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CREEP BEHAVIOR OF BOXES AND CORRUGATED BOARD

✓ Project 1108-30

Report Two

A Preliminary Report

to

TECHNICAL DIVISION FOURDRINIER KRAFT BOARD INSTITUTE, INC.

July 12, 1965

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Appleton, Wisconsin

CREEP BEHAVIOR OF BOXES AND CORRUGATED BOARD

SUMMARY

This study has for its purpose the development of information regarding the long-term load-carrying ability of corrugated board and boxes. Because of the long time intervals required to obtain data this report summarizes a preliminary analysis of the major data trends and explores methods of analyzing the creep deflection vs. time curves.

For the materials evaluated the results to date indicate that

1. The following average box failure lives were obtained from a relationship between applied load ratio and the logarithm of time.

Applied Load Ratio	Box Failure Life, days
0.70	11.5
0.65	. 32
0.60	. 92
0.55	. 260

2. At this stage the effects of flute and board series on the relationship between failure life and the applied load ratio are obscured by test variability.

3. The box failure lives tend to be considerably greater than those reported by Kellicutt and Landt.

4. The variability in box creep failure lives is large though much of the variability is apparently explained by the variability in conventional box

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compression tests. Because of the large variability, use of average failure life can be misleading.

5. In the box creep tests, failure seemed to occur as the deflection reached the value obtained in the conventional box test.

6. Column creep failure lives exhibited a similar relationship with applied load as the boxes but gave somewhat shorter lives, in general. The column creep lives were about as variable as the box creep lives.

7. A preliminary analysis of a portion of the box results was carried out to develop mathematical expressions to fit the box creep deflection <u>vs</u>. time curves. If the path of the creep curves can be predicted from conventional shortterm tests, it should be possible to

(a) estimate creep failure times from the intersection of the creep deflection time equation with the critical box deflection or

(b) estimate the creep time to reach any specified deflection level which might be associated with damage to the contents.

8. For this preliminary analysis, a power function was employed to describe the creep deflection <u>vs</u>. time curves. Regression equations of the following type were obtained:

$$t = \kappa_1 (D - D_0)^{K_2}$$

or

 $\text{Log t} = \text{Log K}_1 + \text{K}_2 \text{ Log D-D}_0$

where

 $\underline{t} = time$ $\underline{D} = deflection at time \underline{t}$

and

 $\underline{D}_{0}, \underline{K}_{1}$ and \underline{K}_{2} are constants.

9. Equations of the above type fitted the creep curves reasonably well. Estimates of box creep failure life were made which were only in fair agreement with observed values. However, creep test variability is large and would prevent close estimates in any case.

10. To be of greatest utility the constants in the regression equations must be constant for most combined board constructions or must vary in some predictable way with the applied load or other factors. Further analysis is needed to determine if generalized equations can be developed. Page 4 Report Two

INTRODUCTION

Stacking performance involves the interrelated variables of load, deformation, and time. The failure as a function of time of boxes exposed to constant loads during warehousing is a major use hazard for corrugated boxes. It is well known that a corrugated box subjected to warehouse stacking will support for a prolonged period only a relatively small fraction of the box compression strength as determined by a laboratory box compression test. For this reason a study is in process to provide information relative to the warehouse stacking (creep) characteristics of corrugated boards and boxes.

Because of the long time intervals and the variability in failure times of the boxes and board, the accumulation of data is slow. For that reason a preliminary analysis of available data is underway to explore the major data trends and methods of analyzing the data. This should facilitate use of the information and permit changes in emphasis of the experimental program where it appears desirable. This preliminary report discusses

(a) the relationship between applied load and creep failure life for boxes and short combined board columns,

(b) the effect of variability in box compression characteristics on the creep failure life variability of boxes, and

(c) analysis of deflection vs. time curves for boxes.

BACKGROUND CONSIDERATIONS

Paper is classified mechanically as a viscoelastic material. Its prerupture response to an applied load may include several types of deformations.

They include (a) immediate elastic deformation, (b) delayed elastic deformations which are recoverable in reasonable lengths of time after removal of load, and (c) nonrecoverable deformations which are not recoverable in a reasonable length of time after removal of load.

If strain or deformation is plotted against time, a curve having the form shown in Fig. 1 will be obtained for nearly all materials. Thus, the responses of most materials to long-term loads are quite similar though somewhat different mechanisms may be involved for dissimilar materials.

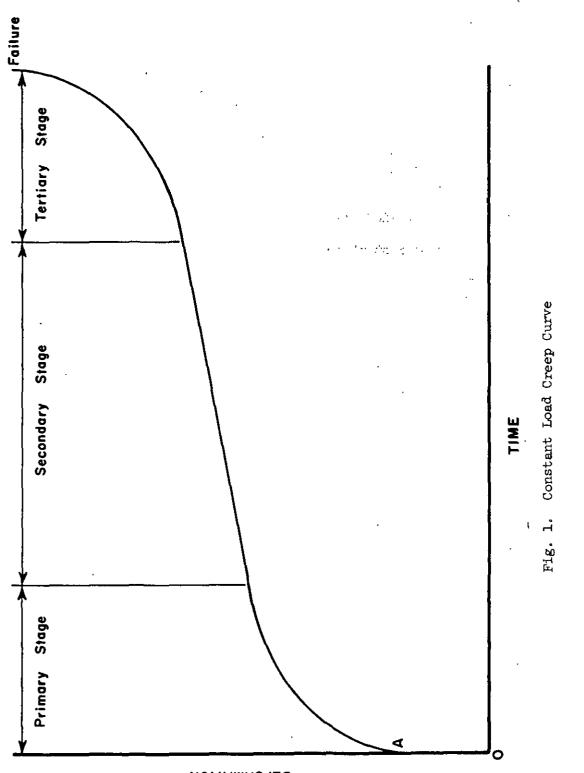
When load is applied the initial deflection OA is obtained. This is usually considered to be instantaneous and is composed of elastic, delayed elastic and plastic deformation.

In the primary stages of the creep curve the <u>rate</u> of deformation continuously decreases. In the second stage the rate of deformation is approximately constant. Although it is often treated as a straight line it may actually be a very flat curve with a point of inflection where the rate of deformation begins to increase again. In the third stage the deformation increases and failure eventually occurs.

The actual creep curve obtained often departs from that shown in Fig. 1. It will depend on such factors as the magnitude of the load applied, moisture content, temperature, etc. For example, if the load applied is large relative to box compression, the secondary stage may be absent entirely and an S-shaped curve will be obtained.

The tensile creep properties of paper have been studied by a number of investigators. For example, Brezinski has carried out a comprehensive study of the





DEFORMATION

creep behavior of paper $(\underline{1})$. A portion of Brezinski's results are shown in Fig.

2. Brezinski found that at early times or low loads his results could be described

in terms of the following equation

$$y/L_{o} = Bt^{a} + c$$
 (1)

where

y = creep deformation, inch (first load) $\overline{L} = initial$ specimen length, inch $\overline{t}^{O} = time$ of loading, seconds

and

B, a, and c are constants.

At longer times (e.g., secondary stage) or higher loads the deformationtime relationship became linear on the semilogarithmic plot and could be described by the following equation

$$y/L_{o} = K_{1} \log t + K_{o}$$
⁽²⁾

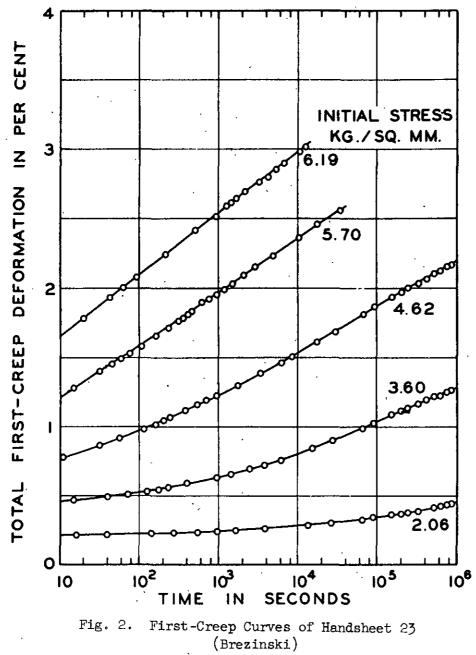
where \underline{K}_1 and \underline{K}_2 are constants.

Rance (2,3) has presented creep curves of paper at several constant loads. In general, the deformation increased rapidly with time and the log of the time to rupture decreased linearly with the applied load. Similar results were obtained by Jacobsen $(\frac{1}{2})$ and with cellulose films by Cheung (5).

There has been little published work relative to the creep characteristics of paperboard or boxes in compression. Kellicutt and Landt ($\underline{6}$) have published results for corrugated boxes showing a semilogarithmic relationship between applied load and time.

Recently Odqvist $(\underline{7})$ reviewed existing theories of ductile fracture by various authors and generalized a theory advanced by Kachanov ($\underline{8}$). The generalized theory explains the shape of the fracture time - applied stress curve.

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In a recent publication, Clouser $(\underline{9})$ reviewed the many mathematical expressions used to describe the creep curves of various materials. For the wood beams he used, a power equation similar to Equation (1) was used to describe the creep curve up to the point of inflection. Failure of the wood beams gave a linear relationship between stress level and the logarithm of time to failure.

There is an extensive literature regarding the creep behavior of various materials. These will be reviewed at a later date.

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MATERIALS

The box samples in Table I are under test in this program.

DISCUSSION OF RESULTS

BOX FAILURE TIMES AND DEFLECTION VS. APPLIED LOAD

The creep failure times and deflections for the boxes are summarized in Tables II and III and illustrated in Fig. 3. Inspection of the data indicates that the variability in stacking life at a given load ratio is large - often varying by a factor of 5 or 10 for a given sample and load ratio. This variability makes it difficult to determine whether other factors such as box size, flute, or board series significantly affect the relationship between failure time and the applied load rates.

Separate regression lines were computed for the A, B, and C flute data and for the combined data. The equations are given in Table IV.

While somewhat different regression equations and, hence, lines were obtained for the separate flutes, the limited data in B and C flutes suggest that the use of separate regression equations might be dangerous. This is particularly true for C-flute where the extremely long stacking times recorded for the 350-1b. series sample boxes had a major influence on the regression line. Whether the long stacking lives should be attributed to flute, to board series, or to some other factor is, at least, debatable at this time.

As a matter of interest the differences in slope were tested using IBM covariance program 6.0.032. The slope differences were not significant, suggesting that lines of common slope could be fitted to the data for the separate flutes.

		•		•										
-	pression Deflection, in.	Standard Deviation	0.0527	0.0251	0.0327	0.0611	0.0271		0.1657	0.0268		0.0071	0.0288	0.0233
	Defle	Av.	0.59	0.64	0.67	0.61	0.114		0.93	0.45		0.38	0.45	0.35
- - - -	Top Load Compression Load, no. Deflecti	Standard Deviation	47	Г ү	. 85	. 52	37		98	64		37	84	43
	Ĩ	Av.	040	975	1180	1060	555		1235	960		665	970	525
		Mrg. Joint	Taped	Taped	Taped	Taped	Taped		Glued	Stitched		Glued	Stitched	Glued
		Dimensions	16 x 12-1/4 x 9-1/2	21 x 17-1/2 x 19	23-1/2 x 23-1/2 x 19	23-1/2 x 14 x 12	13-1/4 x 6-5/8 x 12-1/2	-	12-1/¼ x 12-1/¼ x 19-13/16	16 x 12 x 18-1/2	-	15-3/8 x 10-1/4 x 11-3/4	17-7/8 x 16 x 11-3/8	15-1/2 x 10-1/4 x 14-3/8
		Series	200	200	200	200	175		350	275		200	275	175
		Flute	A	A	А	A	A		υ	C		ф	щ	മ

2457 2510

2498

5511

2497

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TABLE I

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LIST OF MATERIALS

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TABLE II

SUMMARY OF BOX CREEP RESULTS

					F	ailure Ti	ne, days				
Applied Load Ratio	Specimen No.	Sample 2406 A-200	Sample 2407 A-200	Semple 2408 A-200	Sample 2430 A-200	Sample 2456 A-175	Sample 2457 C-350	Sample 2497 B-200	Sample 2498 B-275	Sample 2510 C-275	Sample 2498 B-175
0.75	l	0.22	0.44	1.15		6.46	17.74		0.89	4.88	1.49
	2	0.47	32.08	1.21		4.16	9.36		2,46	13.89	3.27
	3	0.50	15.49	1.07		0.03	26.10		3.11	50.60	4.78
	4	0.32	0.70	1.39		0.24	153.20		6.22	10,39	0.16
	Av.	0,38	12.13	1,20		2.72	51.60		3.14	19.9 ⁴	2.42
0.70	1	over 250				33.1	62.6	8.9	127.0		10.9
	2	19.6	40.6			132.7	over 214	53.8	15.9		3.6
	3	20.0	41.7			over 68		9.5	35.7		10.3
	4	4.5	15.2					4.0	35.2		18.6
	Av.		32.7			82.9		19.0	53.4		10.8
0.675	1.	30.5	87.9								
	2	62.1	76.4								
	3	38.4									
	4	20.4									
	Av.	37.8									
0.625	1	33.8	15.8	16.5	29,8	115.4	72.6	19.9	263.0	over 2	over 30
	2.	4.8	13.6	79.6	62.9	167.4	over 373	7.7	160.4		over 2
	3	155.7	95.9	13.1	90.1	over 373	200.6 ^ª	140.6	214.3		
	4	115.4	93.8	2.6	over 372	over 374	405.8 ^ª	101.5	over 213		
	Av.	77.4	54.8	28.0	60.9	141.4	226.3	67.4			
0.575	1	over 93	over 93				over 274	over 283	over 284		,
	2					over 147					
	3					over 93			•		
0.55	1	113.3	366.6	129.3	-	over 371					
	2	114.5	over 371			over 374					
	3	174.4	over 371								
	4	243.6		over 83							
	Av.	161.4									
0.50	1		344.0								
	2		578.0 ^ª	-							
	3		728.9								
	Av.		550.3								

 $^{\rm A}_{\rm C}$ -velved two hours exposure to modelines as high as 80% R.H.

TABLE III

COMPARISON OF BOX CREEP DEFLECTIONS PRECEDING FAILURE WITH MAXIMUM DEFLECTION IN THE BOX COMPRESSION TEST

				1	Deflection	n, inch				
	Sample 2406	Sample 2407	Sample 2408	Sample 2430	Sample 2456	Sample 2457	Sample 2497	Sample 2498	Sample 2510	Sample 2511
Max. deflection (box			ı							
compression test), inch	0.59	0.64	0.67	0.61	0.44	0.93	0.38	0.45	0.44	0.31
Creep failure defl., inch ^a										
0.75 load ratio										
1	0.64	0.62	0.73		0.44	0.84		0.42	0.45	0.38
2	0,65	0.71	0.78		0.39	1.02		0.49	0.48	0.35
3	0.56	0.61	0.76		0.39	1.06		0.45	0.62	0.33
4	0.59	0.54	0.73		0.37	1,13		0.54	0.44	0.37
Av.	0.61	0.62	0.75		0.40	1.01		0.48	0.50	0.38
0.70 load ratio										
1		0.67			0.51	1.06	0.41	0.51		0.35
2	0.66	0.69			0.46		0.41	0.54		0.37
3	0.67	0.67					0.40	0.50		0.35
4	0.58	0.69					0.40	0.50		0.39
Av.		0.68					0.40	0.51		0.36
0.675 load ratio					I					
1	0.69	0.77							-	
2	0.66	0,70	•							
3	0.61									
4	0.54									
Av.	0.62									
0.625 load ratio										
1	0.63	0.67	0.62	0,61	0.44	0.99	0.38	0.52		
5	0.63	0.56	0.77	0.68	0.49		0.38	0.48		
3	0.70	0.68	0.74	0.61		0.95	0.43	0.52		
4	0.66	0.64	0,76			1.05	0.42			
Av.	0.66	0.64	0.72				0.40		·	
0.55 load ratio										
1	0.60	0.63	0.78		•					
5	0.58		0.74							
3	0.54		0.76							
4	0.64									
Av.	0.59									
0.50 load ratio									•	
1		6.66								
2		0.74								
3		0.75								
Αν.		r. 7 <u>2</u>		•						

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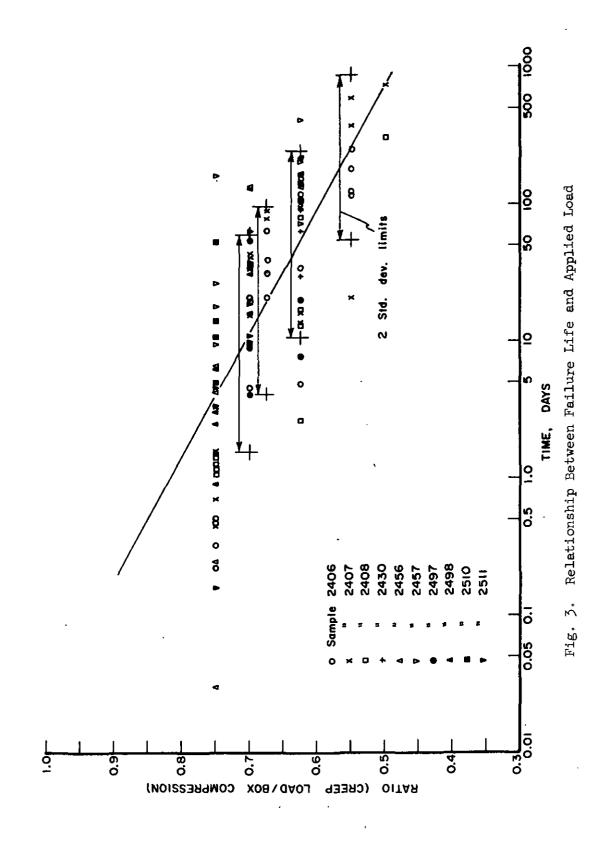


TABLE IV

BOX FAILURE LIFE REGRESSION EQUATIONS

Flute	N	Regression Equation ^a	Corr. Coefficient
А	59	Log <u>t</u> = 7.81655-9.936R	-0.77
С	12	$\log t = 7.10417 - 7.733R$	-0.73
В	27	$\log t = 9.96173 - 12.726R$	-0.77
Composite	98	$\log t = 7.38290 - 9.054R$	-0.71

 $a_{t=time}$ to failure, days and <u>R</u>=applied load ratio

In terms of averages, the composite regression line gives stacking times which are considerably greater than those reported by Kellicutt and Landt ($\underline{6}$). This comparison is shown in Table V. It is understood by the Institute that Kellicutt more recently has obtained longer stacking times with A-flute materials than originally reported. This would probably reduce the differences shown in Table V. In any event, it is evident that survival time for the empty box decreases drastically as the applied load approaches 65 to 75% of the conventional box compression load.

The use of averages such as the above can be misleading since warehouse stacking complaints may occur because of the poor performance of a few boxes in the lot. Therefore, box variability may require consideration.

The effects of "normal" box variability on survival time was estimated as follows.

1. Two standard deviation limits were computed for each box compression average using the data in Table I. Normally, 95% of the individual box tests should be found within this range.

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2. The applied load at each load ratio was divided by the upper and lower two standard deviation load value to give upper and lower limits on <u>the applied</u> load ratio.

3. Survival times at the upper and lower load ratios were read from the curve in Fig. 3.

TABLE V

BOX FAILURE LIFE

Load	J	<u>Box Fai</u>	lure Life, days Kellicutt and
Ratio		Fig. 3	Landt
0.75		4.0	0.6
0.70		11.5	2.0
0.675		18.5	3.6
0.65		· 32	7
0.625		54	1 ¹ 4
0.600		92	25
0.55		260	86

^aFrom Fig. 6, Reference $(\underline{6})$.

The stacking survival limits for empty boxes based on compression variability as computed above are shown in Table VI and the average limits are shown in Fig. 3. This procedure appears to account for much of the variability in the individual box stacking results.

It must be emphasized that the above estimates of variability are only concerned with variability of <u>individual boxes</u> from a given sample lot. To determine if two lots of boxes gave different stacking times, the number of boxes TABLE VI

BOX SURVIVAL TIME LIMITS BASED ON BOX COMPRESSION VARIABILITY

Variation in Failure Time in Days Due to Box Variability at the 2 Standard Deviation Level

	LOV	02	8	37	38	45	30	70		80	72	27	54
	<u>0.55 Ratio</u> gh Av. Lo	260	260	260	260	260	260	260		260	260	260	260
	0.5 High	720	620	>1000	>1000	1000	>1000	720		800	669	~1000	855
	Low	3.21	0.11	6.0	14.8	7.2	р . 8	13.0		10.5	13.4	1 ⁴ . ¹	10.8
	0.625 Ratio h Av. Lov	54	54	54	54	54	54	54		45	54	2 4	54
TOLOT HOT ADT ADT	0.6 High	· 180	105	280	280	250	330	190		200	180	340	234
	tio Low	٥ . 4	5.4	1.8	1.9	2.2	1.4	3.8		3.3	5.0	1.25	4.08
	0.675 Ratio gh Av. Lov	18.5	18 . 5	18.5	18.5	18.5	18.5	18.5		18.5	18.5	18.5	18.5
5	0.(High	68	55	211	108	. 100	130	70		62	69	148	46
	Low	2.3	3.1	0.99	1.08	2.25	0.72	2.2		1.8	2.3	0.65	1.53
	0.70 Ratio High Av. Lov	3 11. 5	11.5	11.5	11.5	11.5	11.5	11.5		11.5	11.5	11.5	п.5
	High.	43	36	70	70	62	52	44		4 6	ł3	88	58
	Series	200	200	200	200	175	350	275		200	275	175	Av.
	Flute	4	Ą	4	4 ,	Ą	σ	U		, да	ക	ф	
	Nc.	2406	2043	24,08	5430	2456	2457	2510	,	2497	2498	2511	

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evaluated for each sample would have to be taken into account in calculating a standard error of the difference so that a "t" test could be made.

DEFLECTION AT FAILURE VS. APPLIED LOAD

The box deflections near failure in the creep tests are summarized in Table III. In general, the creep failure deflections were approximately equal to the deflections at maximum load in the box compression test. This means that if suitable mathematical functions can be found to describe the relationship between deflection and time, the failure time may be estimated without carrying out lengthy creep tests. This will be discussed in a later section.

COLUMN FAILURE TIMES AND DEFLECTION VS. APPLIED LOAD

To provide information relative to the creep behavior of corrugated board in compression, creep tests are in process on short columns of the type used in evaluating the edgewise compression strength of corrugated board. The test is one of the basic factors governing conventional box compression tests.

The creep failure lives and deflections are shown in Tables VII and VIII and Fig. 4. In general, the creep failure lives of the short columns increased as the applied load decreased in much the same manner as the boxes. The variability of the columns creep failure lives is qualitatively about as great as found for the boxes.

The column creep lives show some correlation with the box creep lives as shown in Fig. 5. However, the relationship is not very precise due to the test variability. Refinements in test methodology and analysis will probably be required to use column creep tests in box creep life prediction. Column creep

TABLE VII

PRELIMINARY DATA ON COLUMN CREEP BEHAVIOR OF COMBINED BOARD

				Failur	e Time, d	ays		
Applied Load	Speci- men	Sample 2406	Sample 2407	Sample 2408	Sample 2430	Sample 2456	Sample 2457	Sample 2497
Ratio	No.	A-200	A-200	A-200	A-200	A-175	<u>C-350</u>	B-200
Maximum load,				<u> </u>				he o
1b./in.		51.1	42.2	45.7	51.9	34.3	74.4	45.9
0.75	1 2 3 4 5	1.41 0.31 1.51 0.07 1.01	0.17 0.05 0.16 5.57	0.16 0.52 7.01		1.06 3.55 0.08 0.07 0.36	0.79 0.95 0.33	5.18 0.05 0.04
•	Av.	0.86	1.49	2.56		1.02	0.69	2.03
0.70	1 2 3	14.61 2.00 ^a	3.42 ^b 3.63 ^b	21.17 _b 1.37 ^b		8.19 4.00 25.92	1.03 9.44	0.30 6.90
	Av.	14.61	3.52	11.27		12.70		3.60
0.686	1 2 Av.		0.56 ^b 2.26 1.41			·		
0.625	1 2 3 4 5 6 7 8	20.38 ^b 85.56 ^b 22.08 ^c 4.39 ^b 11.07	12.30 46.70 6.83 6.05 14.44 23.37 ^b 7.79 14.44	26.64 ^b 40.97 ^d 219.49 10.64	38.40 ^b 23.14 21.11 2.00 ^a	30.75 98.99 27.64 46.10 92.03 120.56	45.97 56.63 ^b 28.97 150.00 13.66 ^b	99.08 ^e 15.59 ^b 8.42 ^b
	Av.	28.70	16.49	74.44	27.55	69.34	36.31	41.03

^aSpecimen has not reached failure and is still under test; these values are not included in any of the averages. Failure time is $\frac{1}{10}$ 0.31 days. Failure time is $\frac{1}{10}$ 0.40 days. Failure time is $\frac{1}{10}$ 0.90 days. Failure time is $\frac{1}{10}$ for a second second

TABLE VIII

COLUMN CREEP DEFLECTIONS PRECEDING FAILURE

Applied	Speci-							
Load Ratio	men No.	Sample 2406	Sample 2407	Sample 2408	sample 2430	Sample 2456	Sample 2457	Sample 2497
0.75	1 2 3 4 5 Av.	0.028 0.013 0.056 0.046 0.036	0.034 0.030 0.032	0.031 0.035 0.037		0.025 0.030 0.015 0.023	0.025 0.023 0.062 0.052	0.028 0.023 0.026
ò.70	1 2 3 4	0.047	0.031 0.032 0.010	0.034 0.040		0.024 0.028 0.041	0.028 0.040	0.010 0.021
	Av.		0.024	0.037		0.031	0.034	0.016
0.625	1 2 3 4 5 6 7 8	0.032 0.031 0.034 0.039	0.026 0.032 0.030 0.036 0.041	0.027 0.035 0.030	0.033 0.023	0.042 0.033 0.032 0.027 0.031	0.032 0.039	0.023 0.029 0.022
	Av.	0.034	0.033	0.032	0.028	0.033	0.036	0.025

^aThe creep failure deflection is the last recorded deflection preceding the collapse of the column.

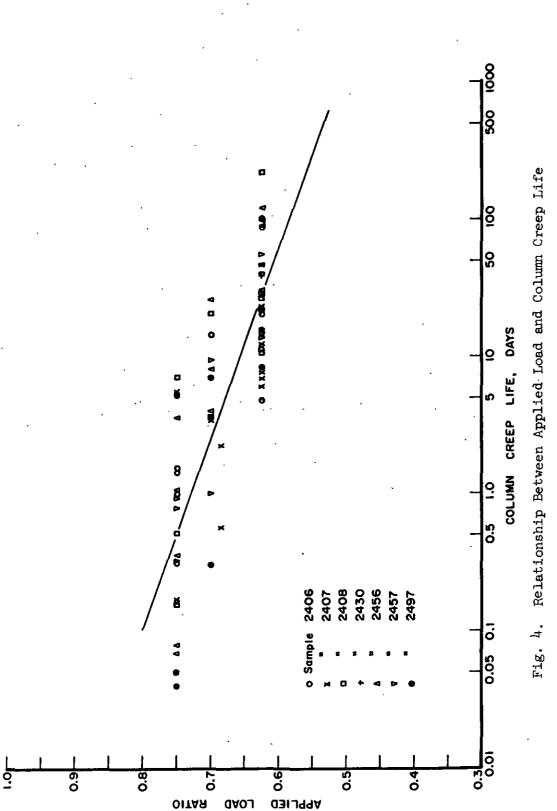
tests can be useful for studying the creep response of materials and the effect of various fabrication effects such as glue skips, finger lines, etc.

RELATIONSHIP BETWEEN BOX DEFLECTION AND TIME UNDER LOAD

During the box creep test, the deflection gradually increases. Failure occurs as the deflection nears the deflection attained in the box compression test. An idealized representation is shown in Fig. 6. When Load \underline{R}_1 is applied

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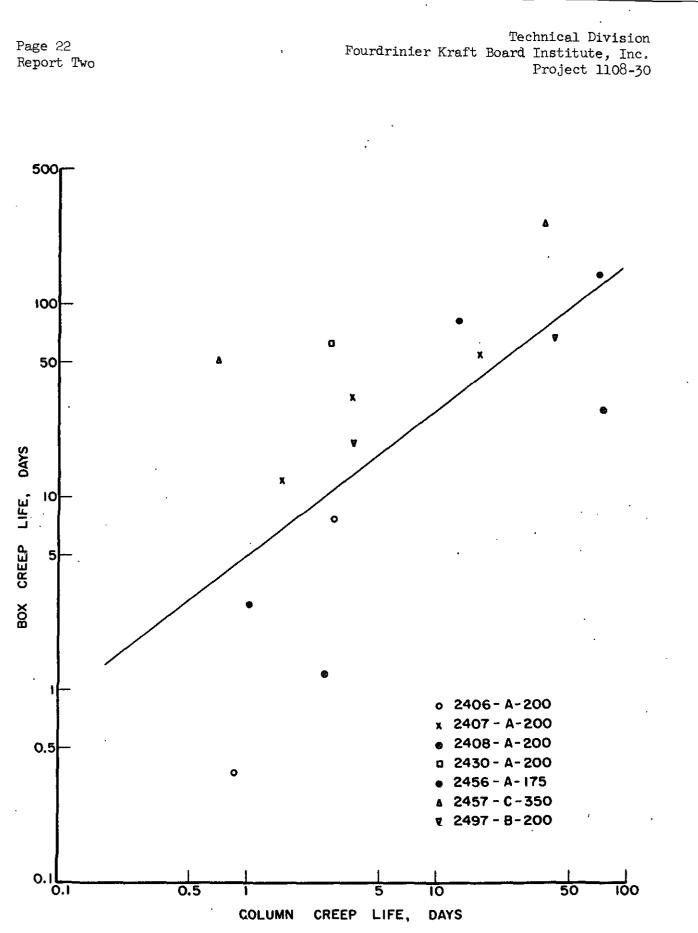
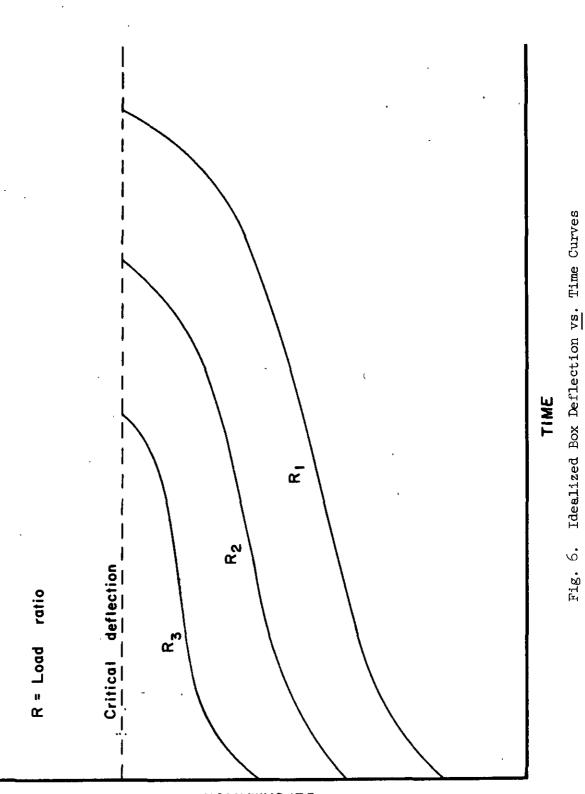


Fig. 5. Relationship Between Short Column and Box Creep Failure Life

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the time to failure is long although failure occurs as the critical deflection is reached. When a larger Load, \underline{R}_2 or \underline{R}_3 , is applied failure occurs in a shorter time at about the same deflection level.

If it were possible to predict the path of a curve from data determined in short term tests, failure time estimates could be made. In addition, if it were known that a given product would be damaged at some deflection less than the critical deflection, the time required to reach such deflections could also be estimated. Therefore, an analysis of the data from this viewpoint should have merit.

Box deflection \underline{vs} . time curves depart considerably from the ideal. For individual boxes the shape of the curves, the amount of deflection at a given time and load, and the deflections reached before failure can be quite variable. This variability complicates the selection of suitable functions to describe the deflection \underline{vs} . time curve and can give rise to poor predictions of creep failure life.

A series of deflection <u>vs</u>. time curves for Sample 2406 are illustrated in Fig. 7. The box data graphed were selected to have stacking lives near the average for the particular ratio. Preliminary trials indicated that curves of the type shown in Fig. 7 could be described by an expression of the following type

$$\log t = \log K_1 + K_2 \log(D-D_0)$$
(3)

where

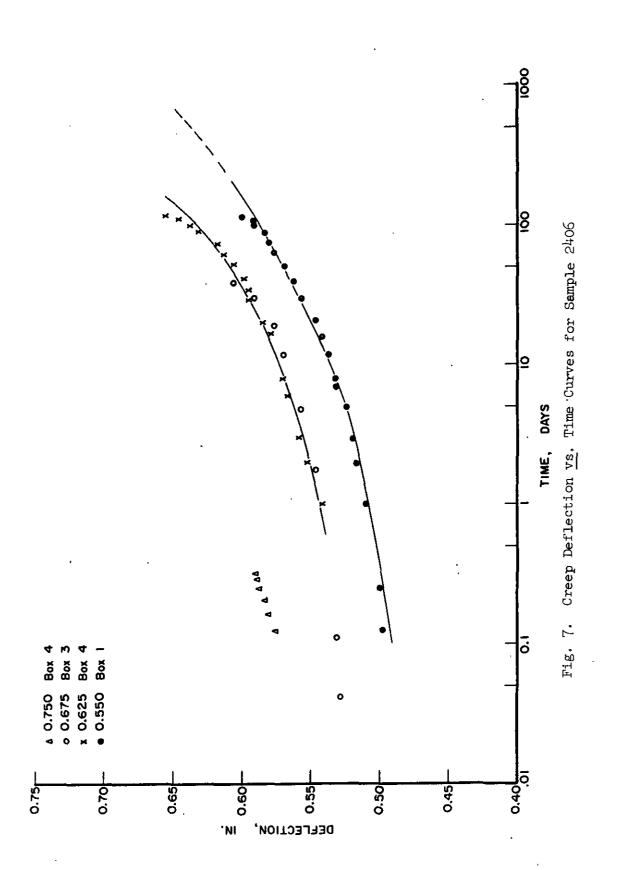
$$\frac{t}{D} = time$$

 $\frac{t}{D} = deflection$

and

$$K_1$$
, K_2 and D_2 are constants.

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Equation (3) is another form of Equation (1) and has been extensively used in creep curve analyses for other materials. The curves shown in Fig. 7 are for the 0.55 and 0.625 boxes and have the form of Equation (3).

Since three constants are involved, a standard least squares solution cannot be directly obtained. For this preliminary analysis, after a few trials the constant \underline{D}_0 was set equal to the average deflection at 60 min. (\underline{D}_6) for the given ratio. The difference $(\underline{D}-\underline{D}_6)$ was then correlated with time to determine the constants \underline{K}_1 and \underline{K}_2 .

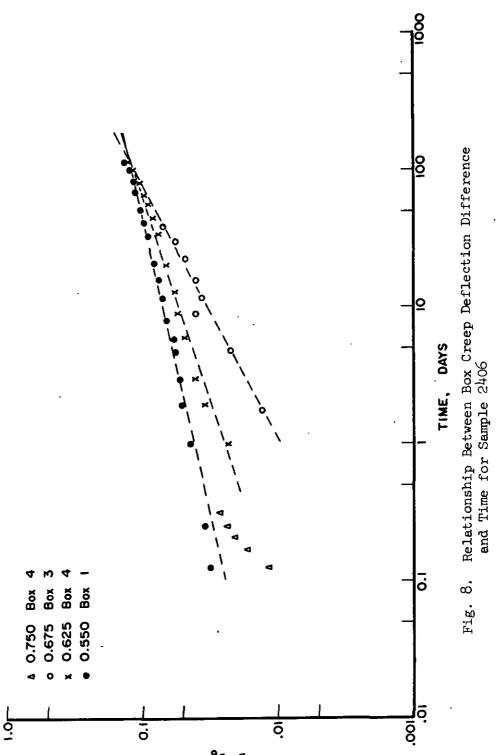
At this stage the choice of \underline{D}_6 was somewhat arbitrary. A more thorough analysis would determine whether other values of \underline{D}_0 might be more generally applicable. Other functional choices should also be investigated.

An analysis of the type described above was carried out for 9 boxes from Samples 2406 and 2407. A graph of $(\underline{D}-\underline{D}_6)$ vs. time is shown in Fig. 8 and 9 for Samples No. 2406 and 2407, respectively. As may be noted, Equation (3) gives a reasonable fit to the data to nearly the failure point. The regression equations are summarized in Table IX. High correlation coefficients were exhibited by all the regression equations.

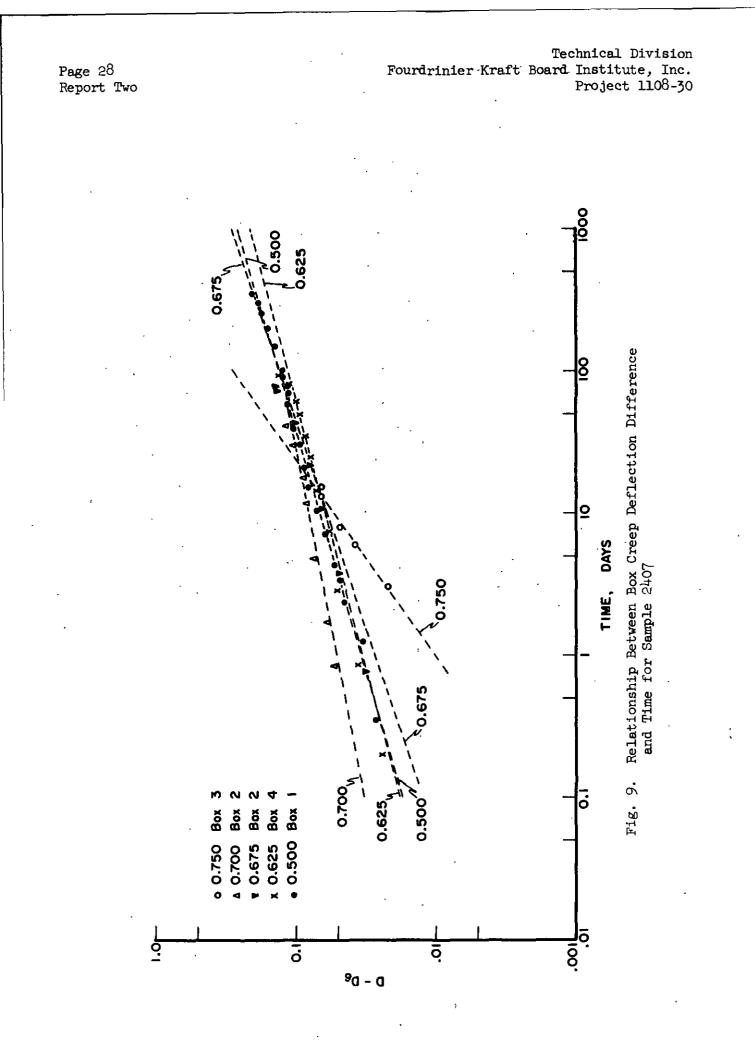
Estimates of box failure life were made by substituting the average value of box deflection from the conventional compression test for <u>D</u>. Estimates were also made at the average deflection plus or minus one standard deviation to allow for variability. These results are shown in Table X. As an example, based on the data for Box 1, Sample 2407, load ratio 0.50, a failure life of 310 days was estimated using the average deflection to failure in the box compression test (0.6400). The box actually failed at 344 days. The difference of 34 days is probably acceptable at this time. Allowing for variation in the average box

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DEFLECTION VS. TIME REGRESSION EQUATIONS

TABLE IX

	Corr. Coeff.	0.97	0.98	0.99	0.99	0.99	0.98	0.99	0.99	0.99	
	Regression Equation	$\log \underline{t} = 1.2371 + 1.120 \log (\underline{D} - \underline{D}_6)$	$Iog t = 3.7188 + 1.845 log (D-D_6)$	$Iog \underline{t} = \frac{1}{4}, 8188 + 3.013 \log (\underline{D} - \underline{D}_6)$	$\operatorname{Log} \underline{t} = 6.0822 + 4.441$ log $(\underline{D} - \underline{D}_6)$	$Iog \underline{t} = 2.7551 + 1.382 \log (\underline{D}-\underline{D}_6)$	$Iog t = 6.3859 + 4.962 log (D-D_6)$	$Iog t = 4.6538 + 3.066 log (D-D_6)$	$\log t = 5.6329 + 3.891 \log (\underline{D} - \underline{D}_6)$	$Iog t = 5.0558 + 3.475 log (\underline{D}-\underline{D}_6)$	- of deta points after discarding negative values of $\underline{D}-\underline{D}_6$ alues less than 0.1 day.
	А Р	0.5634	0.5336	0.5185	0.4656	0.5415	0.5698	0.5665	0.5097	0.4534	liscarding
	a La	9	ZT	35	94	Ś	18	23	30	76	after di 0.1 day.
	Ы	13	17	0 1	52	10	22	28	34	83	ints then
	Specimen No.	ل ہ	б	4		m	N	Q	4	Ъ	^a The number of data points and time values less than
,	Load Ratio	0.75	0.675	0.625	0.55	0.75	0.70	0.675	0.625	0.50	number o time val
	No.	2406				2407					a _T he r and t

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TABLE X

ESTIMATES OF BOX FAILURE LIFE

					Box Creep Life, days							
					Observed							
No.	Load Ratio	$\underline{D}^{\mathbf{b}}$	<u>D-D</u> 6	Estimated	Selected Box	Av. ^a	Max. ^a	Min. ^a				
2406	0.75	0.5900 0.6427 0.5 373	0.0266 0.0793 	0.24 0.98 	0.32	0.38	0.50	0.22				
	0.675	0.5900 0.6427 0.5373	0.0564 0.1091 0.0037	19 85 	38.4	37.8	62.1	20.4				
	0.625	0.5900 0.6427 0.5373	0.0715 0.1242 0.0188	25 115 0.45	115.4	77.4	155.7	4.8				
	0.55	0.5900 0.6427 0.5 373	0.1244 0.1771 0.0717	120 420 11	113.3	161.4	24 3. 6	113.3				
2407	0.75	0.6400 0.6651 0.6149	0.0985 0.1236 0.0734	23 32 12	15.5	- 12	32.1	0.44				
	0.70	0.6400 0.6651 0.6149	0.0702 0.0953 0.0451	4.5 22 0.52	• 40.6	32.7	41.7	20.4				
	0.675	0.6400 0.6651 0.6149	0.0735 0.0986 0.0384	18 41 2.7	76.4							
	0.625	0.6400 [.] 0.6651 0.6149	0.1303 0.1554 0.1052	160 300 72	93.8	54.8	95.9	13.6				
	0.50	0.6400 0.6651 0.6 1 49	0.1866 0.2117 0.1615	310 520 200	344	550.3	728,9	344.0				

Average, maximum and minimum of the four test results at each load ratio. Failure lives read at the average box deflection and the average <u>+</u> 1 standard deviation.

compression deflection of \pm one standard deviation gave estimates of 200 and 520 days. This implies that the odds should be about 1 in 3 that the life is less than 200 or more than 520 days. Even wider limits would be obtained by using 2 or 3 standard deviation limits.

In Table X the agreement with observed values is not good in all cases although a more extensive analysis may improve predictions. Extremely accurate estimates cannot be expected, however, because of the variability in box creep test as shown in Table II.

To be of utility, however, the constants in the regression equations must be constant for most combined board constructions or must vary in some predictable manner with the applied load ratio and short-term tests. In this connection it seems likely that \underline{D}_6 can be related to the applied load ratio, and the maximum deflection in the conventional compression test.

In Table IX the regression equation slope is smallest for the 0.75 ratio. However, the high ratios are not of great practical interest because they result in short box lives. At the lower load ratios, it is possible the slope coefficient might be either constant (Sample 2407) or related to the load ratio (2406). Additional work is needed to clarify whether either hypothesis is workable. The same remarks hold true for the intercept coefficients in Table X.

It should be emphasized that other mathematical expressions might be suitable for this type of analysis. The investigation of alternative solutions should certainly be carried out in future work. Page 32 Report Two Technical Division Fourdrinier Kraft Board Institute, Inc. Project 1108-30

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