos K \Delta X = \frac{P_1' + P_3'}{2P_2'} = \frac{p_1 + p_3}{2p_2} \tag{8}$$

Equation (8) is the fundamental equation which determines K from the measured P_1' , P_2' and P_3' . Using five transducers there are four possible determinations of K at any given frequency, using the first three, the second three, the last three or the first, third, and last transducers. After some algebra the inversion of Equation (8) for K_r and K_i gives

$$\begin{aligned}
K_i &= \frac{\ln(zz + \sqrt{zz^2 - 1})}{\Delta X} \\
K_r &= \epsilon \frac{\cos^{-1}(Y_r/zz)}{\Delta X} + n \frac{\pi}{\Delta X} \quad n = \text{integer} \\
&\quad \epsilon = \pm 1 \\
zz &= \frac{1 + Y_r^2 + Y_i^2 + \sqrt{(1 + Y_r^2 - Y_i^2)^2 - 4Y_r^2}}{2}
\end{aligned} \tag{9}$$

Y_r and Y_i are the real and imaginary parts of Equation (8), determined experimentally. The choice of ϵ and n can be most

easily accomplished by comparison with the theoretical value of K_r ($K_r = \frac{\omega L}{c}$) which can be closely approximated if the speed of sound can be computed. Then if there is an integer n_K such that

$$(2n_K - 2)\pi < K_{r_{\text{theoretical}}} \Delta X \leq (2n_K - 1)\pi$$

it follows that $\epsilon=1$, $n=2(n_K-1)$. If there is an integer n_K such that

$$(2n_K - 1)\pi < K_{r_{\text{theoretical}}} \Delta X \leq 2n_K \pi$$

it follows that $\epsilon=-1$, $n=2n_K$. This problem in the choice of ϵ and n arises because with fixed transducers there is difficulty in knowing how many wavelengths exist between transducers. A and B may now be computed from Equation (6).

The computation of the impedance follows from its definition and the relation between the pressure and velocity⁷ through $\partial p / \partial x = -\rho \partial u / \partial t$. The result is

$$Z = \gamma \frac{K_r}{K} \left[\frac{Ae^{iKX} + Be^{-iKX}}{-Ae^{iKX} + Be^{-iKX}} \right] \quad (10)$$

A computer program has been developed to handle the solution of Equation (6) - (10). For a listing of the program and an explanation of its use see Appendix III.

Some comments concerning accuracy of the results are

in order here. If Y_i is near zero, $zz \approx 1$ from Equations (9). Consequently K_i is known with bad precision, and it is not possible to check that K_i is small. For this reason the calculation for zz is carried out in double precision on the computer. Checking the calculation with known pressure profiles K can be determined with an accuracy of 5×10^{-6} . Another problem occurs in the computation of A and B . Equations (6) use Equations (5) with $j = 1$ and 2 . But clearly $j = 1, 3$ or $j = 2, 3$ could also be used. Because of experimental errors the results will be different. The computer program calculates three values of A and B from the three possibilities and takes an arithmetic average. This average is the expression of the least mean square method in the A and B planes, respectively.

Another problem occurs if Z is large (rigid, closed end tube). Rewriting Equation (10)

$$Z = \gamma \frac{K_r}{K} \left[\frac{A/B e^{2iKX} + 1}{A/B e^{2iKX} - 1} \right]$$

it follows that Z is large if $A/B \approx e^{-2iKX}$. As an example consider

$$\begin{aligned} \gamma &= 1.405 & K &= 3 + 0.01 i \\ X &= 0.9 & B &= 1 \end{aligned}$$

Z is then large if $A \approx 0.646 + 0.786 i$. Computations for Z ,

varying the real and imaginary parts of A , are shown in Figure 6. It is clear that near $A = 0.6462 + 0.7868i$ a 10^{-4} error in A can make a drastic error in the computed impedance. Pressure profiles have been introduced into the main computer program of Appendix III which simulate $\text{Re}(Z) = 10, 100, 1000$ and 10000 . The accuracy with which this may be done is shown in the first two columns of the following table:

Re (Z)	Precision of program		10^{-7} Var. on P_1 gives	
	on A,B	on Z	on A,B	on Z
10	5×10^{-7}	5×10^{-6}	10^{-6}	5×10^{-5}
100	5×10^{-7}	10^{-4}	2×10^{-6}	10^{-2}
1000	5×10^{-7}	5×10^{-2}	2×10^{-6}	10^{-1}
10000	5×10^{-7}	1.	2×10^{-6}	1 to 100

Then a 10^{-7} error in $|P_1|$ is introduced at point X_1 and the resulting variation Z is noted in the last two columns of the table. The error is primarily in $I_m(Z)$ since the pressure profiles were selected so that $I_m(Z) \approx 0$. Nevertheless, if $|Z| > 100$, say, there can be expected a rather poor precision in its determination, particularly when P cannot be experimentally determined within 1%.

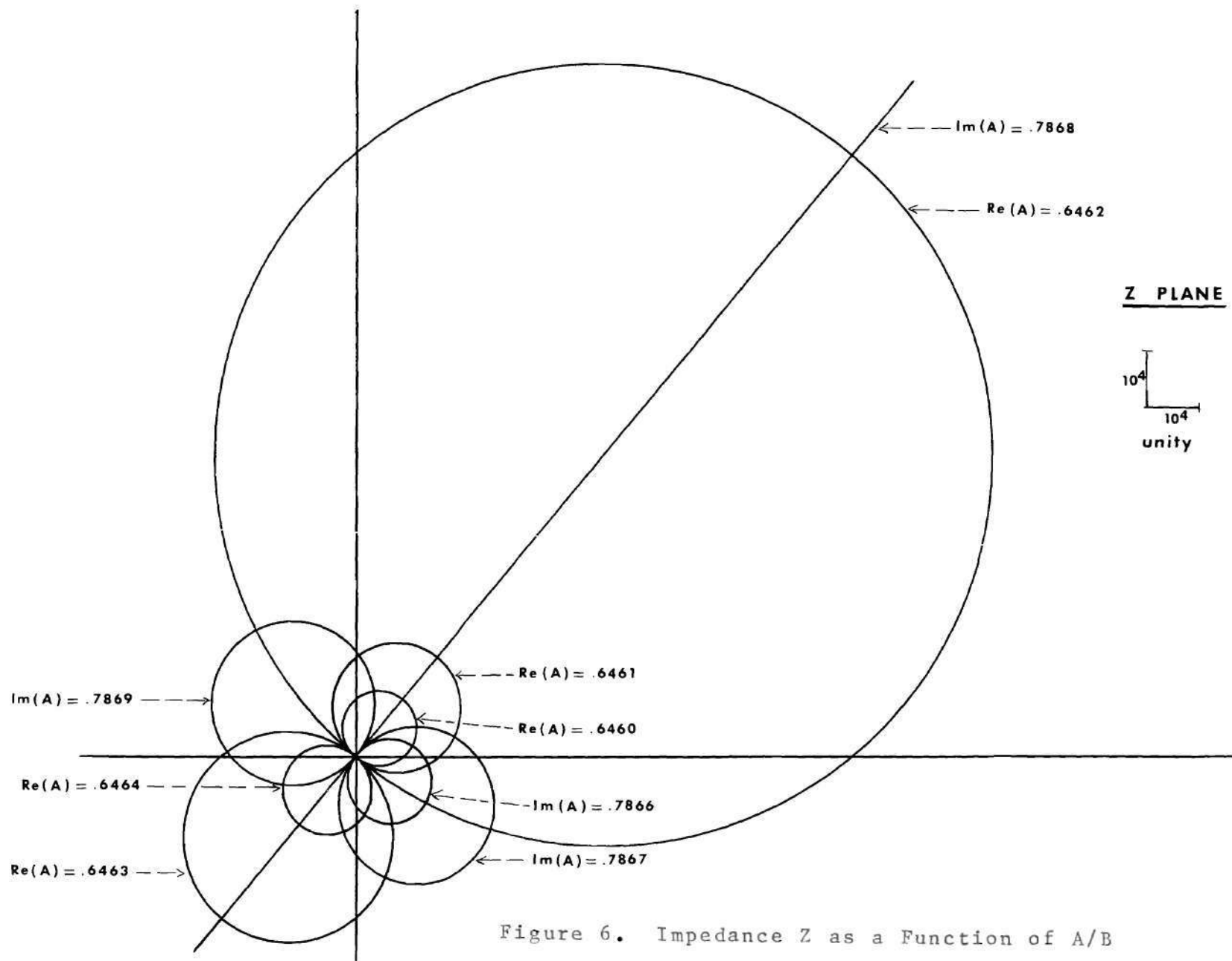


Figure 6. Impedance Z as a Function of A/B

CHAPTER V

EXPERIMENTAL RESULTS

The calibrations of Appendix II reduce to straight lines on a plot of voltage output vs absolute pressure as illustrated for channel 3 in Figure 7. The best straight line fits with pressure linear in voltage output for all channels are shown on the middle line of each box on Table 1. There is a slight variation with frequency. Generally the calibration can be performed within an error of 0.1 db. This corresponds to an error in absolute pressure of approximately 1%.

The pressures obtained from all five channels as a function of frequency are shown in Table 1. They are obtained by putting the voltage, data of the experiment (on the top line of each box, on Table 1) into the calibration formula (on the middle line of each box). These pressures are on the lower line of each box. Table 2 shows the results of the phase measurement along with a repeat of the pressure data of Table 1. The data of Table 2 are used in the computer program of Appendix III, and the computational results are shown in the output of Table 3.

As expected K_i is small compared with K_r , showing small acoustic losses. The three values of A and B are shown in Table 3 for each run along with the average value. The

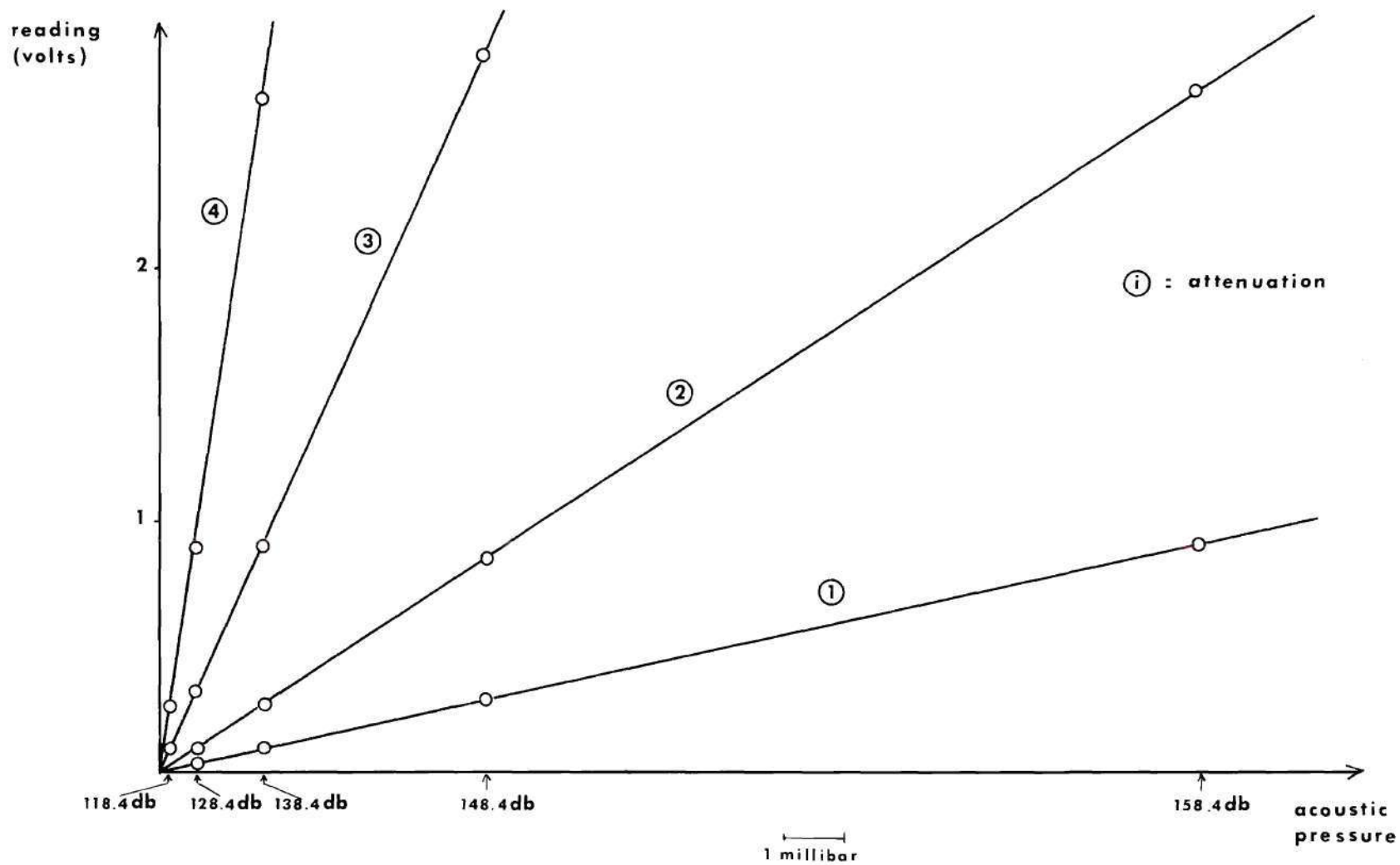


Figure 7. Calibration Curves of the Third Channel at 186 cps.

Table 1. Transformation from the Amplitude Reading (volts) to the Pressure Level (μ bar)

Reading (volts)
and attenuation
Transformation
formula
Pressure (μ bar)

frequency (cps)	channel				
	1	2	3	4	5
186	1.154 3 P=1625V+10 1886	.723 3 P=1313V+9 958	1.280 4 P=623.0V 797	1.082 3 P=1647V+15 1796	.893 3 P=1948V+13 1752
312	1.225 3 P=1627V+4 1996	1.026 3 P=1319V+3 1356	1.039 3 P=1850V+6 2427	1.140 4 P=555V+1.23 633.9	1.310 3 P=1949V+9 2561
443	.992 3 P=1623V+2 1611	.535 2 P=4356V+14 2345	1.135 4 P=620V+2 705.7	1.064 3 P=1657V+4 1766	1.193 3 P=1951V+2 2329
585	1.070 3 P=1617V+4 1733	.704 2 P=4394V 3093	.651 2 P=6110V+13 3991	.723 2 P=5575V 4030	1.688 3 P=1952V+2 3297
882	.607 4 P=545V+4 335	1.261 3 P=1315V+1 1659	1.046 3 P=1836V+8 1928	1.400 4 P=552V+4 776	1.286 4 P=667V-4 853
1025	.643 3 P=1618V 1040	1.274 3 P=1320V 1681	1.179 4 P=627V 740	1.076 3 P=1645V+6 1775	.584 4 P=673V 393

Table 2. Results of the Experiment

phase
(degrees)

amplitude
(μ bar)

frequency (cps)	channel and abscissa				
	$x_1 = .0750167$	$x_2 = .2893503$	$x_3 = .5036838$	$x_4 = .7180174$	$x_5 = .9323509$
186	0 1886	-16.6 958	-151.7 797	-169.5 1796	175.3 1752
312	0 1996	-156.5 1356	-175.8 2427	40.8 633.9	4.7 2561
443	0 1611	-166.5 2345	16.3 705.7	5. 1766	-173.7 2329
585	0 1733	-164.2 3093	9 3991	-162 4030	9.6 3297
882	0 335	-160. 1659	-12. 1928	162.5 776	162.3 853
1025	0 1040	-172. 1681	170.8 740	7. 1775	-5.5 393

Table 3. Results of the Program

FREQUENCY = 186,000

TEST NO 1

X =	.0750167	MAG(P) =	1886.0	PHASE(P) =	.0
X =	.2893503	MAG(P) =	958.0	PHASE(P) =	-16.6
X =	.5036838	MAG(P) =	797.0	PHASE(P) =	-151.7
Y =	.64866903		-.01240699		
AA=	.142092544053314024+001				
UU=	.265704647543574905-003				
ZZ=	.100013284350007403+001				
KR THEOR.=	4.02108127	KR EXPER.=	4.03611571		
		KI EXPER.=	.07604840		
A =	810.9536	-39.6650	B =	1108.4565	135.0231
A =	810.9533	-39.6649	B =	1108.4562	135.0230
A =	810.9532	-39.6652	B =	1108.4561	135.0233
A =	810.9534	-39.6650	B =	1108.4563	135.0231
MAG(Z) =	1.3452	PHASE(Z) =	27.1349		

TEST NO 2

X =	.2893503	MAG(P) =	958.0	PHASE(P) =	-16.6
X =	.5036838	MAG(P) =	797.0	PHASE(P) =	-151.7
X =	.7180174	MAG(P) =	1796.0	PHASE(P) =	-169.5
Y =	.64707354		.07979716		
AA=	.142507174830432432+001				
UU=	.108699323533572479-001				
ZZ=	.100542027647812894+001				
KR THEOR.=	4.02108127	KR EXPER.=	4.06666738		
		KI EXPER.=	.48555640		
A =	948.1633	38.9556	B =	831.5995	136.3410
A =	817.3788	-83.2029	B =	916.5183	163.3556
A =	933.3364	-170.8136	B =	931.3468	54.6304
A =	899.6262	-71.6870	B =	893.1549	118.1090
MAG(Z) =	1.3333	PHASE(Z) =	55.3307		

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	797.0	PHASE(P) =	-151.7
X =	.7180174	MAG(P) =	1796.0	PHASE(P) =	-169.5
X =	.9323509	MAG(P) =	1752.0	PHASE(P) =	175.3
Y =	.68194750		-.06005530		
AA=	.146865901610550976+001				
UU=	.670323842863777435-002				
ZZ=	.100334602128509872+001				
KR THEOR.=	4.02108127	KR EXPER.=	3.84204420		
		KI EXPER.=	.38156459		
A =	916.1474		119.8140	B =	908.6818
					46.3458
A =	916.1474		119.8137	B =	908.6817
					46.3458
A =	916.1476		119.8138	B =	908.6817
					46.3460
A =	916.1475		119.8139	B =	908.6817
					46.3459
MAG(Z) =	1.4999	PHASE(Z) =	45.1198		

TEST NO 4

X =	.0750167	MAG(P) =	1886.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	797.0	PHASE(P) =	-151.7
X =	.9323509	MAG(P) =	1752.0	PHASE(P) =	175.3
Y =	-.11996873		-.03769016		
AA=	.101581304367559963+001				
UU=	.144126186303697510-002				
ZZ=	.100072037146399542+001				
KR THEOR.=	4.02108127	KR EXPER.=	3.94471076		
		KI EXPER.=	.08854153		
A =	828.3503		-1.4541	B =	1112.8911
					91.5156
A =	828.3503		-1.4541	B =	1112.8911
					91.5156
A =	828.3503		-1.4541	B =	1112.8911
					91.5156
A =	828.3503		-1.4541	B =	1112.8911
					91.5156
MAG(Z) =	1.4669	PHASE(Z) =	27.5963		

Table 3. Results of the Program (Continued)

FREQUENCY = 312.000

TEST NO 1

X = .0750167	MAG(P) = 1996.0	PHASE(P) = .0
X = .2893503	MAG(P) = 1356.0	PHASE(P) = -156.5
X = .5036838	MAG(P) = 2427.0	PHASE(P) = -175.8
Y = .16967312	-.00230605	
AA= .102879428580346488+001		
UU= .547549659706244998-005		
ZZ= .100000273774455091+001		
KR THEOR.= 6.74503952	KR EXPER.= 6.53326792	
	KI EXPER.= .01091745	
A = 1019.5775	366.7428	B = 1496.8974 -110.5920
A = 1019.5780	366.7418	B = 1496.8967 -110.5912
A = 1019.5779	366.7428	B = 1496.8976 -110.5918
A = 1019.5778	366.7425	B = 1496.8972 -110.5917
MAG(Z) = 2.6909		PHASE(Z) = 24.1418

TEST NO 2

X = .2893503	MAG(P) = 1356.0	PHASE(P) = -156.5
X = .5036838	MAG(P) = 2427.0	PHASE(P) = -175.8
X = .7180174	MAG(P) = 633.9	PHASE(P) = 40.8
Y = .15881496	.01446867	
AA= .102543153320053560+001		
UU= .214757772285104208-003		
ZZ= .100010737312164892+001		
KR THEOR.= 6.74503952	KR EXPER.= 6.58470702	
	KI EXPER.= .06837051	
A = 1119.5097	357.3352	B = 1405.6723 -95.3780
A = 1085.3777	589.0588	B = 1368.2175 113.6123
A = 1046.2904	364.9835	B = 1472.3223 -78.6463
A = 1083.7259	437.1258	B = 1415.4040 -20.1373
MAG(Z) = 2.4726		PHASE(Z) = 22.2606

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	2427.0	PHASE(P) =	-175.8
X =	.7180174	MAG(P) =	633.9	PHASE(P) =	40.8
X =	.9323509	MAG(P) =	2561.0	PHASE(P) =	4.7
Y =	.09530137		-.04882113		
AA=	.101146585304982844+001				
UU=	.240529604124648573-002				
ZZ=	.100120192570791956+001				
KR THEOR.=	6.74503952	KR EXPER.=	6.88396508		
		KI EXPER.=	.22872838		
A =	1283.2256	100.0176	B = 1230.7254	159.2346	
A =	1283.2267	100.0173	B = 1230.7246	159.2342	
A =	1283.2258	100.0178	B = 1230.7255	159.2347	
A =	1283.2260	100.0176	B = 1230.7252	159.2345	
MAG(Z) =	2.0809		PHASE(Z) =	27.3007	

TEST NO 4

X =	.0750167	MAG(P) =	1996.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	2427.0	PHASE(P) =	-175.8
X =	.9323509	MAG(P) =	2561.0	PHASE(P) =	4.7
Y =	-.93768893		.02551255		
AA=	.187991140168023346+001				
UU=	.519532769091319852-002				
ZZ=	.100259429865270688+001				
KR THEOR.=	6.74503952	KR EXPER.=	6.48474425		
		KI EXPER.=	.16800066		
A =	1319.5823	365.8434	B = 1183.0588	-135.6409	
A =	1129.1228	268.8670	B = 1339.3036	-136.1564	
A =	949.9395	315.1233	B = 1474.9331	-17.1297	
A =	1132.8815	316.6113	B = 1332.4318	-96.3090	
MAG(Z) =	3.1089		PHASE(Z) =	36.7003	

Table 3. Results of the Program (Continued)

FREQUENCY = 443.000

TEST NO 1

X =	.0750167	MAG(P) =	1611.0	PHASE(P) =	.0
X =	.2893503	MAG(P) =	2345.0	PHASE(P) =	-166.5
X =	.5036838	MAG(P) =	705.7	PHASE(P) =	16.3
Y =	-.48429536		.07283767		
AA=	.123984731379972203+001				
UU=	.691636131374878890-002				
ZZ=	.100345222173940538+001				
KR THEOR.=	9.57709134			KR EXPER.=	9.67860639
				KI EXPER.=	.38756917
A =	1329.5989	543.1717		B =	986.2412 112.2863
A =	1314.8620	61.5104		B =	1080.9186 -199.7945
A =	976.6632	289.9548		B =	1332.7239 91.5341
A =	1207.0414	298.2123		B =	1133.2946 1.3419
MAG(Z) =		2.7638		PHASE(Z) =	49.5566

TEST NO 2

X =	.2893503	MAG(P) =	2345.0	PHASE(P) =	-166.5
X =	.5036838	MAG(P) =	705.7	PHASE(P) =	16.3
X =	.7180174	MAG(P) =	1766.0	PHASE(P) =	5.0
Y =	-.43250316		-.16401427		
AA=	.121395966330526323+001				
UU=	.328501568327785612-001				
ZZ=	.101629235795256207+001				
KR THEOR.=	9.57709134			KR EXPER.=	9.37971687
				KI EXPER.=	.84106395
A =	1501.3415	928.2124		B =	768.6944 -28.6615
A =	1501.3412	928.2108		B =	768.6940 -28.6611
A =	1501.3405	928.2117		B =	768.6940 -28.6618
A =	1501.3410	928.2116		B =	768.6941 -28.6615
MAG(Z) =		2.9535		PHASE(Z) =	68.5078

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	705.7	PHASE(P) =	16.3
X =	.7180174	MAG(P) =	1766.0	PHASE(P) =	5.0
X =	.9323509	MAG(P) =	2329.0	PHASE(P) =	-173.7
Y =	-.46330145		.02418989		
AA=	.121523337772327977+001				
UU=	.744929376111575773-003				
ZZ=	.100037239534890784+001				
KR THEOR.=	9.57709134	KR EXPER.=	9.57534742		
		KI EXPER.=	.12732491		
A =	1280.9306	358.0667	B = 1127.8069	-87.9741	
A =	1181.0310	297.2900	B = 1210.9342	-127.9072	
A =	1187.8026	407.8063	B = 1195.4251	-31.7550	
A =	1216.5880	354.3876	B = 1178.0554	-82.5454	
MAG(Z) =		3.9967	PHASE(Z) =	17.6848	

TEST NO 4

X =	.0750167	MAG(P) =	1611.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	705.7	PHASE(P) =	16.3
X =	.9323509	MAG(P) =	2329.0	PHASE(P) =	-173.7
Y =	-.52952471		-.03381687		
AA=	.128153998535108832+001				
UU=	.158820018810373481-002				
ZZ=	.100079378504670168+001				
KR THEOR.=	9.57709134	KR EXPER.=	9.69246888		
		KI EXPER.=	.09294319		
A =	1158.1202	240.1141	B = 1217.1843	-48.1890	
A =	1114.3890	202.4153	B = 1265.6331	-51.3202	
A =	1163.4198	181.4747	B = 1239.4977	-96.9411	
A =	1145.3097	208.0014	B = 1240.7717	-65.4834	
MAG(Z) =		3.3272	PHASE(Z) =	20.6201	

Table 3. Results of the Program (Continued)

FREQUENCY = 585.000

TEST NO 1

X =	.0750167	MAG(P) =	1733.0	PHASE(P) =	.0
X =	.2893503	MAG(P) =	3093.0	PHASE(P) =	-164.2
X =	.5036838	MAG(P) =	3991.0	PHASE(P) =	9.0
Y =	-.91019212		.15266986		
AA=	.185175776116342661+001				
UU=	.955905010171439152-001				
ZZ=	.104670459109394562+001				
KR THEOR.=	12.64694917	KR EXPER.=	12.24793983		
		KI EXPER.=	1.42045841		
A =	2947.8187	1522.4212	B =	1180.8026	-111.2670
A =	1577.9561	360.7932	B =	1562.6576	85.1742
A =	653.5620	676.9771	B =	1612.1762	873.0749
A =	1726.4456	853.3972	B =	1451.8788	282.3274
MAG(Z) =	1.6084		PHASE(Z) =	99.7786	

TEST NO 2

X =	.2893503	MAG(P) =	3093.0	PHASE(P) =	-164.2
X =	.5036838	MAG(P) =	3991.0	PHASE(P) =	9.0
X =	.7180174	MAG(P) =	4030.0	PHASE(P) =	-162.0
Y =	-.88344094		-.12486334		
AA=	.179605873171887659+001				
UU=	.592399952979458288-001				
ZZ=	.102919385700554287+001				
KR THEOR.=	12.64694917	KR EXPER.=	12.14417064		
		KI EXPER.=	1.12465604		
A =	2722.5905	1524.7995	B =	1275.6751	-198.7907
A =	2722.5896	1524.8016	B =	1275.6747	-198.7909
A =	2722.5907	1524.8028	B =	1275.6749	-198.7916
A =	2722.5903	1524.8013	B =	1275.6749	-198.7911
MAG(Z) =	2.3331		PHASE(Z) =	100.8448	

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	3991.0	PHASE(P) =	9.0
X =	.7180174	MAG(P) =	4030.0	PHASE(P) =	-162.0
X =	.9323509	MAG(P) =	3277.0	PHASE(P) =	9.6
Y =	-.89127902		.13685470		
AA=	.181310748141067624+001				
UU=	.722686112343977399-001				
ZZ=	.103550403728541674+001				
KR THEOR.=	12.64694917			KR EXPER.=	12.16551471
				KI EXPER.=	1.23961537
A =	3699.5894	6179.8989		B =	397.8446 -192.7381
A =	4463.8852	2892.9008		B =	664.4281 -394.7351
A =	2987.1077	1568.5667		B =	1184.0461 -162.8101
A =	3716.8607	3547.1221		B =	748.7729 -250.0944
MAG(Z) =		4.2816		PHASE(Z) =	72.3989

TEST NO 4

X =	.0750167	MAG(P) =	1773.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	3991.0	PHASE(P) =	9.0
X =	.9323509	MAG(P) =	3277.0	PHASE(P) =	9.6
Y =	.62991627		-.03044879		
AA=	.139772162936246996+001				
UU=	.153545445011422708-002				
ZZ=	.100076743274854534+001				
KR THEOR.=	12.64694917			KR EXPER.=	12.58135116
				KI EXPER.=	.09138759
A =	2067.4614	731.8848		B =	1924.0204 -68.3382
A =	1916.4865	508.4149		B =	2098.6096 -214.0945
A =	1743.3446	687.8730		B =	2213.6054 3.3387
A =	1909.0975	642.7242		B =	2078.7451 -93.0313
MAG(Z) =		6.1354		PHASE(Z) =	29.5652

Table 3. Results of the Program (Continued)

FREQUENCY = 882.000

TEST NO 1

X = .0750167	MAG(P) = 335.0	PHASE(P) = .0
X = .2893503	MAG(P) = 1629.0	PHASE(P) = -160.0
X = .5036838	MAG(P) = 1928.0	PHASE(P) = -12.0
Y = -.58765313	.34245395	
AA= .146261089712096060+001		
UU= .166588658008988783+000		
ZZ= .108008733813936953+001		
KR THEOR.= 19.06770778	KR EXPER.= 19.30213618	
	KI EXPER.= 1.85502538	
A = -858.7740	-509.9441	B = -688.7990 72.9119
A = -858.7790	-509.9390	B = -688.7979 72.9117
A = -858.7802	-509.9420	B = -688.7994 72.9097
A = -858.7777	-509.9417	B = -688.7988 72.9111
MAG(Z) = 1.4010		PHASE(Z) = 91.4480

TEST NO 2

X = .2893503	MAG(P) = 1659.0	PHASE(P) = -160.0
X = .5036838	MAG(P) = 1928.0	PHASE(P) = -12.0
X = .718074	MAG(P) = 776.0	PHASE(P) = 162.5
Y = -.56518120	-.20870333	
AA= .136298686578819983+001		
UU= .622867031394315490-001		
ZZ= .103067293703649343+001		
KR THEOR.= 19.06770778	KR EXPER.= 19.27829361	
	KI EXPER.= 1.15265334	
A = -1980.1441	767.8135	B = -424.7850 51.8790
A = -1493.1708	-918.0146	B = -526.2512 -266.8620
A = -670.5241	-236.5195	B = -939.0921 93.9006
A = -1381.2797	-130.2402	B = -630.0428 -40.3608
MAG(Z) = 1.8381		PHASE(Z) = 73.7715

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	1928.0	PHASE(P) =	-12.0
X =	.7180174	MAG(P) =	776.0	PHASE(P) =	162.5
X =	.9323509	MAG(P) =	853.0	PHASE(P) =	162.3
Y =	-.68693875		-.12098547		
AA=	.148652232605982082+001				
UU=	.270785583507004946-001				
ZZ=	.101344884347987713+001				
KR THEOR.=	19.06770778	KR EXPER.=	18.51128483		
		KI EXPER.=	.76433134		
A =	-1859.8222	70.9163	B = -311.1569	430.5788	
A =	-1953.4611	-281.1778	B = -336.9973	346.8733	
A =	-1695.0984	-327.8542	B = -449.8636	392.0349	
A =	-1836.1272	-179.3719	B = -366.0059	389.8290	
MAG(Z) =	7.5012		PHASE(Z) =	53.6996	

TEST NO 4

X =	.0750167	MAG(P) =	335.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	1928.0	PHASE(P) =	-12.0
X =	.9323509	MAG(P) =	853.0	PHASE(P) =	162.3
Y =	-.13514079		.04003391		
AA=	.101986574626703169+001				
UU=	.163247911426662764-002				
ZZ=	.100081590670525747+001				
KR THEOR.=	19.06770778	KR EXPER.=	18.63783026		
		KI EXPER.=	.09422921		
A =	-1131.9288	-119.2405	B = -748.3518	480.5179	
A =	-796.3926	381.7147	B = -554.3341	947.6118	
A =	-970.9573	-146.3284	B = -892.5096	515.9677	
A =	-966.4262	38.7153	B = -731.7318	648.0324	
MAG(Z) =	8.5351		PHASE(Z) =	38.0754	

Table 3. Results of the Program (Continued)

FREQUENCY = 1025.000

TEST NO 1

X = .0750167	MAG(P) = 1040.0	PHASE(P) = .0
X = .2893503	MAG(P) = 1681.0	PHASE(P) = -172.0
X = .5036838	MAG(P) = 740.0	PHASE(P) = 170.8
Y = -.09606574	-.02203575	
AA= .100971420024036330+001		
UU= .490094814641664172-003		
ZZ= .100024501739055999+001		
KR THEOR.= 22.15918422	KR EXPER.= 21.53744078	
	KI EXPER.= .10327955	
A = -895.9608	-630.5575	B = -724.7538 360.4879
A = -701.6606	-977.4825	B = -648.1250 10.4376
A = -820.6571	-616.1635	B = -796.7046 354.2685
A = -806.0928	-741.7012	B = -723.1945 241.7313
MAG(Z) = 12.6428		PHASE(Z) = -44.3621

TEST NO 2

X = .2893503	MAG(P) = 1681.0	PHASE(P) = -172.0
X = .5036838	MAG(P) = 740.0	PHASE(P) = 170.8
X = .7180174	MAG(P) = 1775.0	PHASE(P) = 7.0
Y = -.06668833	.00126829	
AA= .100444894122852291+001		
UU= .161575282775263374-005		
ZZ= .100000080787608754+001		
KR THEOR.= 22.15918422	KR EXPER.= 21.67486548	
	KI EXPER.= .00593058	
A = -880.1067	-549.9387	B = -774.8504 313.3073
A = -880.1078	-549.9365	B = -774.8517 313.3049
A = -880.1071	-549.9387	B = -774.8508 313.3073
A = -880.1072	-549.9379	B = -774.8510 313.3065
MAG(Z) = 7.5286		PHASE(Z) = -34.0054

Table 3. Results of the Program (Continued)

TEST NO 3

X =	.5036838	MAG(P) =	740.0	PHASE(P) =	170.8
X =	.7180174	MAG(P) =	1775.0	PHASE(P) =	7.0
X =	.9323509	MAG(P) =	393.0	PHASE(P) =	-5.5
Y =	-.09209376		.03419524		
AA=	.100965057576871305+001				
UU=	.117930491348411795-002				
ZZ=	.100058947871416483+001				
KR THEOR.=	22.15918422			KR EXPER.=	21.55619884
				KI EXPER.=	.16019076
A =	-906.1895	-661.2070		B =	-694.4929 321.3182
A =	-906.1711	-661.1882		B =	-694.4736 321.3212
A =	-906.1912	-661.2023		B =	-694.4933 321.3143
A =	-906.1839	-661.1991		B =	-694.4866 321.3179
MAG(Z) =		13.7690		PHASE(Z) =	-17.4620

TEST NO 4

X =	.0750167	MAG(P) =	1040.0	PHASE(P) =	.0
X =	.5036838	MAG(P) =	740.0	PHASE(P) =	170.8
X =	.9323509	MAG(P) =	393.0	PHASE(P) =	-5.5
Y =	-.95865029		-.12948553		
AA=	.193577687414791910+001				
UU=	.101296304528206702+000				
ZZ=	.104942665514470648+001				
KR THEOR.=	22.15918422			KR EXPER.=	22.96366668
				KI EXPER.=	.73047043
A =	-415.4815	-869.1096		B =	91.5219 115.4497
A =	-36.9221	-885.4733		B =	56.8184 206.0798
A =	104.9610	-740.9022		B =	139.4041 367.7427
A =	-115.8142	-831.8284		B =	95.9148 229.7574
MAG(Z) =		2.6951		PHASE(Z) =	164.7745

dispersion generally increases with frequency. Another problem concerning accuracy was uncovered during the data reduction. In Equation (8), if P_1' and P_3 are nearly 180° out of phase and of comparable magnitude, the experimental precision on Y is poor as shown in Figure 8.

The phase and magnitude of Z are shown on Figures 9 and 10. Clearly, the precision of the measurement decreases with a frequency increase, as more wavelengths occur between transducers. It is not immediately obvious why this should be so. However some findings are the following:

1. As frequency increases more pairs of transducers 1 and 3 are 180° out of phase yielding the accuracy problem mentioned above. However, rejecting those measurements does not appear to reduce the data scatter.
2. Nonlinearities at acoustic resonance may alter the waveform and it was specifically assumed that only the fundamental component of the wave was present. However, an independent spectral analysis of the tape yielded only very small higher harmonic content.
3. The first and fifth transducers were rather near the acoustic source and end plate, respectively. Since these may not be located in a region of a sufficiently one dimensional waveform, errors may occur. A check on this assumption could not be made

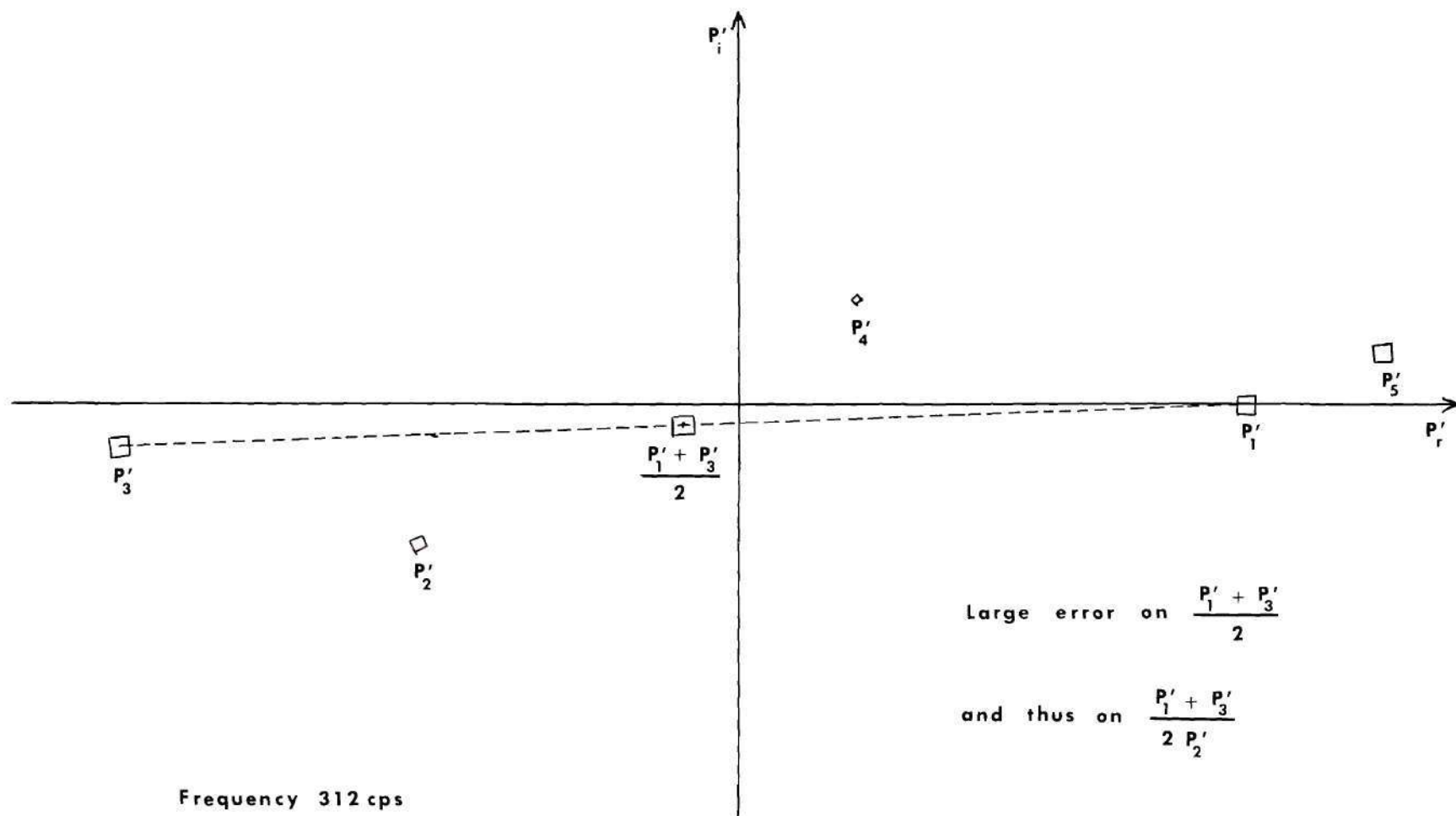


Figure 8. Pressure on Complex Plane

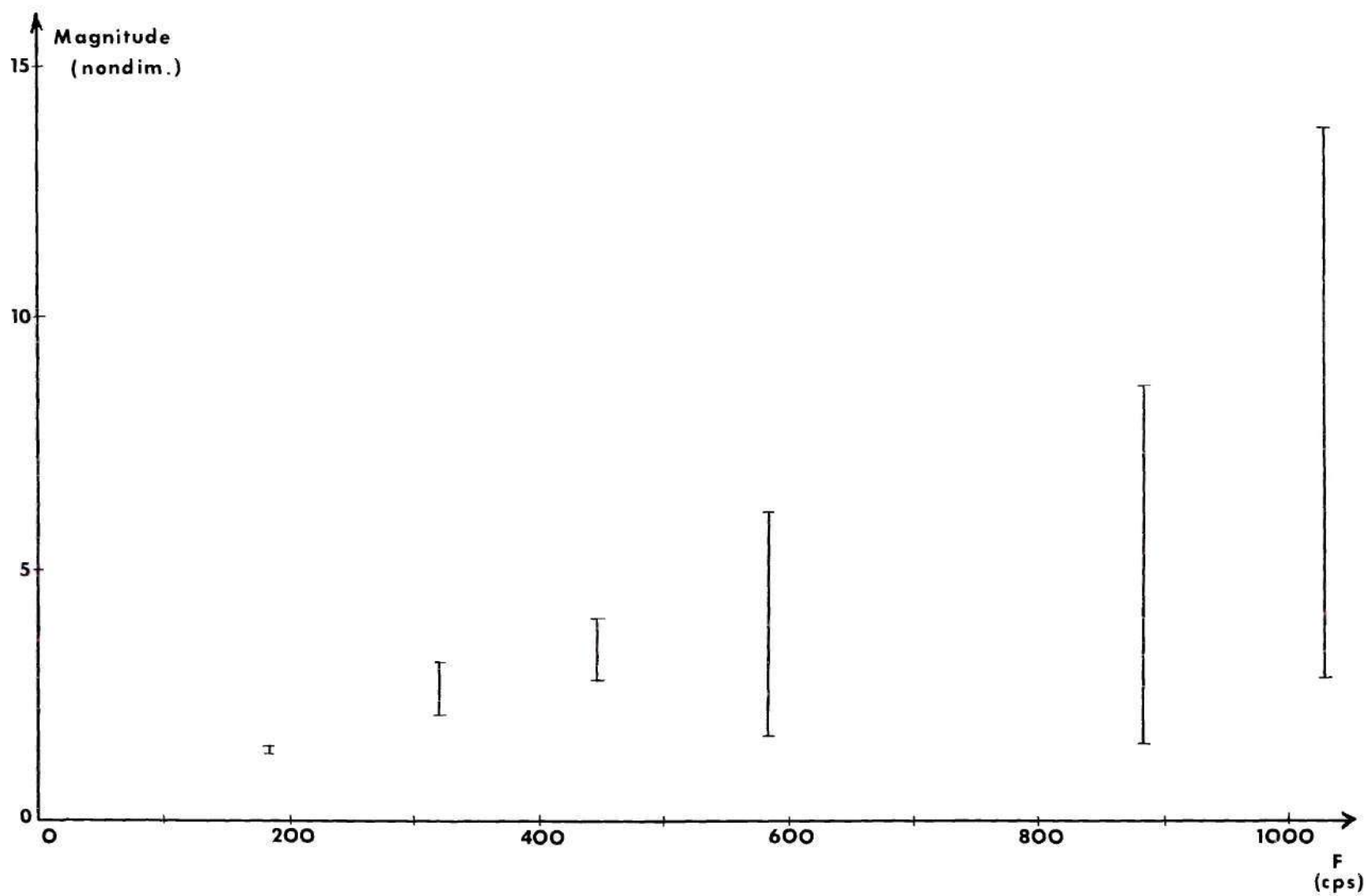


Figure 9. Magnitude of the Impedance vs. the Frequency

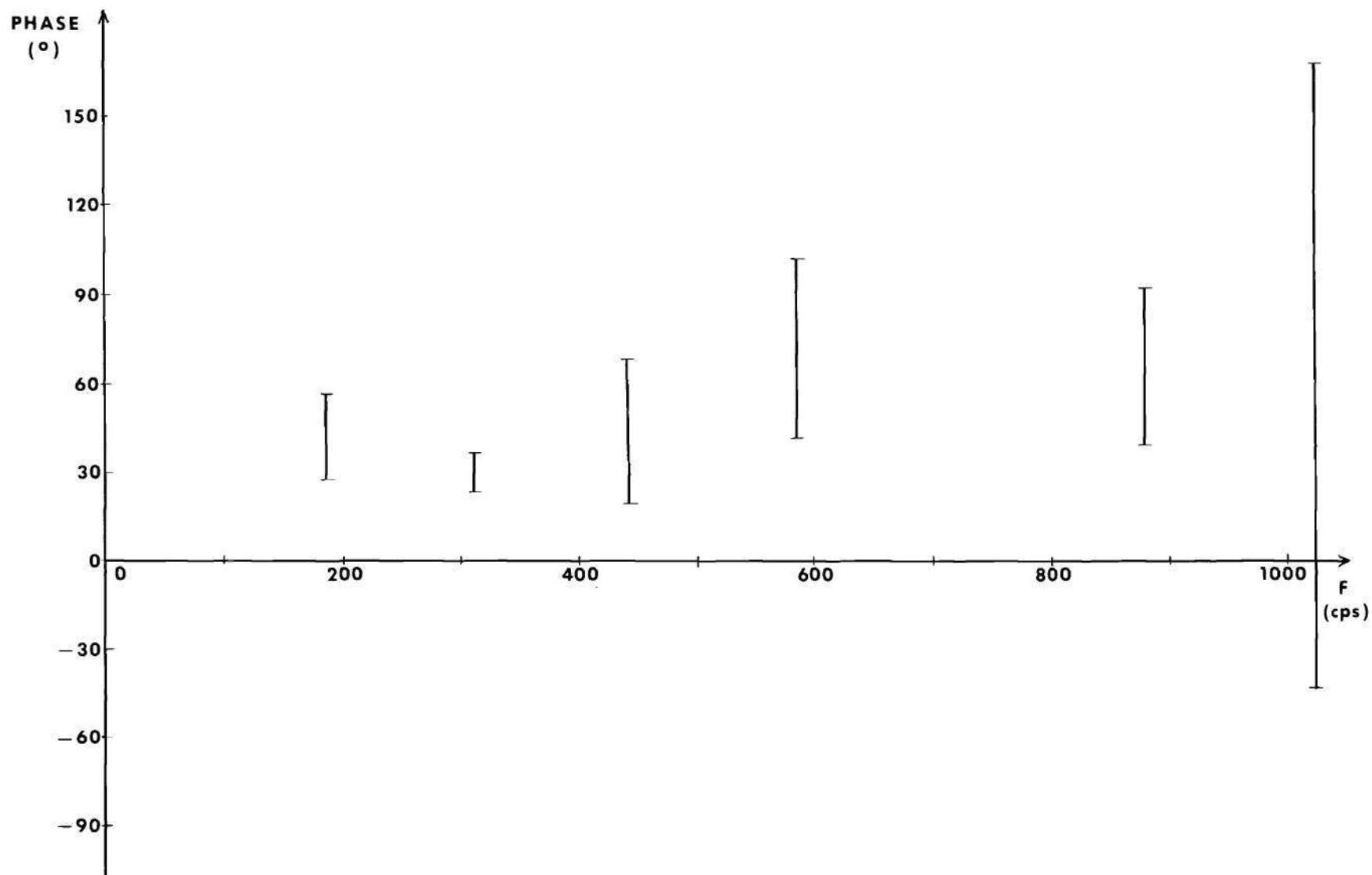


Figure 10. Phase of the Impedance vs. the Frequency

because rejection of data from these two transducers yield only one measurement with the middle three transducers.

It should finally be noted that the precision obtained in the impedance measurement is insufficient to extract the vaporization response with sufficient accuracy. Therefore, changes in the experiment should be made to improve the accuracy.

CHAPTER VI

SUMMARY AND CONCLUSIONS

1. Measurement of the impedance of a vaporizing spray was shown to be not feasible due to an insufficient feedback to the acoustic wave in an impedance tube using a practical experimental setup for spray injection.
2. An experiment was designed to measure the impedance of a large vaporizing surface upon which a mean flow was impinging.
3. Measurement of the impedance of the vaporization surface assembly without air or liquid flow was accomplished using an impedance tube with fixed pressure transducers.
4. The precision of the impedance measurement was poor, especially at the higher frequencies. It is recommended that
 - a) The transducers be moved closer together so that no more than one wavelength is under surveillance by the transducers.
 - b) The two outside transducers be moved farther from the driver and endplate.
 - c) The tape recordings be filtered to remove components of the wave at frequencies higher than the driving frequency.

These changes should improve the precision of the measurements.

APPENDIX I

CALIBRATION OF THE MICROPHONES

	1	2	3	4	5
Serial No.	176067	176152	176154	176330	276991
Given correction factor K	33.0 db	37.7 db	37.5 db	33.7 db	36.5 db
120 db (calibrator)	121.6 db	124.6 db	121.7 db	122.4 db	121.0 db
130 db (calibrator)	131.6	134.5	131.6	132.5	130.8
140 db (calibrator)	141.6	144.6	141.6	142.4	141.0
150 db (calibrator)	151.6	154.6	151.7	152.5	150.9
Thus, mean correction to add, k	-1.6	-4.6	-1.6	-2.5	-0.9

Correction factor to	31.4 db	33.1 db	35.9db	31.2 db	35.6 db
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APPENDIX II

CALIBRATION TABLES OF THE FIVE CHANNELS

CALIBRATION OF CHANNEL No. 1

Voltage output (Volt) for a given
pressure input (dB).

P (μbar)	P (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	1.024	3.07						186
5260.	148.4	.321	.961	3.23					
1663.4	138.4	.104	.303	1.013	3.02				
526.0	128.4	.034	.096	.317	.946	3.16			
166.34	118.4	.012	.031	.102	.301	1.006	3.01		
52.60	108.4	.005	.010	.032	.095	.317	.948	3.17	
16634.	158.4	1.003	3.01						312
5260.	148.4	.321	.958	3.23					
1663.4	138.4	.102	.302	1.012	3.02				
526.0	128.4	.035	.097	.321	.956	3.20			
166.34	188.4	.013	.031	.101	.302	1.010	3.02		
52.60	108.4	.004	.010	.032	.096	.320	.956	3.21	

CALIBRATION OF CHANNEL No. 1 (Continued)

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	1.019	3.06						443
5260.	148.4	.322	.962	3.24					
1663.4	138.4	.104	.305	1.019	3.04				
526.0	128.4	.034	.097	.323	.959	3.22			
166.34	118.4	.012	.031	.102	.302	1.009	3.02		
52.60	108.4	.004	.010	.032	.096	.320	.956	3.20	
16634.	158.4	1.025	3.07						585
5260.	148.4	.323	.965	3.25					
1663.4	138.4	.105	.306	1.024	3.05				
526.0	128.4	.032	.096	.323	.960	3.23			
166.34	118.4	.012	.031	.102	.303	1.017	3.04		
52.60	108.4	.007	.010	.032	.096	.321	.960	3.21	

CALIBRATION OF CHANNEL No. 1 (Continued)

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	1.017	3.06						882
5260.	148.4	.324	.972	3.27					
1663.4	138.4	.103	.307	1.028	3.04				
526.0	128.4	.033	.096	.322	.950	3.22			
166.34	118.4	.011	.030	.101	.297	1.005	3.01		
52.60	108.4	.003	.009	.031	.094	.318	.952	3.19	
16634.	158.4	1.014	3.05						1025
5260.	148.4	.323	.968	3.25					
1663.4	138.4	.102	305	1.025	3.02				
526.0	128.4	.034	.097	.325	.953	3.24			
166.34	118.4	.010	.030	.101	.299	1.015	3.04		
52.60	108.4	.005	.011	.032	.095	.322	.964	3.23	

CALIBRATION OF CHANNEL No. 2

Voltage output (Volt) for a given
pressure input (dB).

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
11776.	155.4	.893	2.68						186
3724.	145.4	.282	.846	2.83					
1177.6	135.4	.088	2.63	.881	2.63				
372.4	125.4	.027	.082	.277	.830	2.77			
117.76	115.4	.008	.025	.087	.265	.885	2.65		
37.24	105.4	.003	.007	.027	.082	.280	.83	2.75	
11776.	155.4	.884	2.65						312
3724.	145.4	.281	.842	2.82					
1177.6	135.4	.089	.265	.887	2.65				
372.4	125.4	.029	.084	.280	.838	2.79			
117.76	115.4	.009	.026	.088	.265	.887	2.66		
37.24	105.4	.003	.008	.027	.083	.279	.84	2.78	

CALIBRATION OF CHANNEL No. 2 (Continued)

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
11776.	155.4	.900	2.70						443
3724.	145.4	.281	.842	2.82					
1177.6	135.4	.089	.267	.897	2.67				
372.4	125.4	.028	.084	.281	.839	2.80			
117.76	115.4	.009	.026	.088	.263	.880	2.64		
37.24	105.4	.003	.009	.028	.083	.279	.838	2.78	
11776.	155.4	.894	2.68						585
3724.	145.4	.284	.849	2.84					
1177.6	135.4	.090	.268	.896	2.66				
372.4	125.4	.027	.084	.283	.839	2.81			
117.76	115.4	.010	.026	.089	.264	.880	2.64		
37.24	105.4	.006	.008	.027	.082	.276	.830	2.75	

CALIBRATION OF CHANNEL No. 2 (Continued)

p	p	Attenuation							Frequency
(μ bar)	(dB)	1	2	3	4	5	6	7	(cps)
11776.	155.4	.893	2.68						
3724.	145.4	.282	.845	2.83					
1177.6	135.4	.089	.267	.894	2.64				
372.4	125.4	.028	.084	.282	.834	2.81			882
117.76	115.4	.011	.027	.090	.264	.894	2.68		
37.24	105.4	.007	.010	.028	.082	.278	.833	2.76	
11776.	155.4	.900	2.70						
3724.	145.4	.283	.851	2.85					
1177.6	135.4	.088	.264	.884	2.60				
372.4	125.4	.029	.085	.283	.834	2.82			1025
117.76	115.4	.011	.027	.090	.264	.894	2.68		
37.24	105.4	.006	.010	.028	.082	.277	.832	2.76	

CALIBRATION OF CHANNEL No. 3

Voltage output (Volt) for a given
Pressure input (dB).

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	.902	2.71						186
5260.	148.4	.283	.852	2.84					
1663.4	138.4	.088	.266	.890	2.67				
526.0	128.4	.030	.084	.279	.835	2.80			
166.34	118.4	.010	.027	.089	.267	.890	2.66		
52.60	108.4	.003	.008	.027	.084	.277	.840	2.83	
16634.	158.4	.888	2.68						312
5260.	148.4	.283	.849	2.84					
1663.4	138.4	.091	.268	.893	2.68				
526.0	128.4	.028	.084	.281	.841	2.82			
166.34	118.4	.009	.026	.089	.265	.890	2.67		
52.60	108.4	.002	.008	.027	.083	.280	.846	2.81	

CALIBRATION OF CHANNEL No. 3 (Continued)

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	.904	2.72						443
5260.	148.4	.286	.855	2.86					
1663.4	138.4	.089	.267	.894	2.68				
526.0	128.4	.030	.085	.283	.845	2.84			
166.34	118.4	.009	.026	.088	.265	.885	2.65		
52.60	108.4	.002	.008	.027	.083	.280	.846	2.82	
16634.	158.4	.899	2.72						585
5260.	148.4	.284	.851	2.86					
1663.4	138.4	.091	.270	.899	2.70				
526.0	128.4	.030	.086	.283	.844	2.84			
166.34	118.4	.009	.026	.089	.265	.893	2.67		
52.60	108.4	.005	.009	.028	.083	.283	.845	2.83	

CALIBRATION OF CHANNEL No. 3 (Continued)

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
16634.	158.4	.892	2.68						882
5260.	148.4	.285	.857	2.86					
1663.4	138.4	.089	.269	.899	2.66				
526.0	128.4	.027	.084	.282	.836	2.83			
166.34	118.4	.008	.026	.089	.264	.896	2.68		
52.60	108.4	.003	.008	.027	.082	.280	.843	2.83	
16634.	158.4	.900	2.72						1025
5260.	148.4	.288	.863	2.88					
1663.4	138.4	.091	.269	.897	2.65				
526.0	128.4	.030	.086	.283	.835	2.85			
166.34	118.4	.011	.027	.090	.264	.897	2.69		
52.60	108.4	.004	.009	.028	.083	.280	.845	2.82	

CALIBRATION OF CHANNEL No. 4

Voltage output (Volt) for a given
pressure input (dB).

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
14998.	157.5	.903	2.70						186
4743.	147.5	.284	.852	2.87					
1499.8	137.5	.088	.265	.894	2.68				
474.3	127.5	.027	.082	.279	.840	2.81			
149.98	117.5	.008	.026	.088	.267	.890	2.67		
47.43	107.5	.005	.009	.028	.084	.280	.841	2.81	
14998.	157.5	.893	2.68						312
4743.	147.5	.280	.847	2.86					
1499.8	137.5	.091	.268	.901	2.70				
474.3	127.5	.029	.085	.284	.851	2.85			
149.98	117.5	.011	.027	.090	.268	.899	2.69		
47.43	107.5	.002	.008	.028	.085	.286	.848	2.84	

CALIBRATION OF CHANNEL No. 4 (Continued)

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
14998.	157.5	.913	2.74						443
4743.	147.5	.284	.850	2.86					
1499.8	137.5	.089	.267	.904	2.71				
474.3	127.5	.028	.084	.284	.850	2.84			
149.98	117.5	.009	.026	.089	.267	.900	2.69		
47.43	107.5	.002	.008	.028	.084	.287	.853	2.86	
14998.	157.5	.895	2.69						585
4743.	147.5	.286	.855	2.90					
1499.8	137.5	.089	.269	.905	2.71				
474.3	127.5	.028	.084	.286	.853	2.87			
149.98	117.5	.008	.026	.089	.267	.900	2.69		
47.43	107.5	.005	.009	.028	.084	.283	.849	2.84	

CALIBRATION OF CHANNEL No. 4 (Continued)

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
14998.	157.5	.899	2.70						882
4743.	147.5	.284	.850	2.86					
1499.8	137.5	.089	.269	.904	2.71				
474.3	127.5	.028	.084	.285	.844	2.86			
149.98	117.5	.011	.027	.090	.265	.900	2.69		
47.43	107.5	.002	.008	.028	.083	.283	.848	2.83	
14998.	157.5	.900	2.70						1025
4743.	147.5	.286	.858	2.88					
1499.8	137.5	.091	.270	.904	2.67				
474.3	127.5	.028	.084	.285	.840	2.85			
149.98	117.5	.010	.027	.090	.264	.899	2.69		
47.43	107.5	.003	.008	.028	.083	.284	.849	2.83	

CALIBRATION OF CHANNEL No. 5

Voltage output (Volt) for a given
pressure input (dB).

p (μ bar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
18032.	159.1	.918	2.75						
5702.	149.1	.289	.865	2.92					
1803.2	139.1	.089	.269	.910	2.72				
570.2	129.1	.030	.085	.286	.856	2.85			186
180.32	119.1	.008	.026	.090	.272	.907	2.72		
57.02	109.1	.003	.008	.028	.085	.283	.850	2.84	
18032.	159.1	.899	2.70						
5702.	149.1	.290	.866	2.92					
1803.2	139.1	.092	.272	.915	2.73				
570.2	129.1	.030	.086	.288	.860	2.86			312
180.32	119.1	.011	.028	.092	.274	.913	2.73		
57.02	109.1	.005	.009	.028	.085	.285	.855	2.85	

CALIBRATION OF CHANNEL No. 5 (Continued)

P (μbar)	P (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
18032.	159.1	.924	2.77						443
5702.	149.1	.289	.868	2.92					
1803.2	139.1	.091	.273	.919	2.74				
570.2	129.1	.028	.086	.291	.868	2.90			
180.32	119.1	.011	.028	.091	.271	.905	2.71		
57.02	109.1	.005	.009	.029	.086	.286	.860	2.86	
18032.	159.1	.916	2.75						585
5702.	149.1	.289	.870	2.92					
1803.2	139.1	.091	.274	.925	2.75				
570.2	129.1	.028	.086	.291	.868	2.90			
180.32	119.1	.011	.027	.091	.271	.908	2.71		
57.02	109.1	.005	.009	.028	.085	.285	.855	2.85	

CALIBRATION OF CHANNEL No. 5 (Continued)

p (μbar)	p (dB)	Attenuation							Frequency (cps)
		1	2	3	4	5	6	7	
18032.	159.1	.903	2.72						882
5702.	149.1	.291	.875	2.95					
1803.2	139.1	.090	.273	.920	2.71				
570.2	129.1	.029	.087	.294	.867	2.92			
180.32	119.1	.009	.027	.093	.277	.931	2.79		
57.02	109.1	.002	.008	.028	.085	.290	.870	2.90	
18032.	159.1	.900	2.70						1025
5702.	149.1	.288	.864	2.91					
1803.2	139.1	.090	.272	.915	2.69				
570.2	129.1	.028	.084	.285	.839	2.83			
180.32	119.1	.009	.027	.092	.271	.922	2.75		
57.02	109.1	.002	.008	.028	.085	.291	.872	2.90	

APPENDIX III

LISTING AND USE OF THE COMPUTER PROGRAM

The data required consist of $2+2n$ cards where n is the number of frequencies for which experiments were run.

Card #1	F15.8	Length of the impedance tube (ft)
	I2	Number of frequencies, n
	F10.7	Dimensionless value of X at which the impedance measurement is computed (usually $X = 1.0$)
Card #2	5(F10.7)	Dimensionless X locations of the five transducers
Card #3	F10.3	Frequency (cps)
Card #4	5(F10.3,F6.1)	Five pairs of values of the pressure amplitude (any units) followed by the phase (degrees) for the five transducers

Repeat cards #3 and #4 n times

The program gives four results for each frequency. Test #1 is the result for channels 1, 2 and 3, Test #2 is for 2, 3 and 4, Test #3 is for 3, 4 and 5, and Test #4 is for channels 1, 3 and 5. The value $KR\ THEOR = \frac{\omega L}{c}$ where $c = 1130\ ft/sec$. $KR\ EXPER$ and $KI\ EXPER$ are the experimentally determined values of K . The first three pairs

of values for A and B are determined from p_j measurements with $j = 2, 3$, $j = 3, 1$, and $j = 1, 2$, respectively. The fourth pair is the average which is used in these impedance calculation.

The following is a program listing:

```

COMPLEX P(3)
DIMENSION X(5),AP(2,3),AX(5),AAP(2,5)
COMPLEX CI,A,B,K,Y,DEN,Z,A1,A2,A3,B1,B2,B3
REAL KR,KI
DOUBLE PRECISION AA,UU,ZZ
PI=3.141592
CI=CMPLX(0,1)
READ(5,100)RL,N,XX
100 FORMAT(F15.8,I2,F10.7)
READ(5,110)(AX(I),I=1,5)
110 FORMAT(5(F10.7))
C=1130.
GAM=1.405
DO 10 II=1,N
  READ(5,101)F
101 FORMAT(F10.3)
  WRITE(6,1110)F
1110 FORMAT(1H1,11HFREQUENCY =,F10.3/////////)
  READ(5,200)((AAP(I,J),I=1,2),J=1,5)
200 FORMAT(5(F10.3,F6.1))
  RKTH=2.*PI*F*RL/C
  DO 9 JJ=1,4
    DO 1 J=1,3
      DO 16 I=1,2
        IF(JJ-2)13,14,15
13  AP(I,J)=AAP(I,J)
    X(J)=AX(J)
    GO TO 16
14  AP(I,J)=AAP(I,J+1)
    X(J)=AX(J+1)
    GO TO 16
15  IF(JJ-4)17,18,18
17  AP(I,J)=AAP(I,J+2)
    X(J)=AX(J+2)
    GO TO 16
18  AP(I,J)=AAP(I,2*J-1)
    X(J)=AX(2*J-1)
16  CONTINUE
  1  P(J)=AP(1,J)*CEXP(CI*AP(2,J)*PI/180.)
  Y=(P(1)+P(3))/2./P(2)
  AA=1.+CABS(Y)*CABS(Y)
  YR=REAL(Y)

```

```

YI=AIMAG(Y)
UU=YI*YI*(.5+(1.+YR*YR+YI*YI*.5)/(1.-YR*YR+SQRT(AA*AA-4
                                *YR*YR)))
ZZ=SQRT(1.+UU)
KR=ACOS(REAL(Y)/ZZ)/(X(2)-X(1))
KI=ALOG(SQRT(1.+UU)+SQRT(UU))/(X(2)-X(1))
RKT=RKTH*(X(2)-X(1))/PI
DO 2 NK=1,100
  RNK1=RKT-2.*FLOAT(NK)+1.
  RNK2=RKT-2.*FLOAT(NK)
  IF(RNK1)3,3,4
3  KR=KR+2.*FLOAT(NK-1)*PI/(X(2)-X(1))
  GO TO 6
4  IF(RNK2)5,5,2
5  KR=-KR+2.*FLOAT(NK)*PI/(X(2)-X(1))
  GO TO 6
2  CONTINUE
6  CONTINUE
  WRITE(6,1000)JJ
1000 FORMAT(1H ,7HTEST NO,I2/)
  WRITE(6,1500)(X(J),(AP(I,J),I=1,2),J=1,3)
1500 FORMAT(1H ,10X,3HX =,F10.7,4X,8HMAG(P) =,F8.1,3X,10H
                                PHASE(P) =,F8.11)
  TEST=ABS(X(1)+X(3)-2.*X(2))-0.001
  IF(TEST)11,12,12
  12 WRITE(6,1200)
1200 FORMAT(1H ,37HX(3)-X(2) = X(2)-X(1) IS NOT VERIFIED)
  11 CONTINUE
  WRITE(6,1111)Y,AA,UU,ZZ
1111 FORMAT(1H ,10X,3HY =,2(F15.8)/1H ,10X,3HAA=,D25.18/1H
                                ,10X,3HUU=,D125.18/1H ,10X,3HZZ
                                =,D25.18)
  WRITE(6,2000)RKTH,KR
2000 FORMAT(1H ,10X,10HKR THEOR.=,F15.8,10X,10HKR EXPER.=
                                ,F15.8)
  WRITE(6,3000)KI
3000 FORMAT(1H ,45X,10HKI EXPER.=,F15.8)
  K=CMPLX(KR,KI)
  DEN=2.*CI*CSIN(K*(X(2)-X(3)))
  A1=(P(2)*CEXP(-CI*K*X(3))-P(3)*CEXP(-CI*K*X(2)))/DEN
  B1=(P(3)*CEXP(CI*K*X(2))-P(2)*CEXP(CI*K*X(3)))/DEN
  DEN=2.*CI*CSIN(K*(X(3)-X(1)))
  A2=(P(3)*CEXP(-CI*K*X(1))-P(1)*CEXP(-CI*K*X(3)))/DEN
  B2=(P(1)*CEXP(CI*K*X(3))-P(3)*CEXP(CI*K*X(1)))/DEN
  DEN=2.*CI*CSIN(K*(X(1)-X(2)))
  A3=(P(1)*CEXP(-CI*K*X(2))-P(2)*CEXP(-CI*K*X(1)))/DEN
  B3=(P(2)*CEXP(CI*K*X(1))-P(1)*CEXP(CI*K*X(2)))/DEN
  A=(A1+A2+A3)/3.
  B=(B1+B2+B3)/3.
  Z=-GAM*KR/K*(A*CEXP(2.*CI*K*XX)+B)/A*CEXP(2.*CI*K*XX)-B)
  ARGZ=CABS(Z)
  ZR=REAL(Z)

```



```

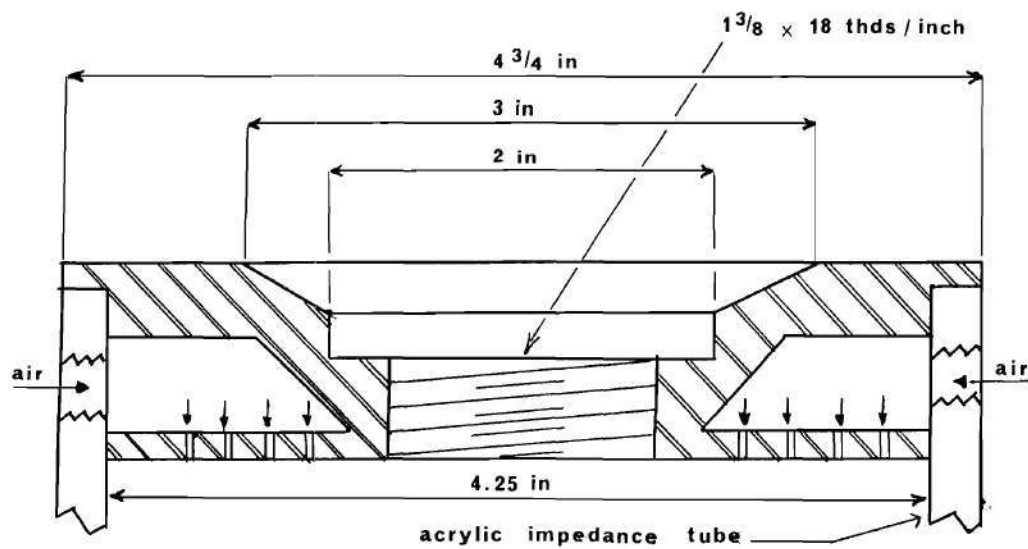
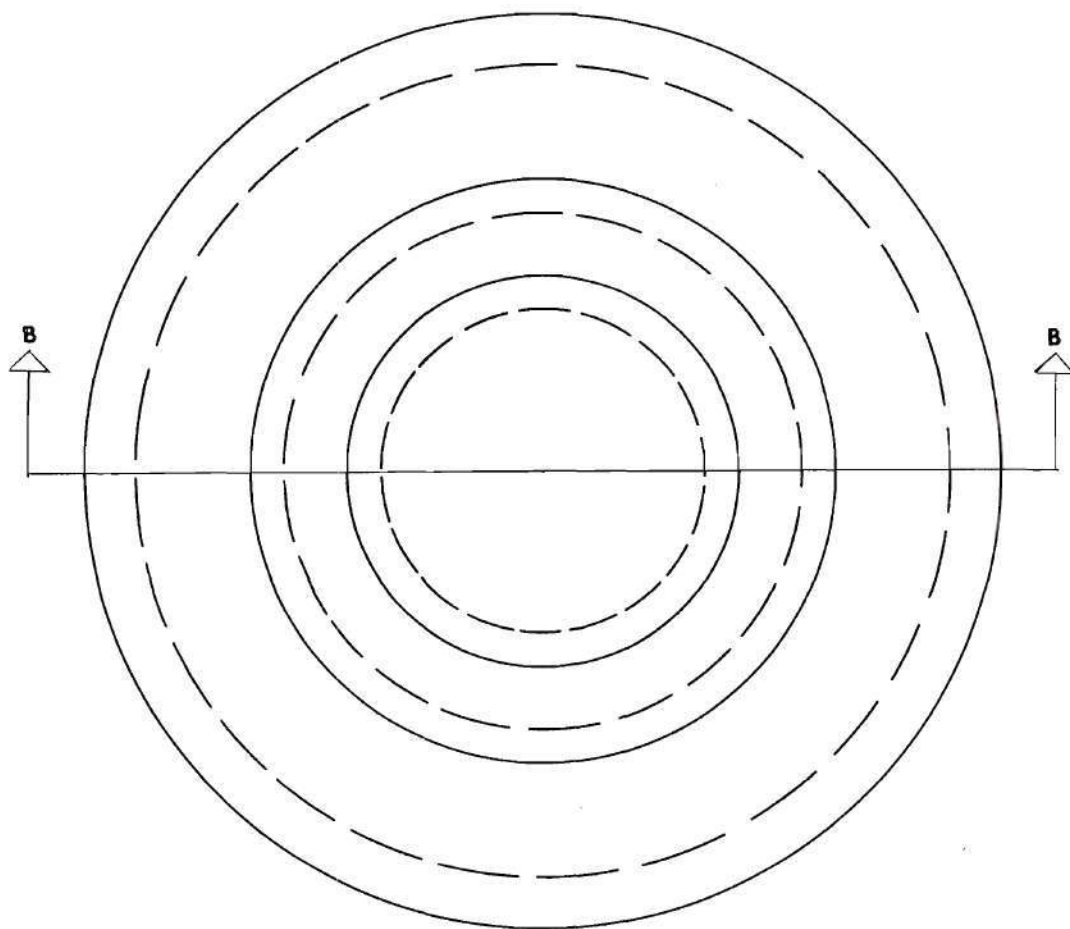
      ZI=AIMAG(Z)
      PHZ=ATAN 2(ZR,ZI)*180./PI
      WRITE(6,3500)A1,B1,A2,B2,A3,B3
3500  FORMAT(3(1H ,10X,3HA =,2(F13.4),6X,3HB =,2(F13.4)/))
      WRITE(6,3600)A,B
3600  FORMAT(1H ,10X,3HA =,2(F13.4),6X,3HB =,2(F13.4))
      WRITE(6,4000)ARGZ,PHZ
4000  FORMAT(1H ,13X,8HMAG(Z) =,F15.4,10X,10HPHASE(Z) =
      ,F15.4////////)
      9  CONTINUE
      10 CONTINUE
      STOP
      END

```

END OF COMPILATION: NO DIAGNOSTICS.

APPENDIX IV

ADDITIONAL FIGURES



SECTION B-B. MATERIAL ALUMINUM

Figure 2b. Enlargement

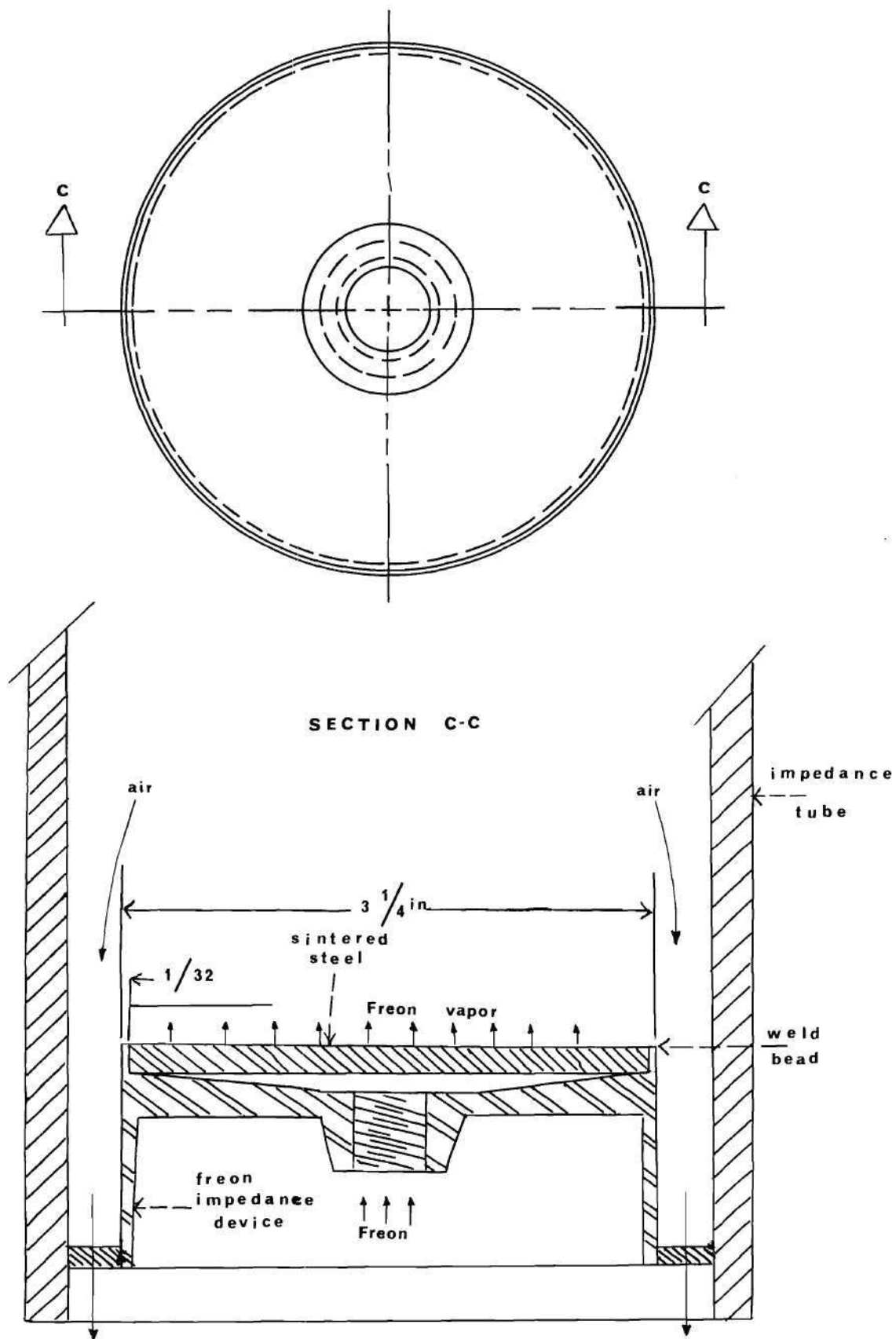


Figure 2c. Enlargement

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